



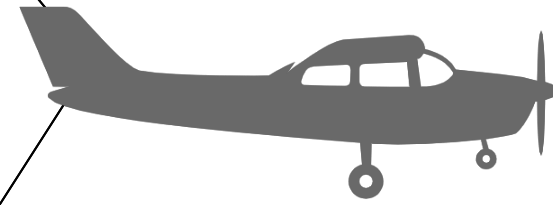
***EML4552-2***

**TEAM 513: SAE AERO  
DESIGN COMPETITION**

06-Feb-20



# Team Introductions



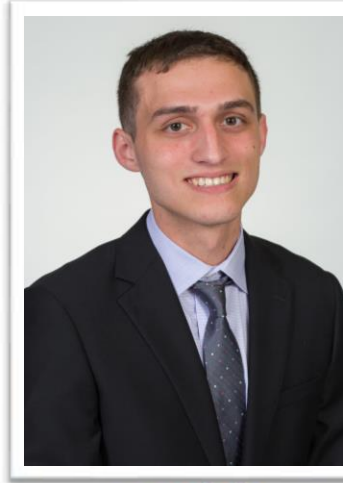
**Nestor  
Aguirre**  
Aeronautics/  
3D Printing  
Engineer



**Zachary  
Silver**  
CAD  
Engineer



**Martina  
Kvitkovicova**  
Electronics  
Test Engineer



**David  
Litter**  
3D Printing  
Engineer

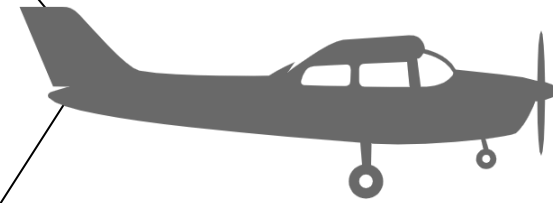


**Hebert  
Lopez**  
Electrical  
Design  
Engineer



**Leah Evans**  
Aeronautics  
Engineer/  
Financial  
Advisor

# Sponsor and Advisor



Florida Space  
Grant Consortium

Providing  
Funding



Seminole RC  
Club

Providing  
Equipment



3D Solutech

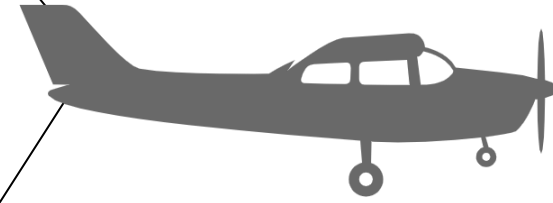
Providing  
Various  
Filaments



Dr. Shih

Providing  
Technical  
Knowledge

# Objective



- ✈ The objective of this project is to design and manufacture a 3D printed remote controlled (RC) airplane that complies with all rules and regulations for competing in the regular class of the SAE Aero Design East competition.



Overview

Mission Requirements

Performance Analysis

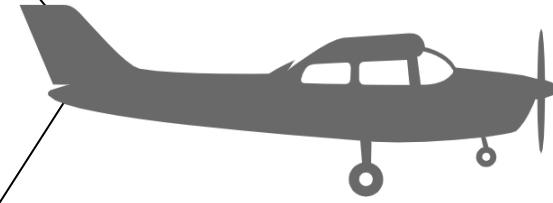
Design Overview

Detailed Design

Future Work

Review

# Project Summary



- ✈️ Compete in the SAE Aero Design East Competition in March 2020
- ✈️ Use additive manufacturing
- ✈️ Improve upon prior teams' designs
- ✈️ Innovate novel solutions



Overview

Mission Requirements

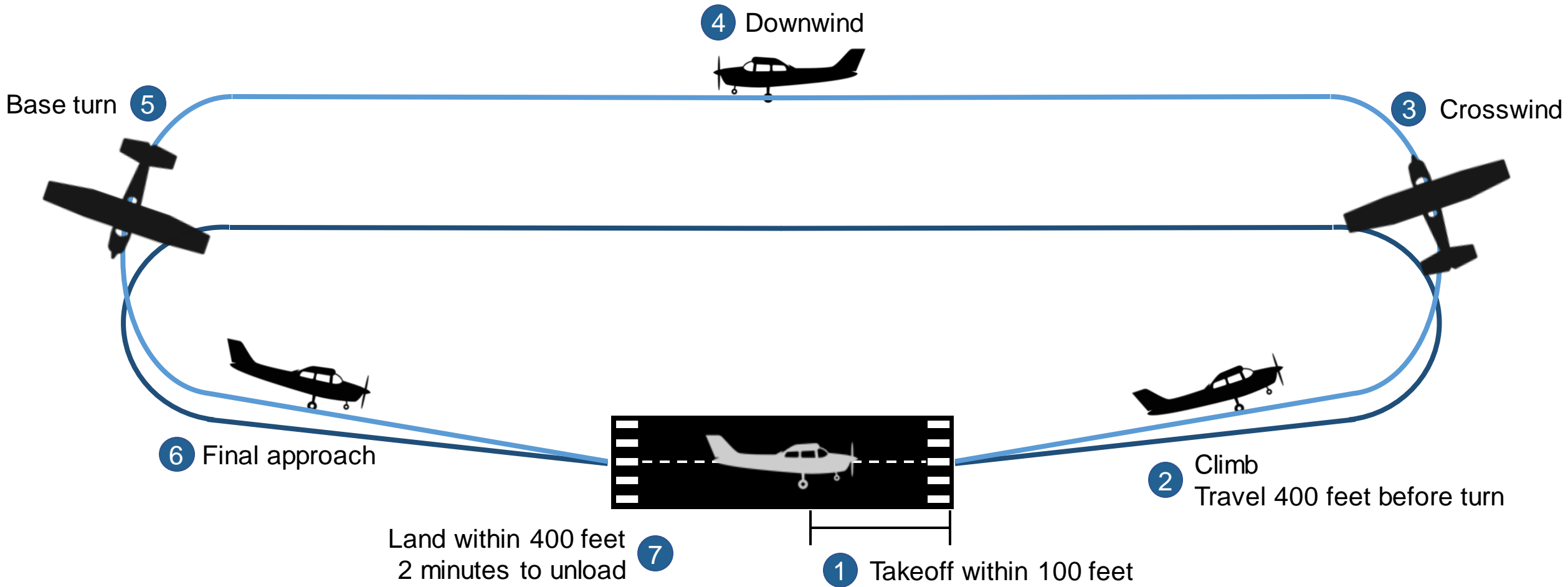
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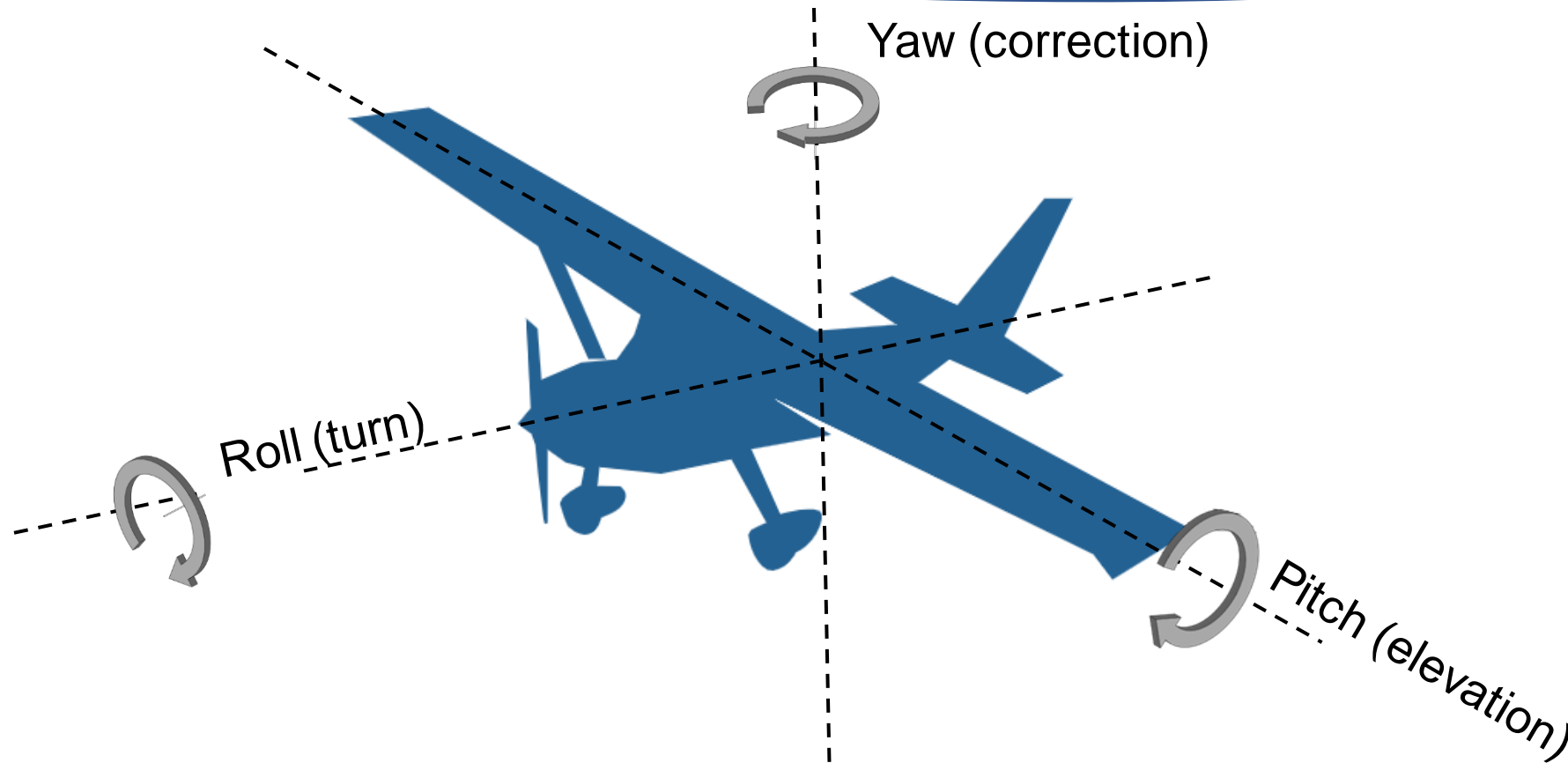
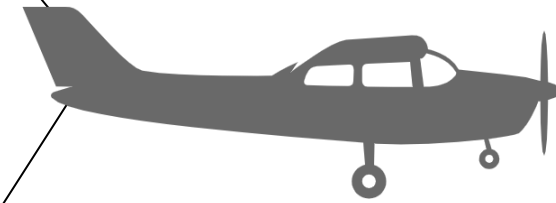
Review



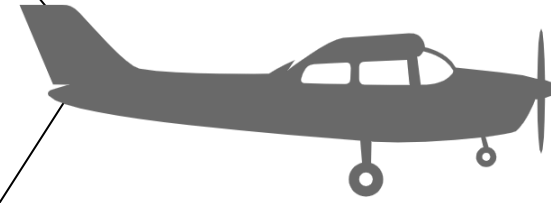
# Competition Mission Requirements



# Functional Decomposition: Maneuver in Flight

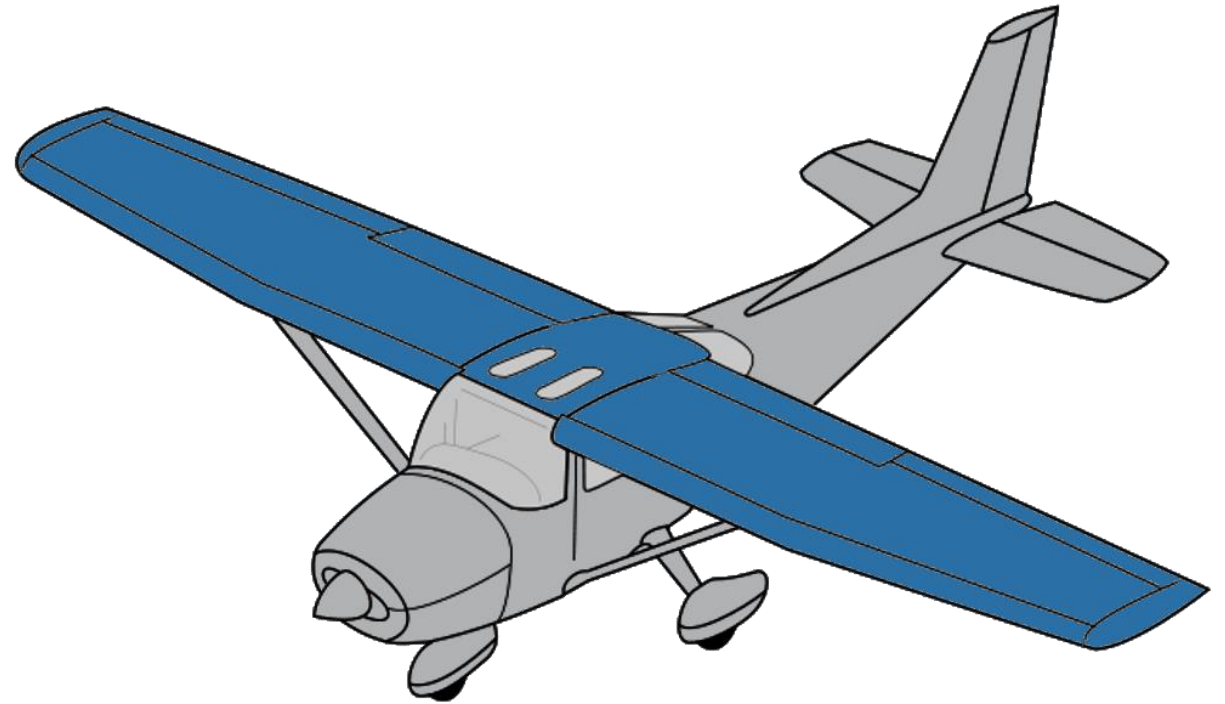


# Mission Requirements



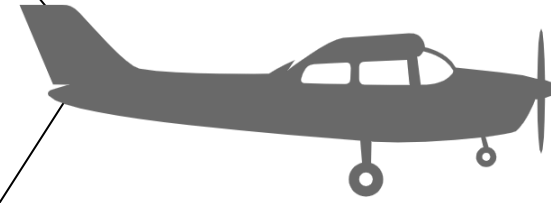
## Generate Lift:

- ✈ Wingspan: 60 – 120 inches
- ✈ Wing loading: 10 – 20 oz/in<sup>2</sup>
- ✈ Lift coefficient: 1.4 – 2.5



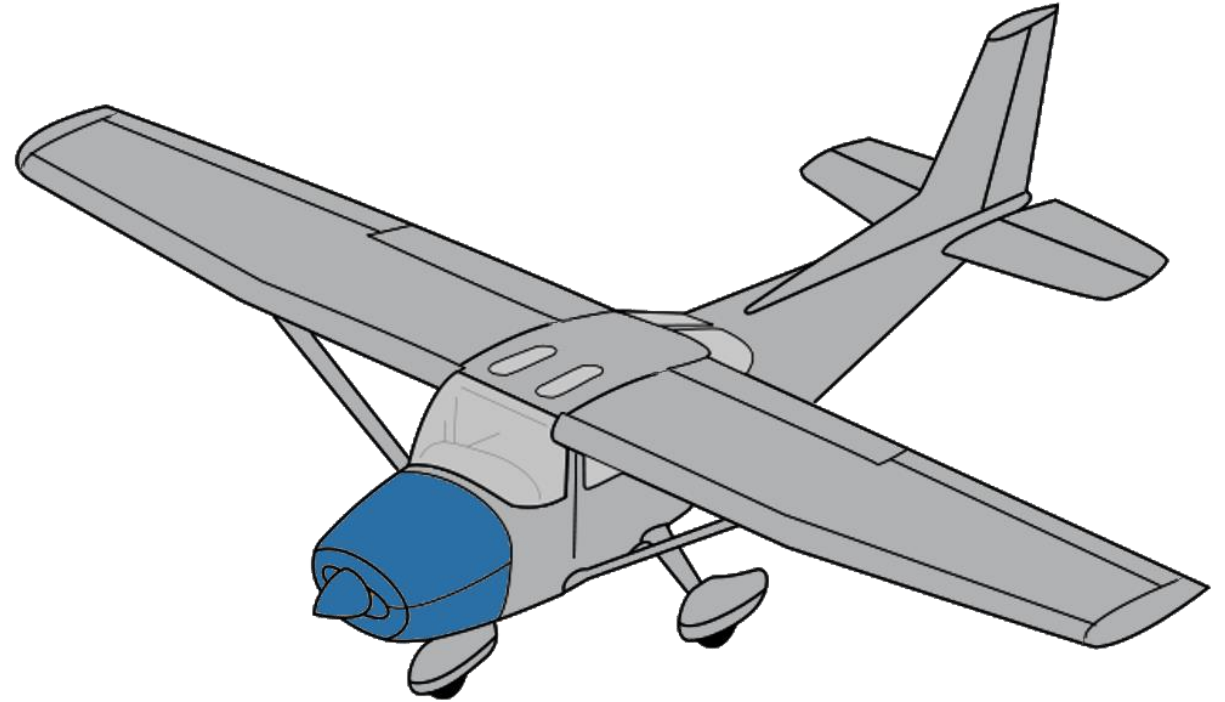


# Mission Requirements

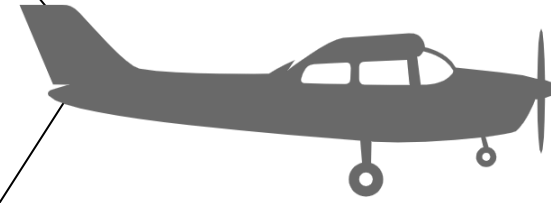


Accelerate/Decelerate:

- ✈ Static Thrust: 8 – 12 lbs
- ✈ Takeoff Speed: 20 – 30 mph
- ✈ Takeoff Distance: < 100 ft

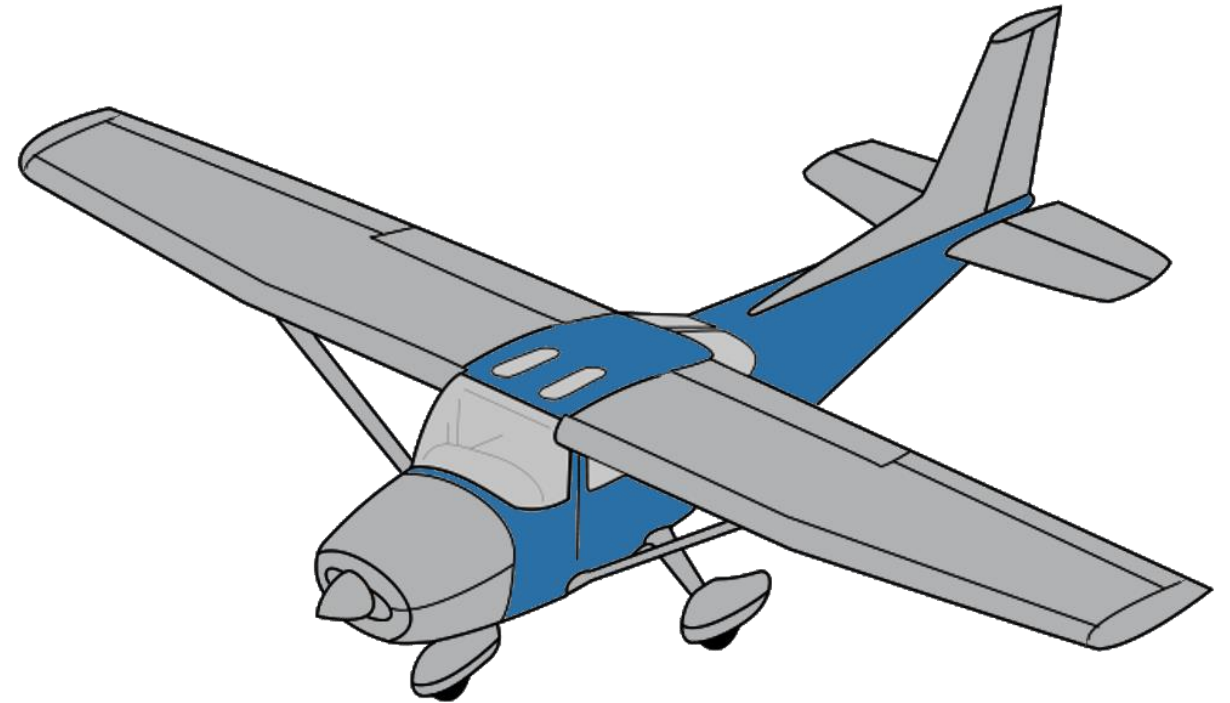


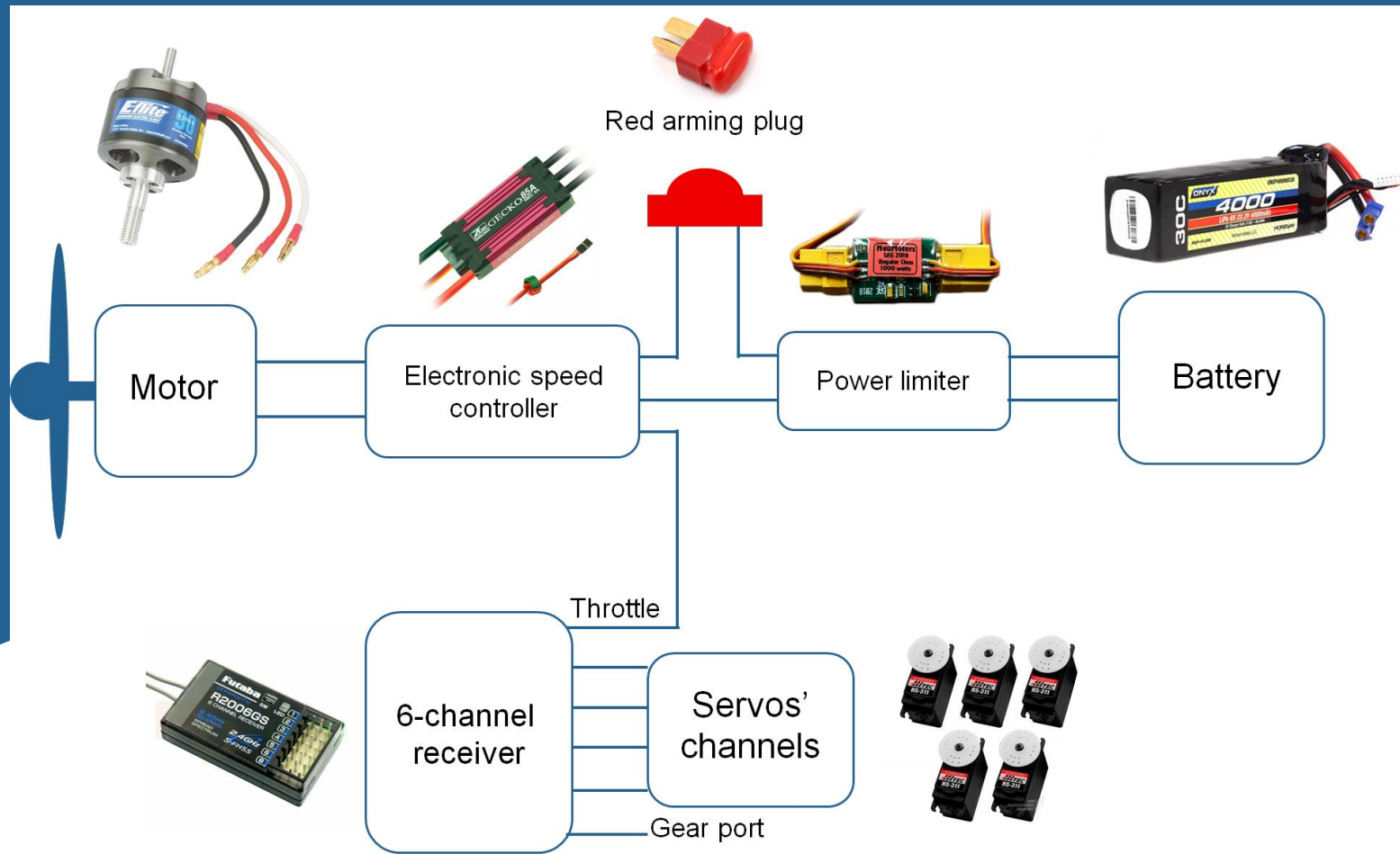
# Mission Requirements



## Transport Payload:

- ✈ Unload time: < 2 minutes
- ✈ Payload capacity: < 5 lbs
- ✈ Payload area: 10 inch<sup>2</sup>

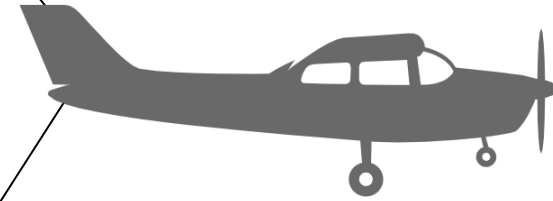




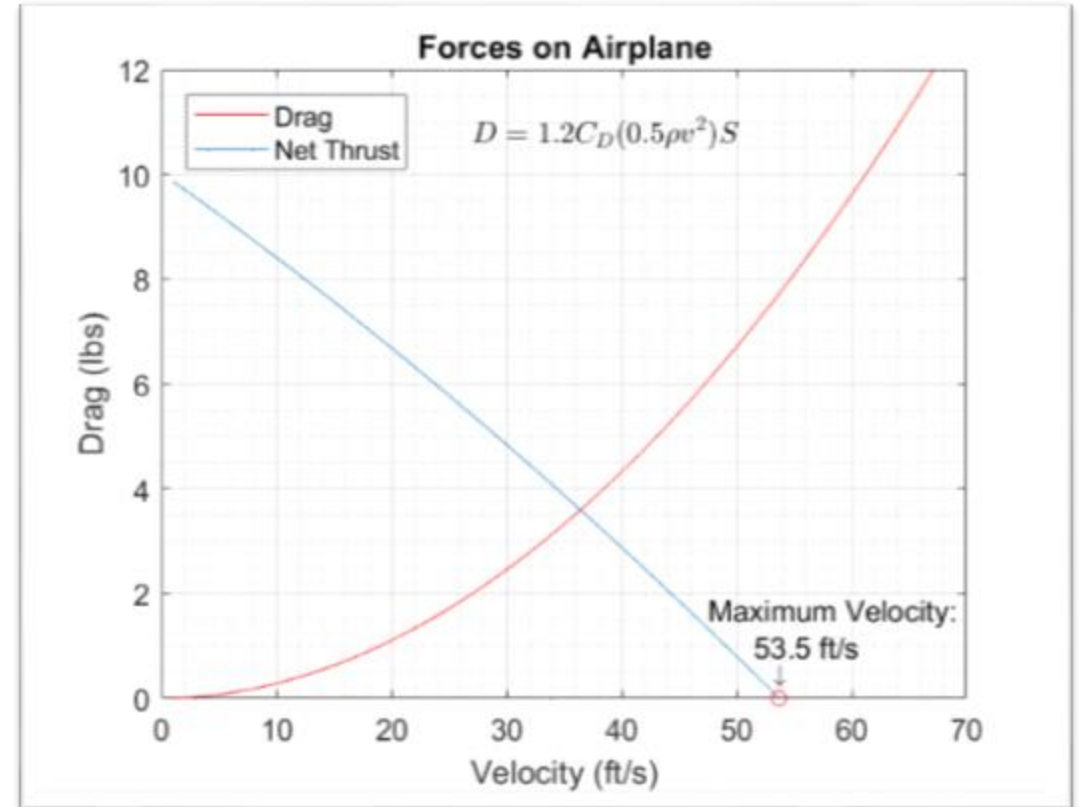
## Electronics Setup

The final wiring diagram for the plane, showing the acquired parts and components to complete the circuit

# Testing



## Thrust Performance Analysis



Overview

Mission Requirements

Performance Analysis

Design Overview

Detailed Design

Future Work

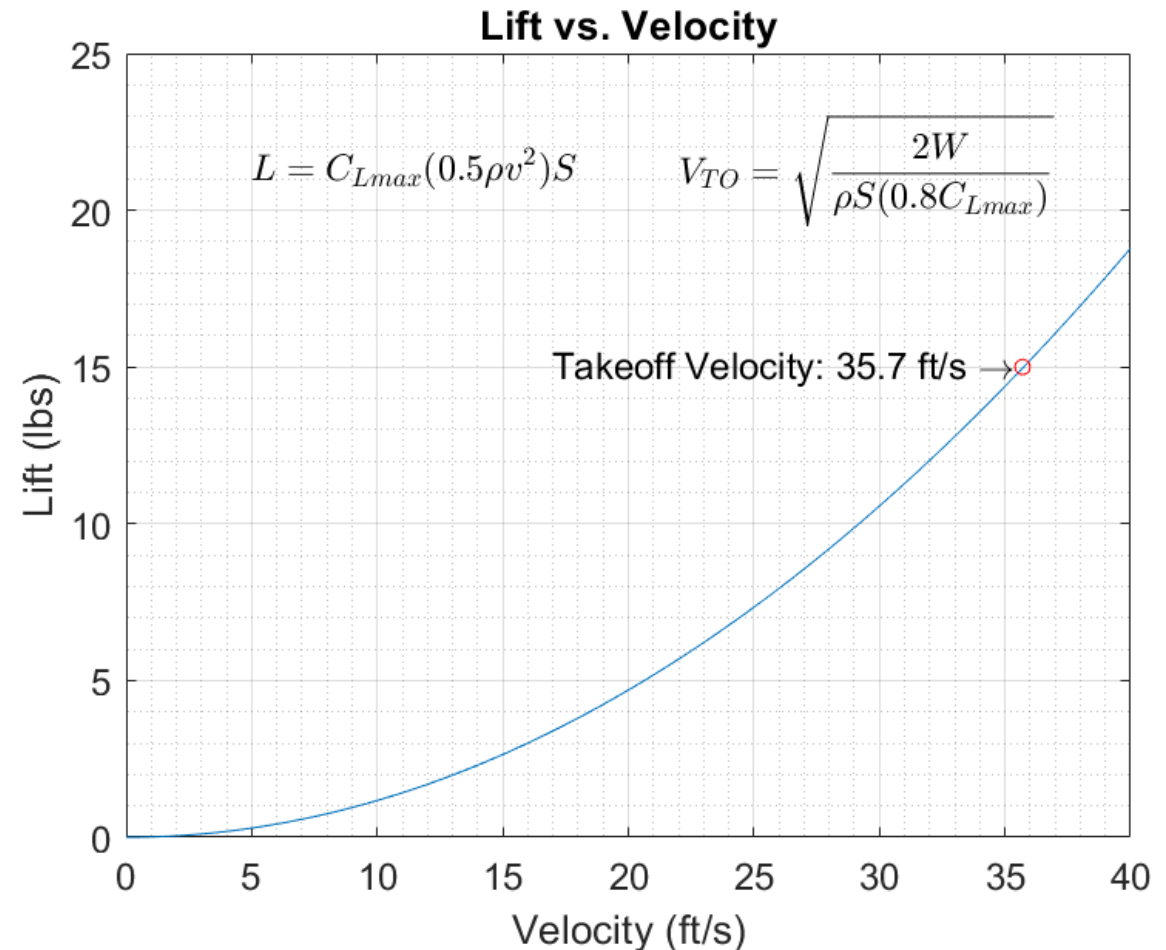
Review

# Performance Analysis: Lift

## ✈ Variables:

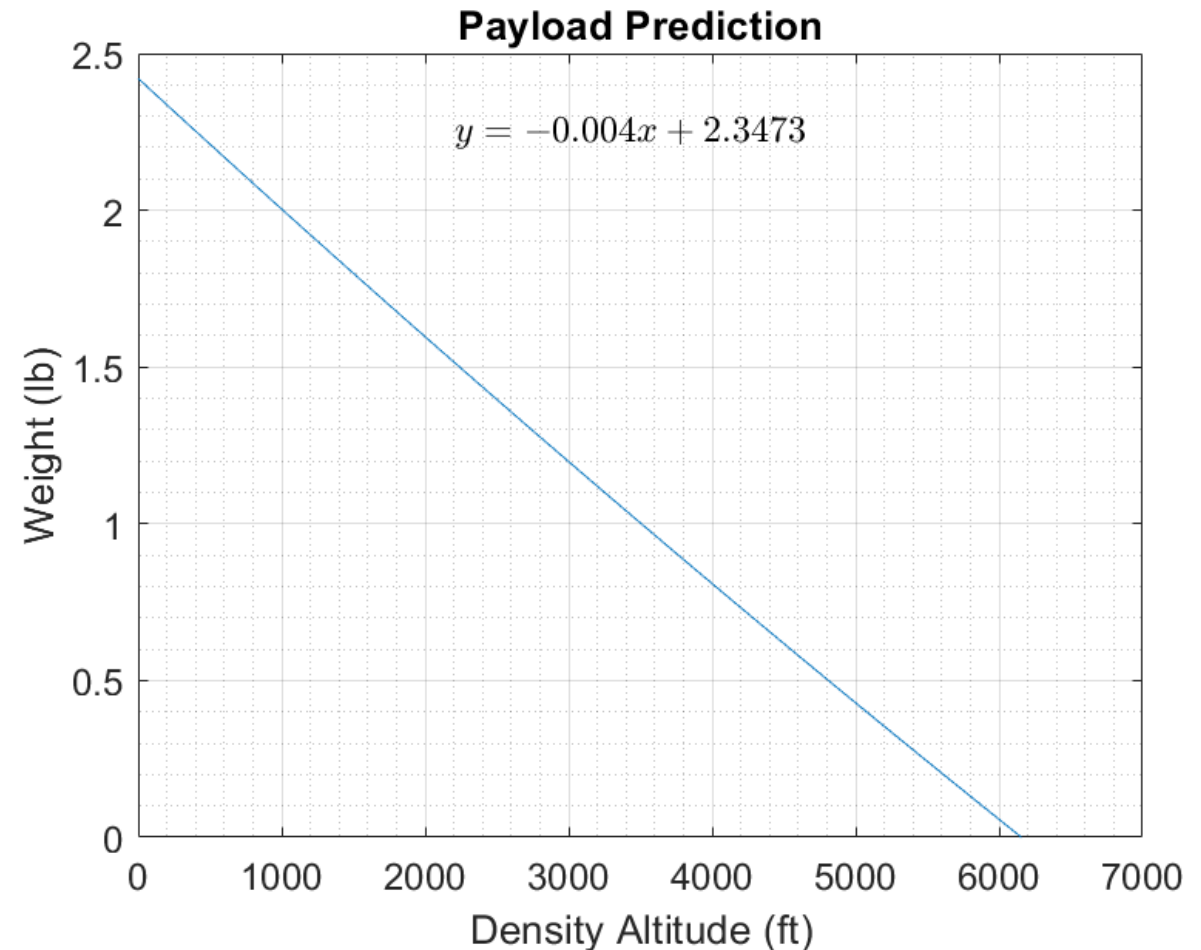
- ✈ Lift (L)
- ✈ Takeoff Velocity ( $V_{TO}$ )
- ✈ Lift coefficient ( $C_L$ )
- ✈ Air density ( $\rho$ )
- ✈ Velocity (v)
- ✈ Wing area (S)
- ✈ Weight (W)

✈ Yields 35.7 ft/s (24.3 mph) takeoff velocity for 15 lb plane



# Performance Analysis: Payload

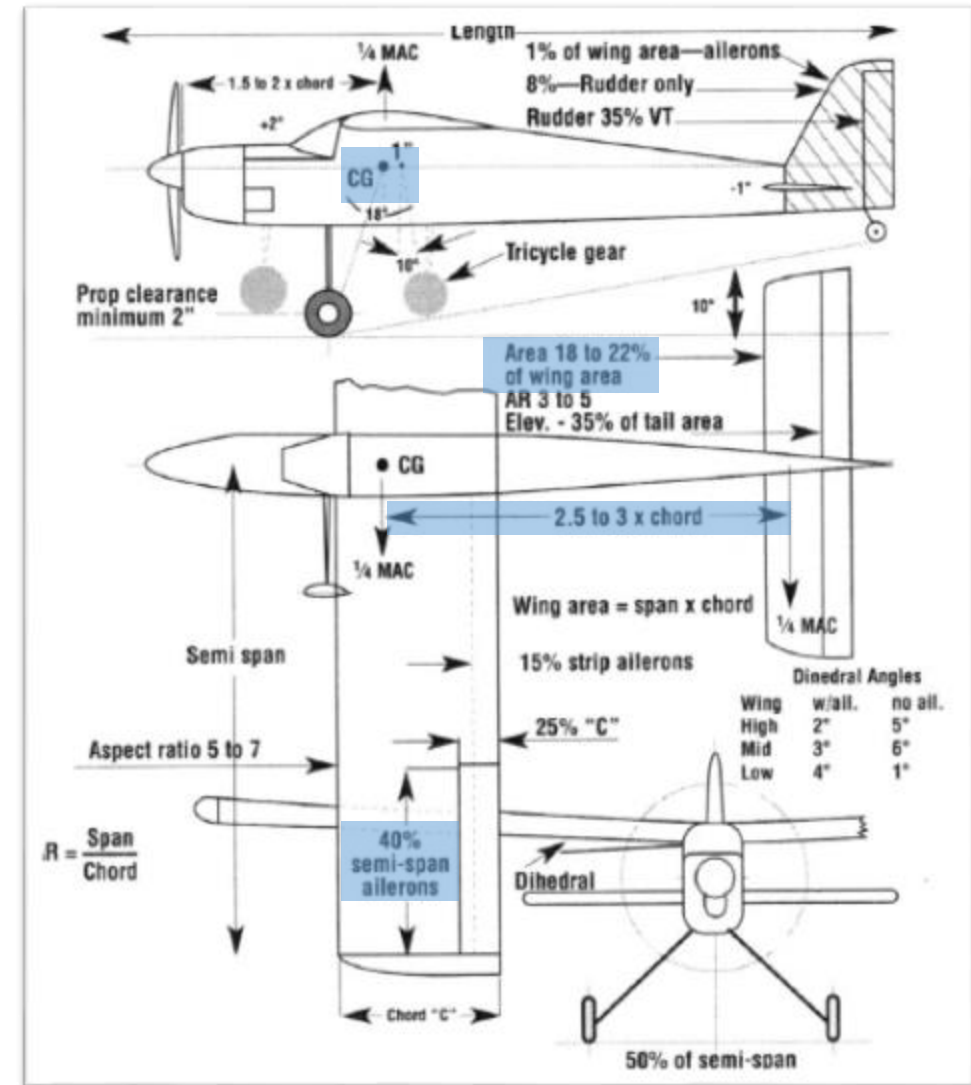
- ✈ Competition scoring based on ability to carry high payload
- ✈ For Lakeland, Florida (100 ft density altitude):
  - ✈ Max payload: 2.34 lbs

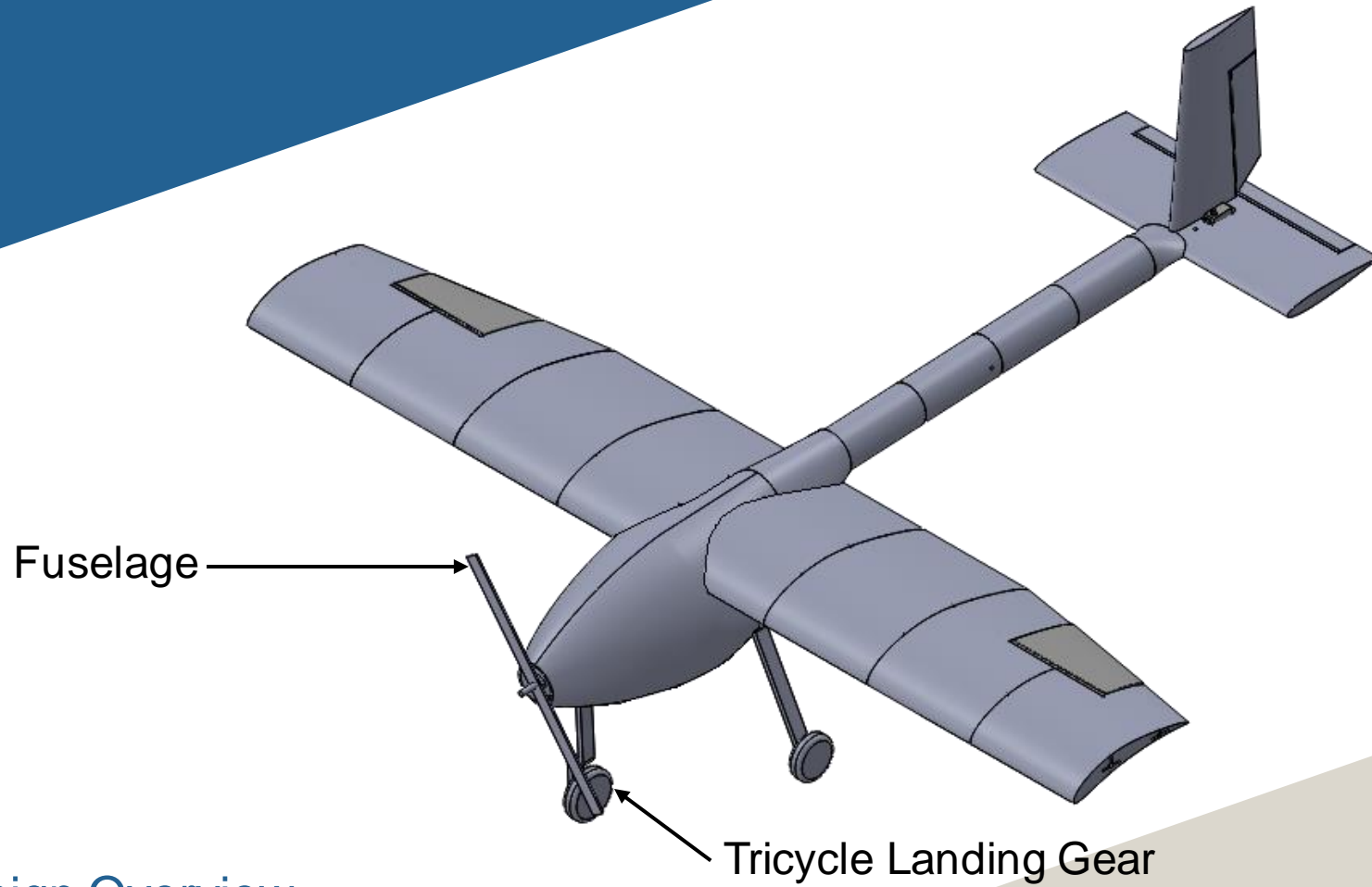




# Airplane Proportions

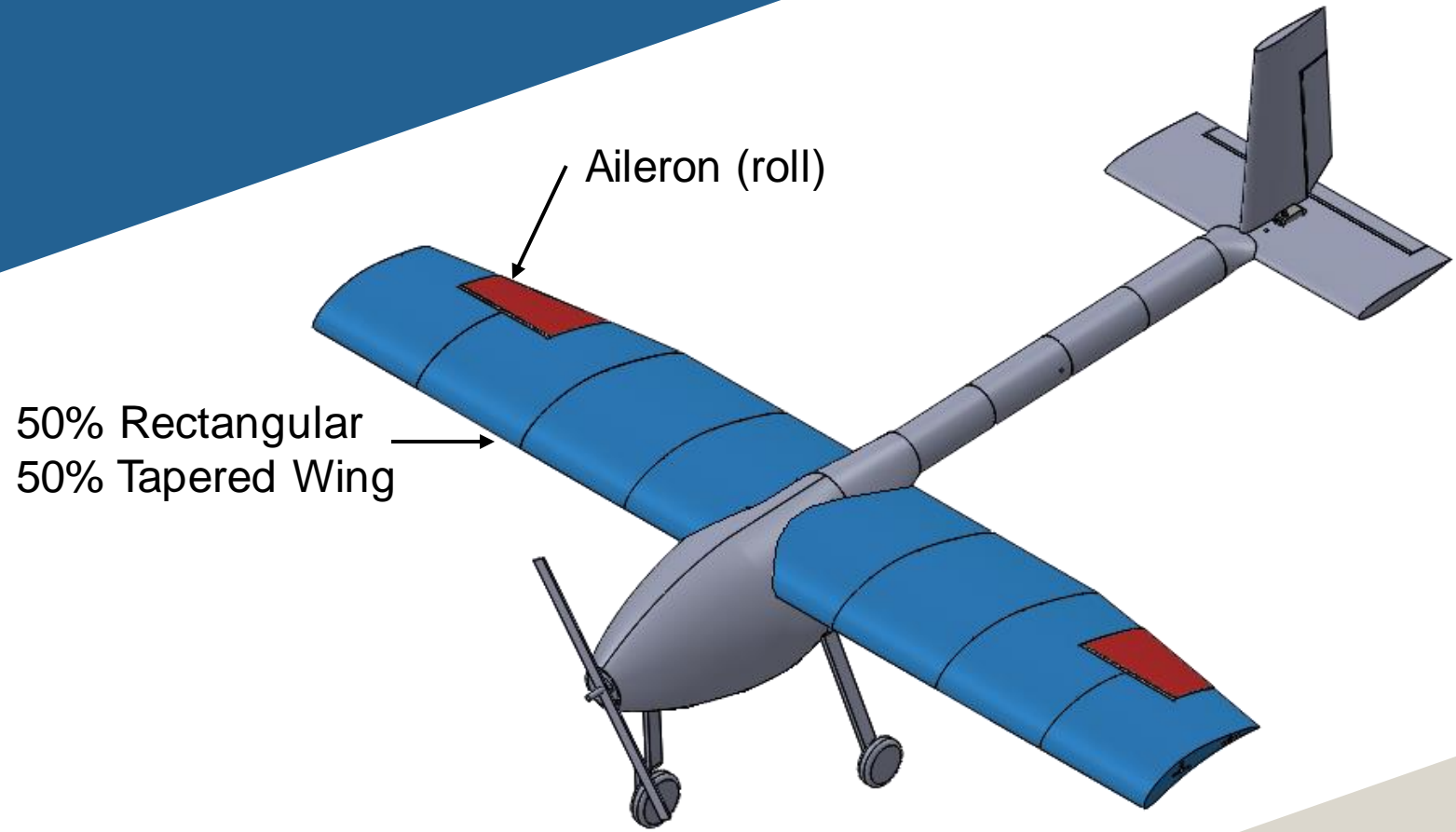
- ✈️ Sized airplane using RC standards
- ✈️ Wing size determines overall dimensions of airplane
- ✈️ Performance sensitivities:
  - ✈️ Length from wing to tail
  - ✈️ Control surface area
  - ✈️ Size of tail
  - ✈️ Landing gear location





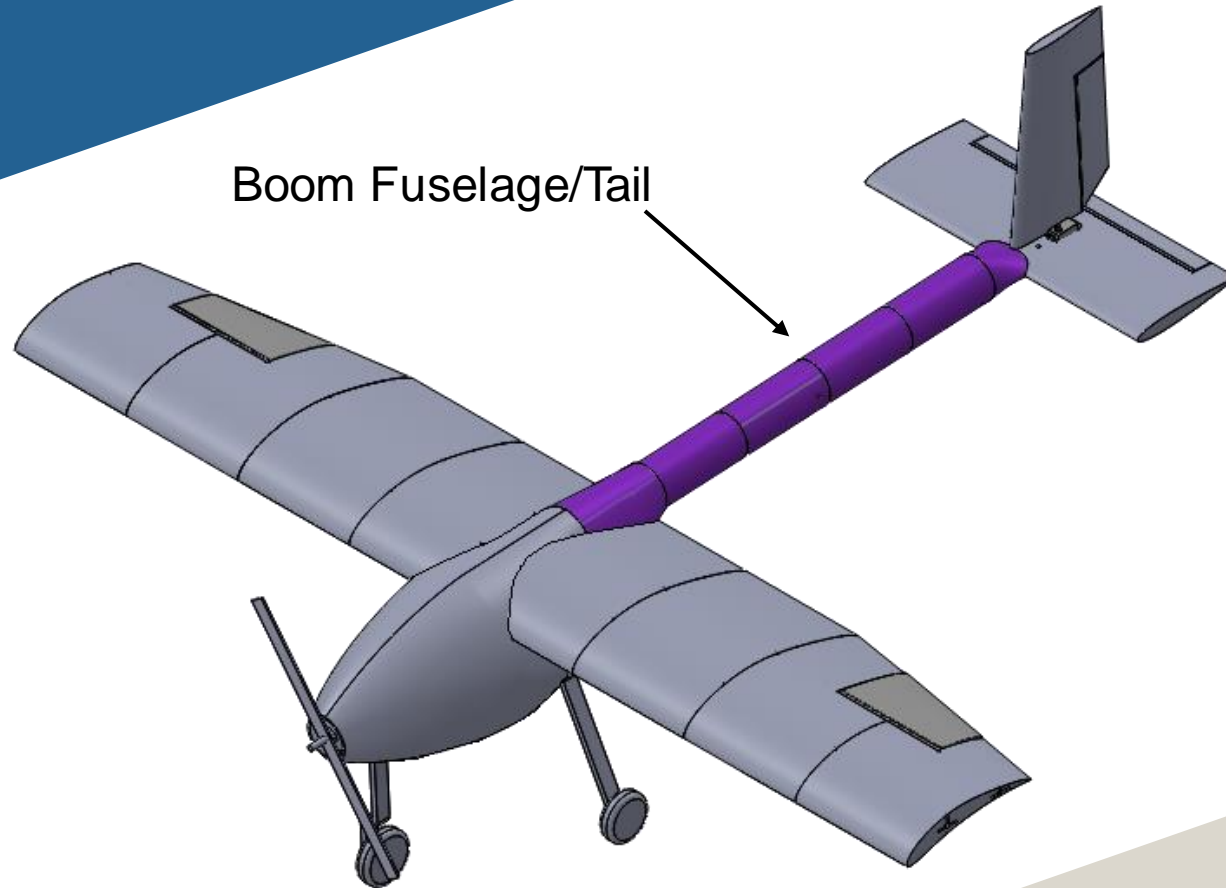
## Final Design Overview

Fuselage and landing gear will not be discussed.



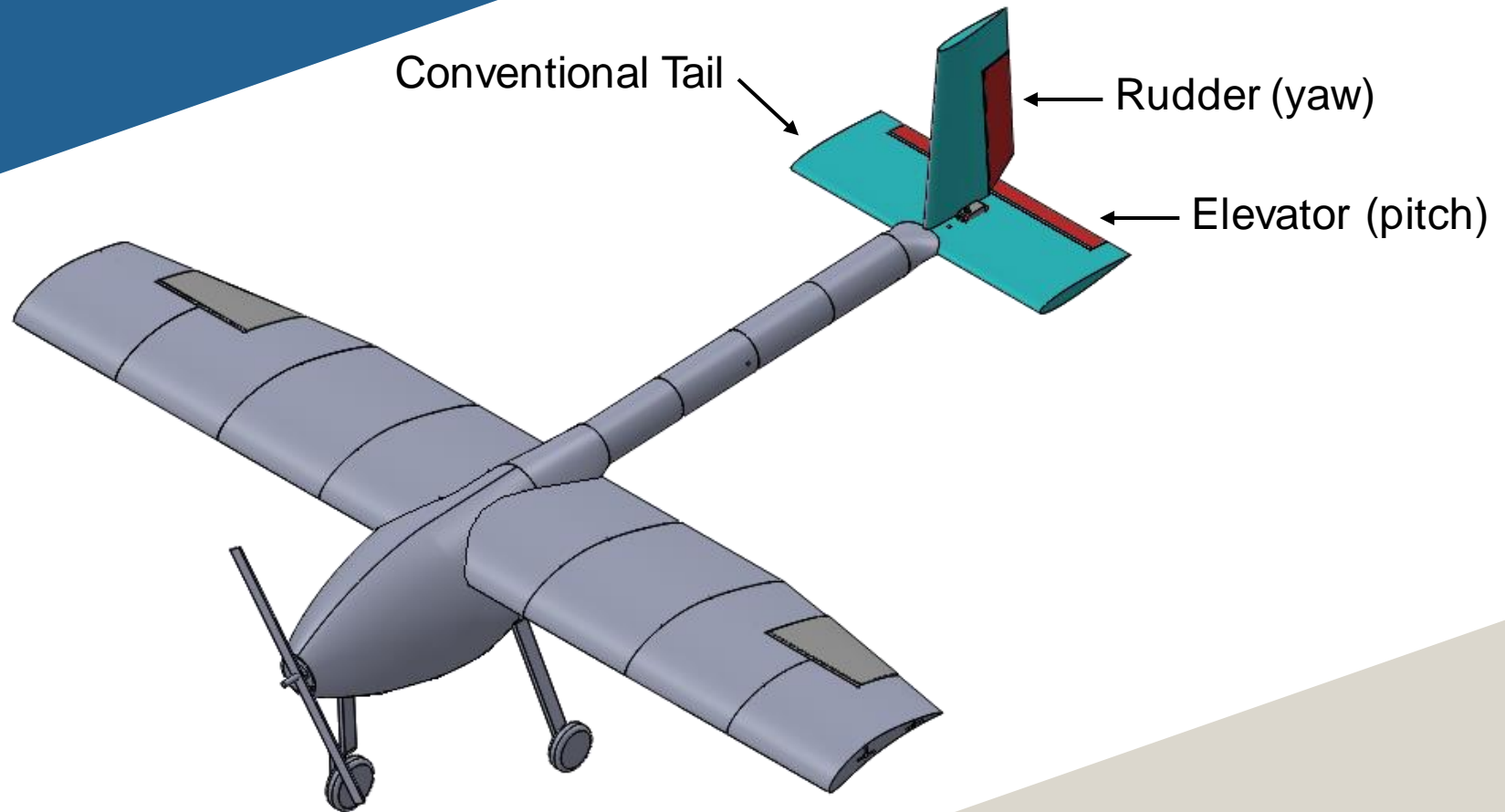
## Final Design Overview

Main wing and control surfaces.



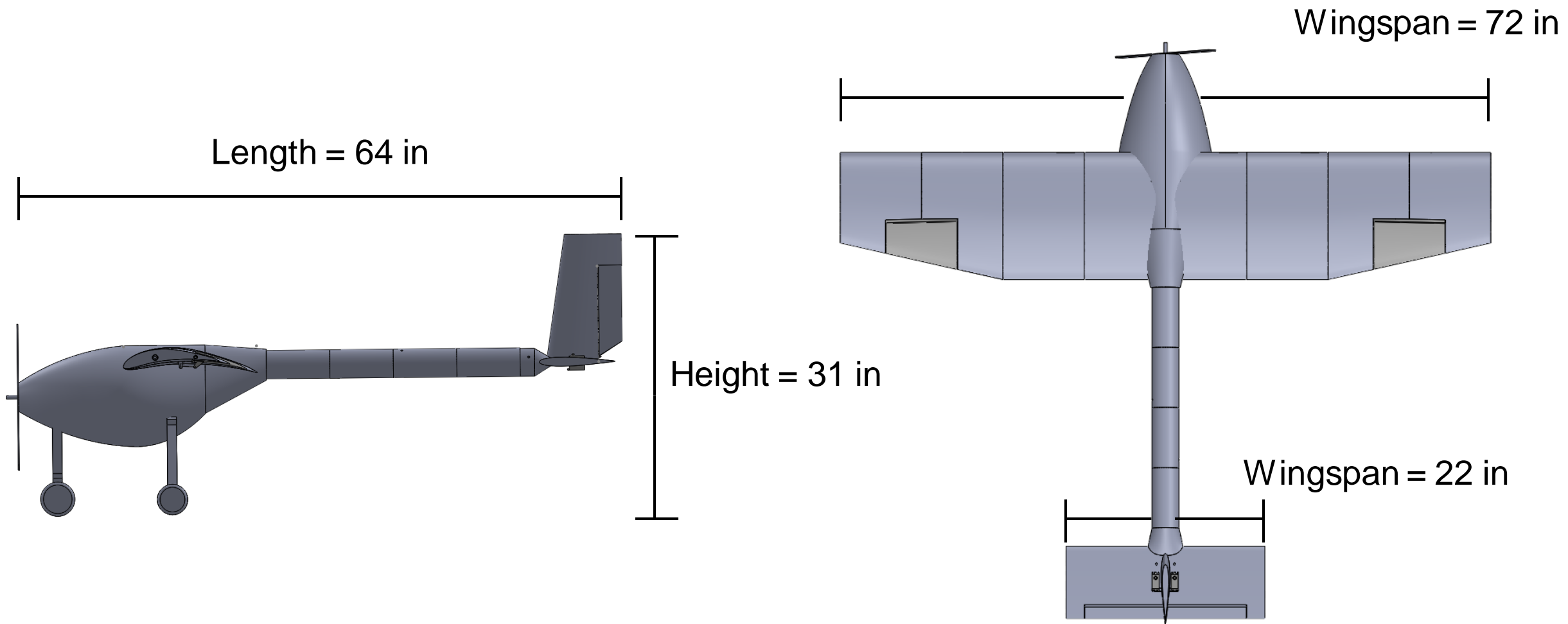
## Final Design Overview

Boom connects fuselage and wing to the tail.



## Final Design Overview

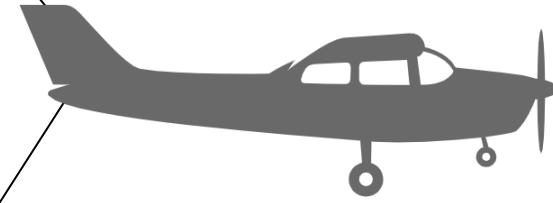
Conventional tail design with vertical and horizontal stabilizer.



## Final Design: Dimensions

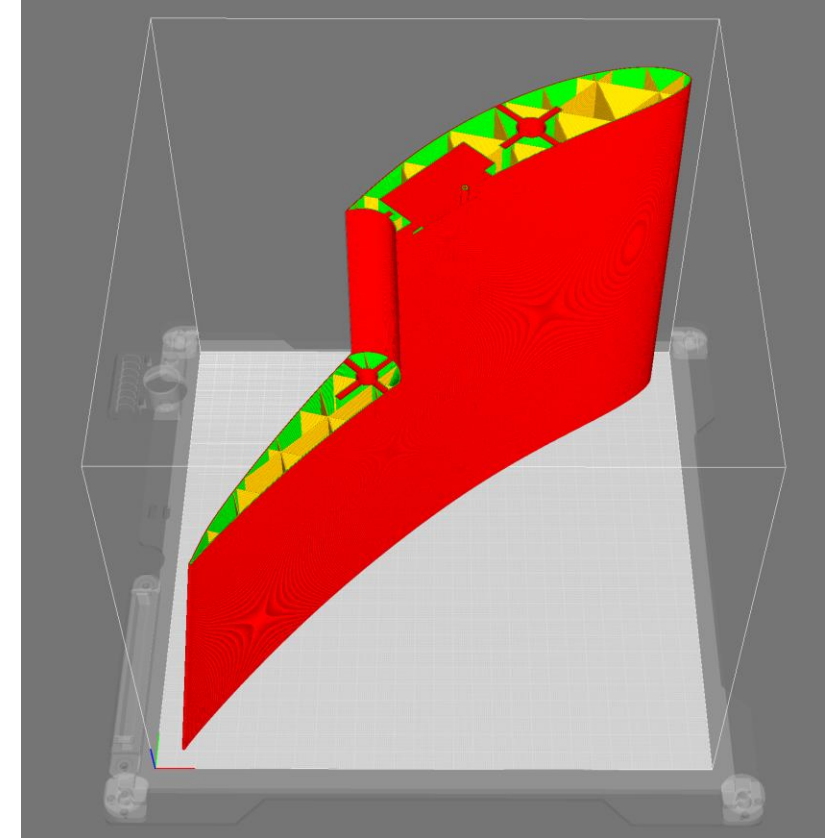
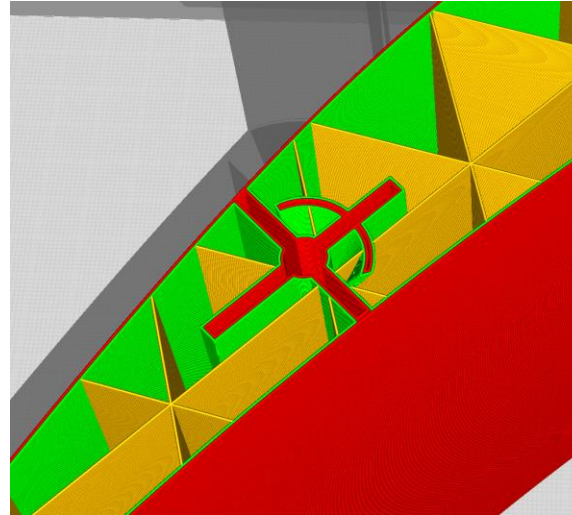


# 3D Printing



## ✈ Designing parts to be 3D printed

- ✈ Orientation
- ✈ Internal structure
- ✈ Custom supports
- ✈ CADing in negative space
- ✈ Looking at slice layer-by-layer



Overview

Mission Requirements

Performance Analysis

Design Overview

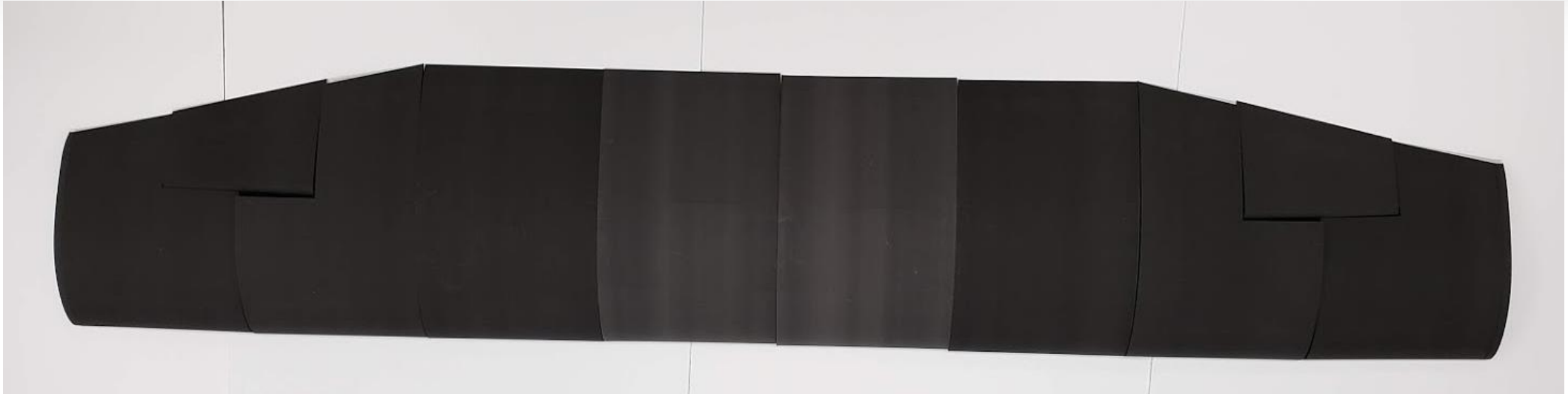
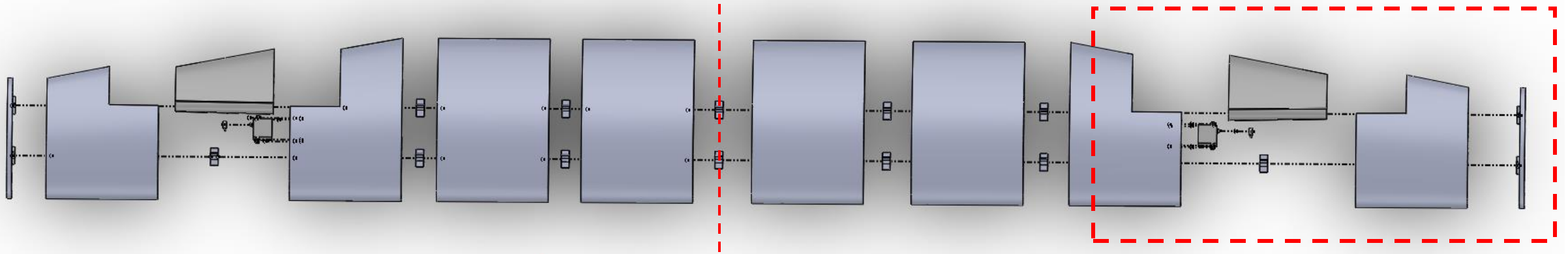
Detailed Design

Future Work

Review

Wing Midpoint

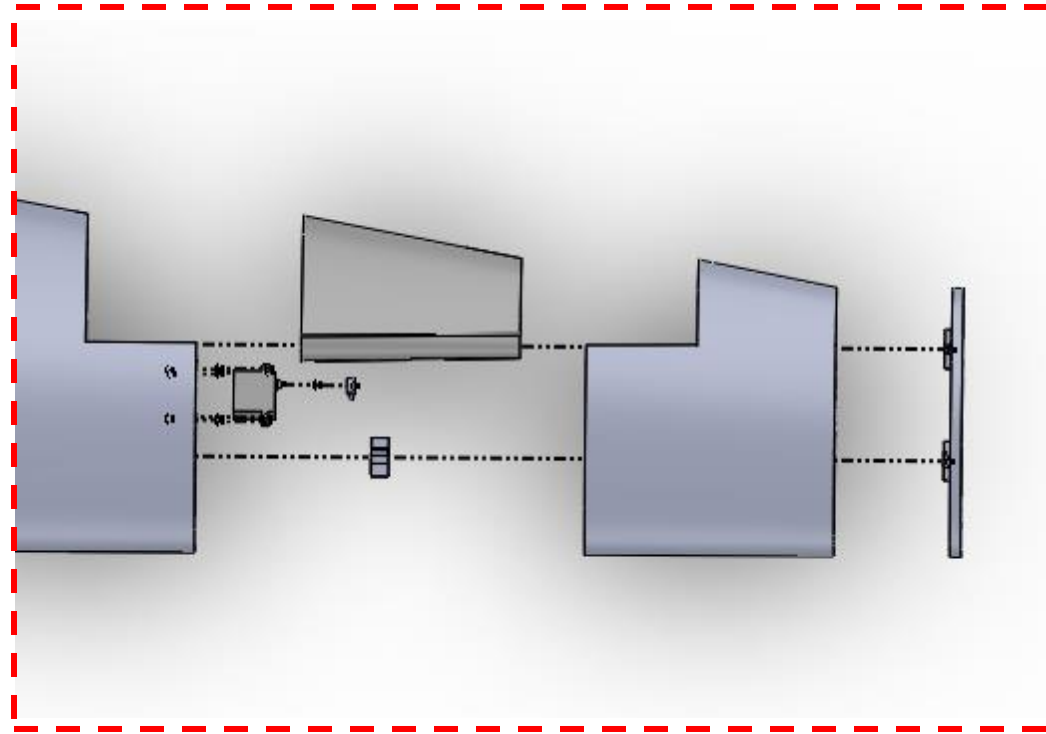
Detail A



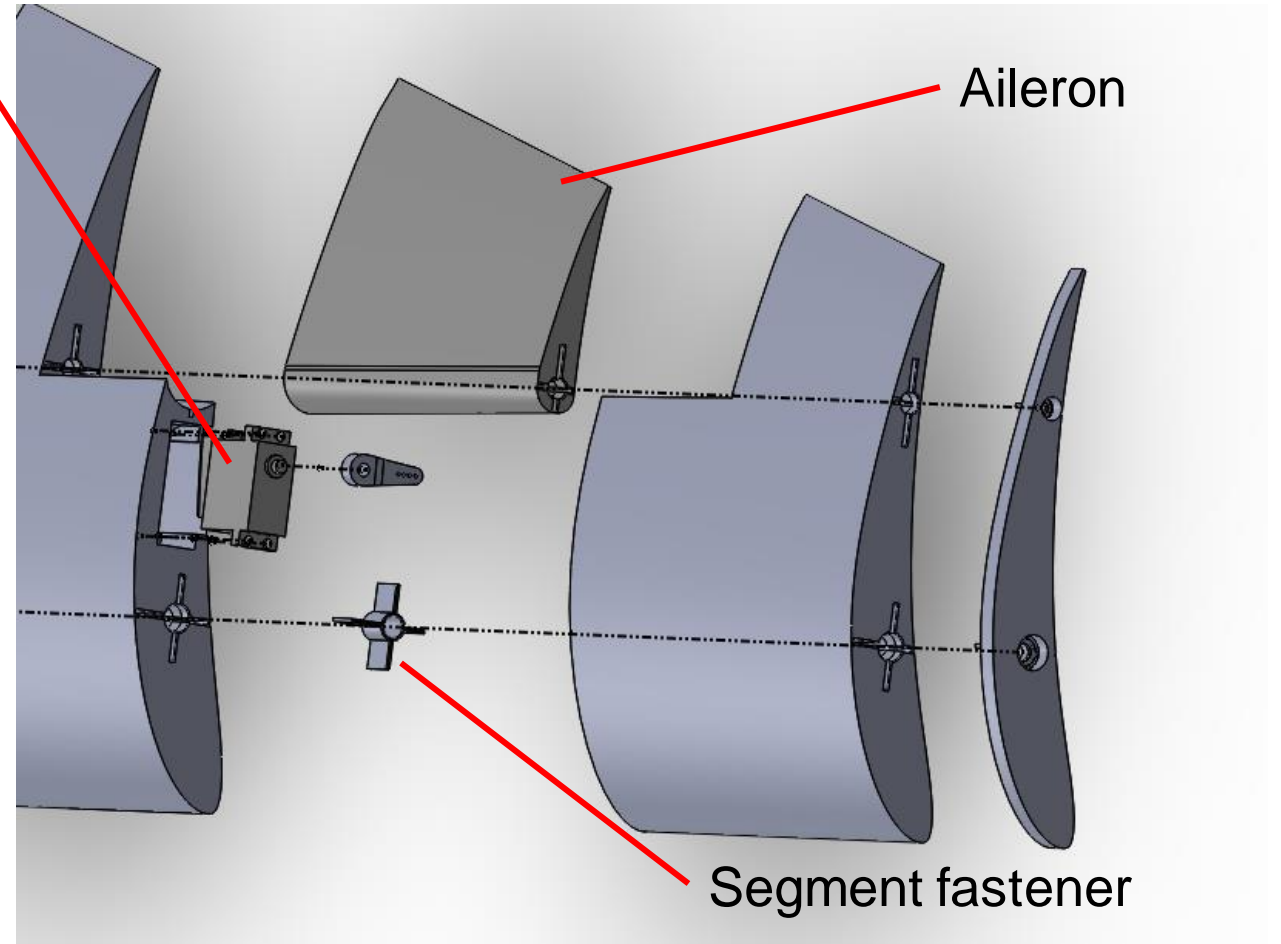
## Detailed Design: Wing Assembly

The wing is comprised of eight segments, each nine inches in length, plus the ailerons and end caps.

## Detail A:



Servo Motor



Aileron

Segment fastener

## Detailed Design: Servo Motor Integration

Servo motor to control ailerons is located inside of the wing, affixed between two segments.

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Vertical Stabilizer

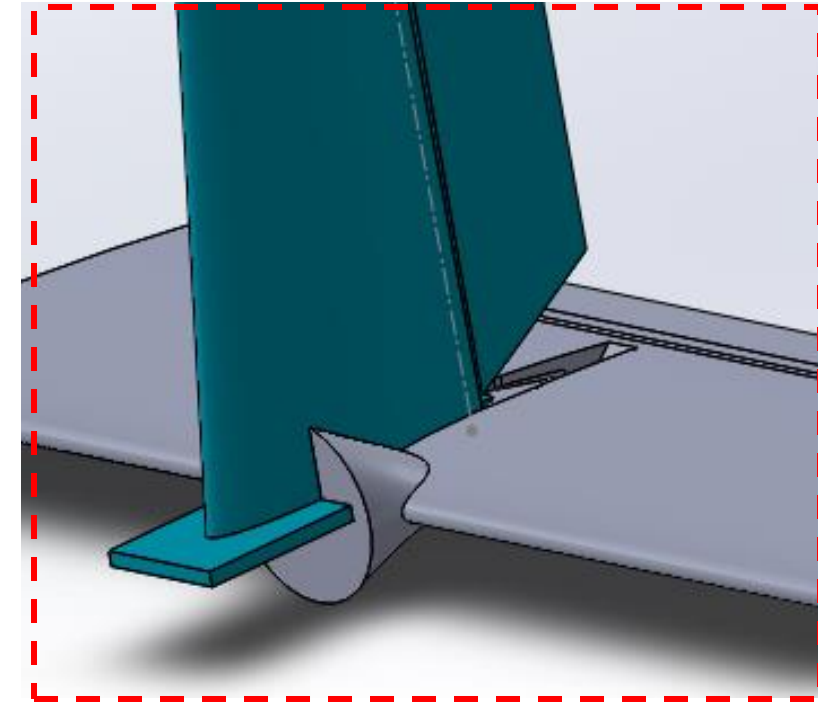
Rudder

Horizontal Stabilizer

Detail B

Elevator

Detail B:



## Detailed Design: Horizontal and Vertical Stabilizer

The vertical stabilizer slides and locks into the horizontal stabilizer.

Overview

Mission Requirements

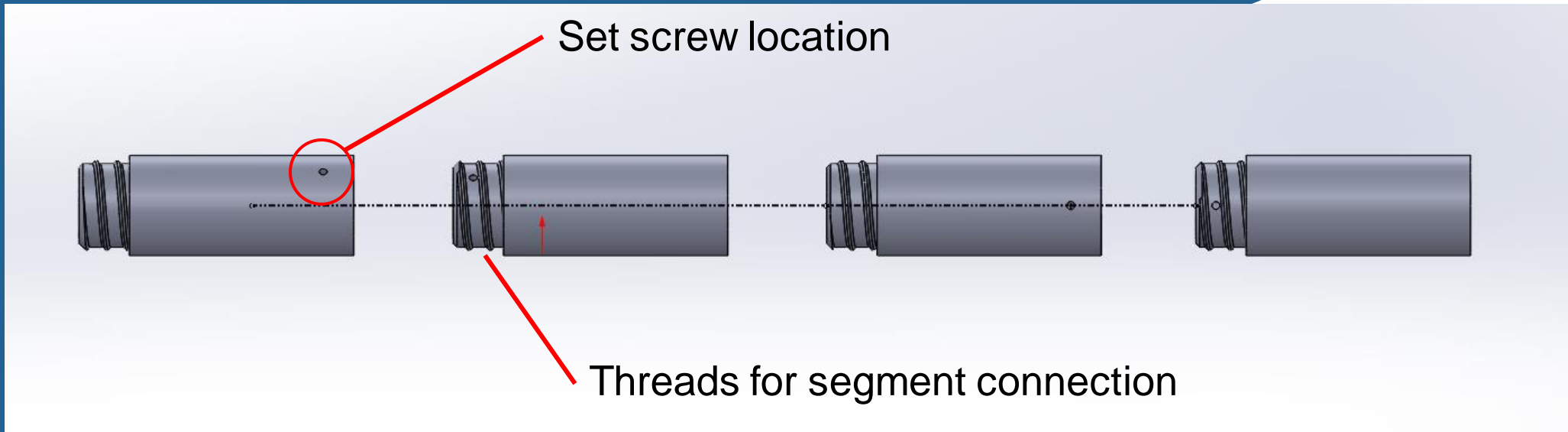
Performance Analysis

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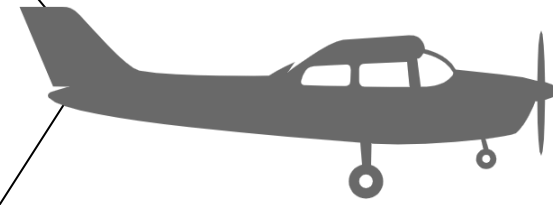


## Detailed Design: Boom Tail

The fuselage connects to the tail using a boom, made of four modular segments that fasten using screw threading

Overview > Mission Requirements > Performance Analysis > Design Overview > **Detailed Design** > Future Work > Review

# Future Work



- ✈ Design mounting points for electronics
- ✈ Design guppy cargo door
- ✈ Create landing gear
- ✈ Confirm center of gravity
- ✈ 3D print remaining parts
- ✈ Initial test flight



Overview

Mission Requirements

Performance Analysis

Design Overview

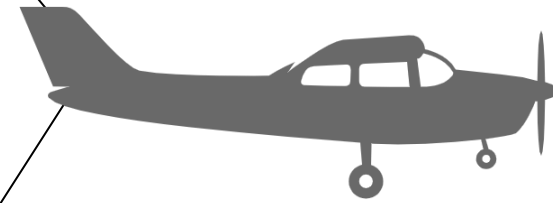
Detailed Design

Future Work

Review

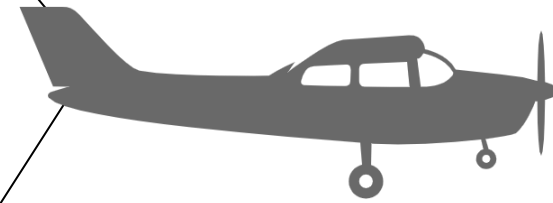


# Most Important Points



1. The goal is to design and manufacture a 3D printed airplane capable of carrying a soccer ball and one pound of steel cargo.
2. Calculations are completed that estimate airplane performance, dimensioning is also complete.
3. Wing, boom tail, and tail is 3D printed. Fuselage and landing gear is next priority.
4. The test flight of airplane will take place February 29<sup>th</sup>.

# References



Anderson, J. D. (2016). *Fundamentals of Aerodynamics* (6th ed.). New York: McGraw-Hill Education.

*Flap (Aeronautics)*. (2019, September 29). Retrieved from Wikipedia: [https://en.wikipedia.org/wiki/Flap\\_\(aeronautics\)](https://en.wikipedia.org/wiki/Flap_(aeronautics))

*Flight: Tail Designs*. (n.d.). Retrieved from what-when-how: <http://what-when-how.com/flight/tail-designs/>

*Fuselage*. (2019, September 14). Retrieved from Wikipedia: <https://en.wikipedia.org/wiki/Fuselage>

Nicolai, L. M., & Carichner, G. (2010). *Fundamentals of Aircraft and Airship Design, Volume I, Aircraft Design*. Reston: AIAA.

Özgen, S. (2015, November). *The Philosophy of Airplane Design*. Retrieved from Middle East Technical University: [http://www.ae.metu.edu.tr/~ae451/landing\\_gear.pdf](http://www.ae.metu.edu.tr/~ae451/landing_gear.pdf)

*RC Airplanes*. (2019). Retrieved from Horizon Hobby: <https://www.horizonhobby.com/category/airplanes>

SAE International. (2020, September 05). *Series Resources 2020 SAE Aero Design Rules*. Retrieved from SAE Aero Design:

<https://www.saeerodesign.com/cdsweb/gen/DocumentResources.aspx>

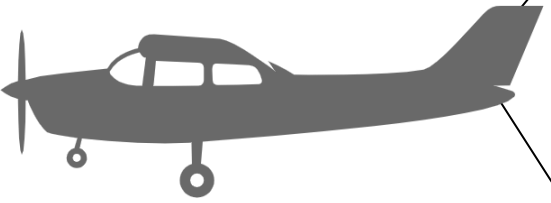
Simons, M. (2015). *Model Aircraft Aerodynamics*. Englad: Special Interest Model Books Ltd.

Staples, G. (2013, July 16). *Propeller Static & Dynamic Thrust Calculation*. Retrieved September 27, 2019, from ElectricRCaircraftGuy:

<https://www.electricrcaircraftguy.com/2013/09/propeller-static-dynamic-thrust-equation.html>

*Wing Configuration*. (2019, September 9). Retrieved from Wikipedia: [https://en.wikipedia.org/wiki/Wing\\_configuration](https://en.wikipedia.org/wiki/Wing_configuration)





# Questions?

# Backup Slides

Functional Decomp

Concept Selection

Concept  
Generation

Detailed Concepts

Detailed Math

Bill of Materials

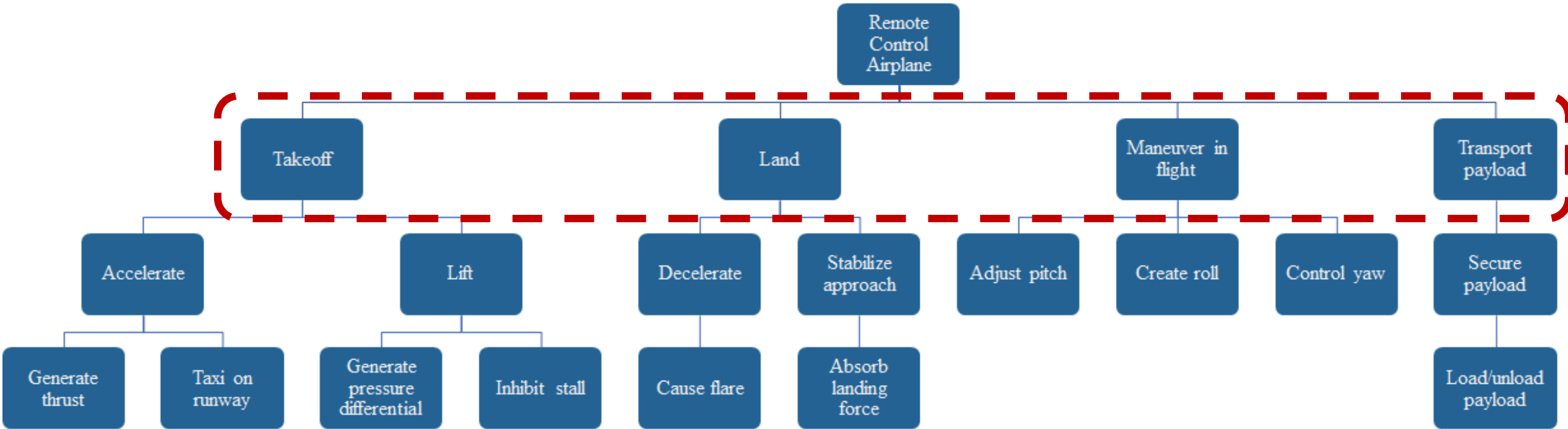
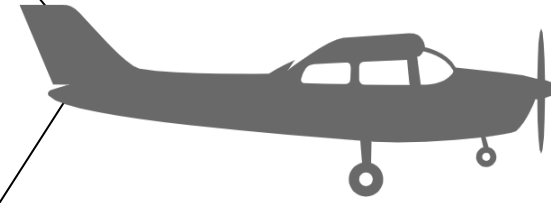
Targets and  
Metrics

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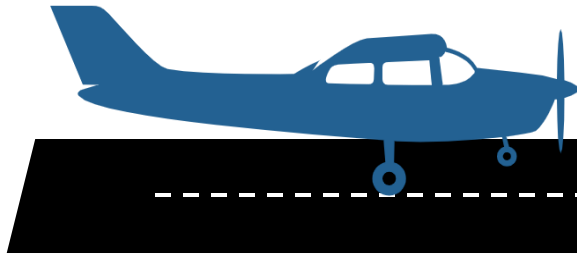
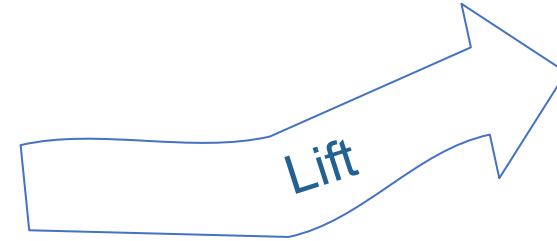
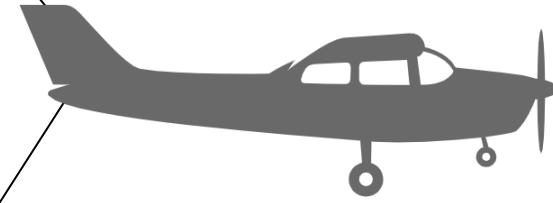
# Functional Decomp Backup

# Functional Decomposition

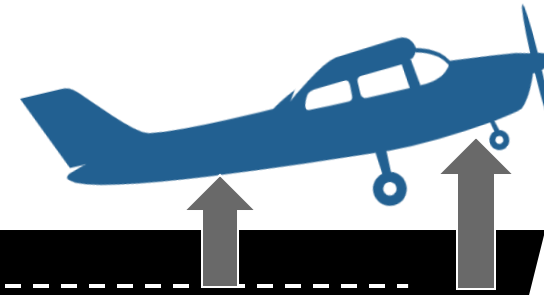




# Functional Decomposition: Takeoff

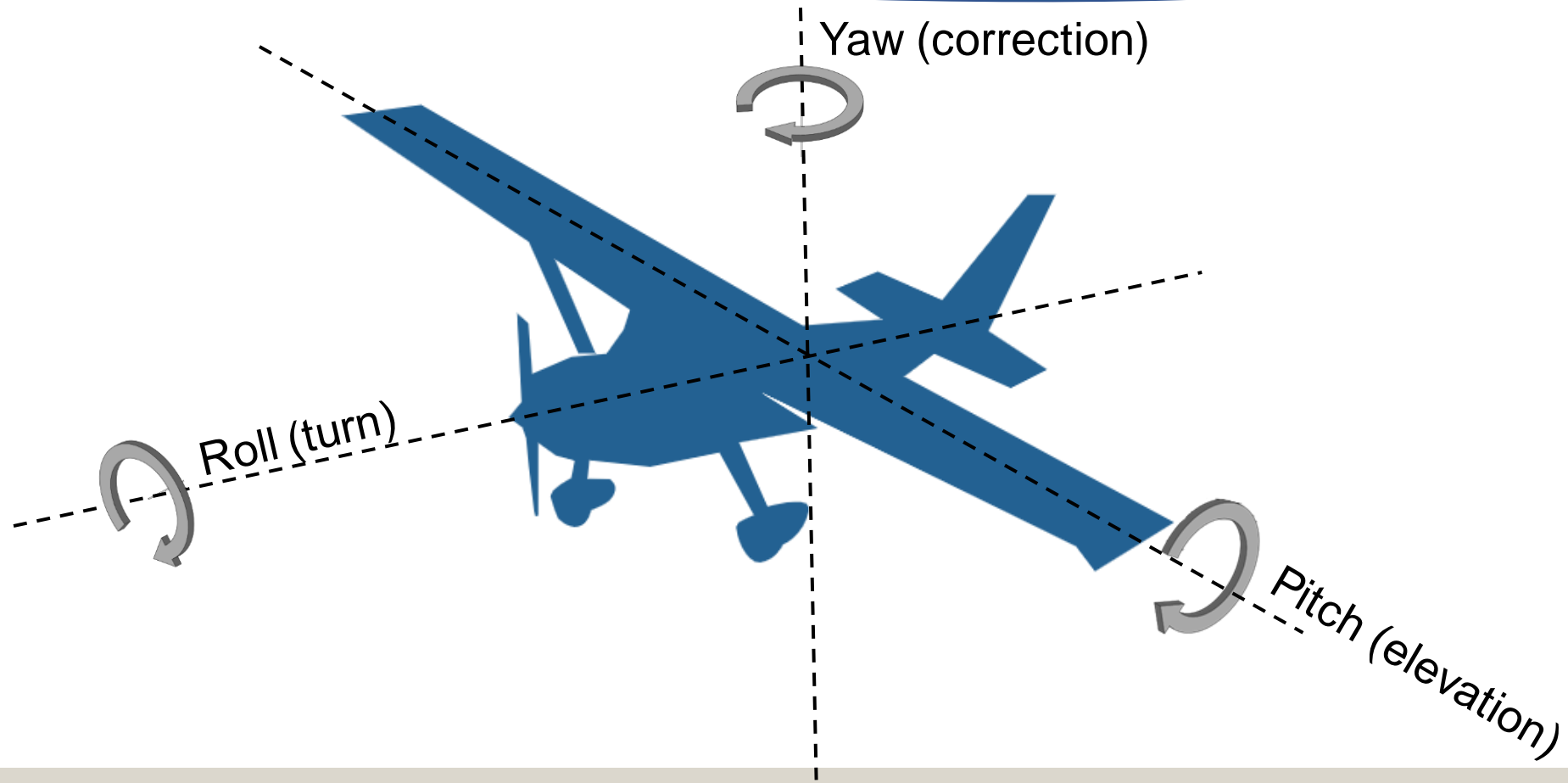
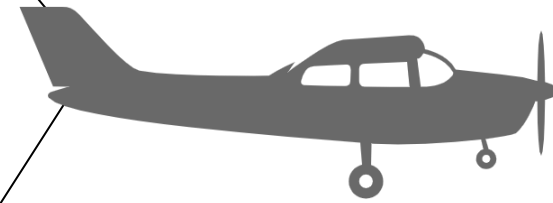


Generate Thrust  
Taxi on Runway

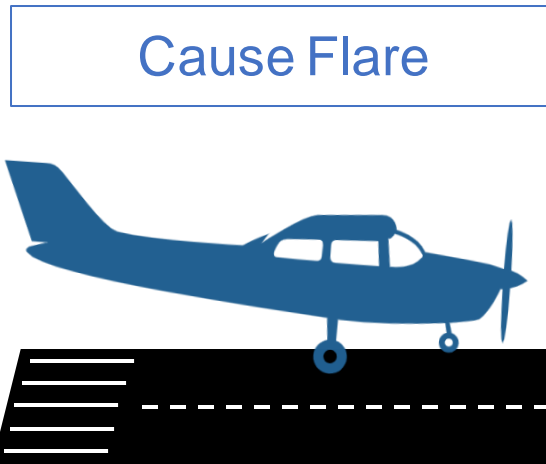
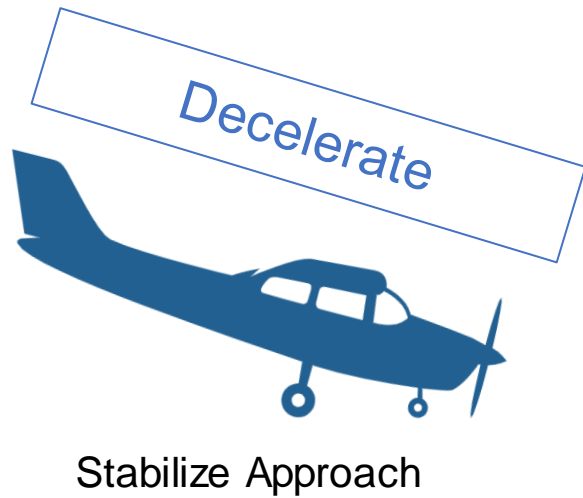
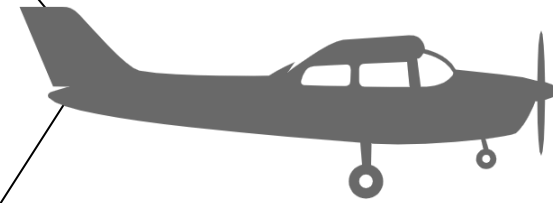


Generate Pressure Differential  
Inhibit Stall

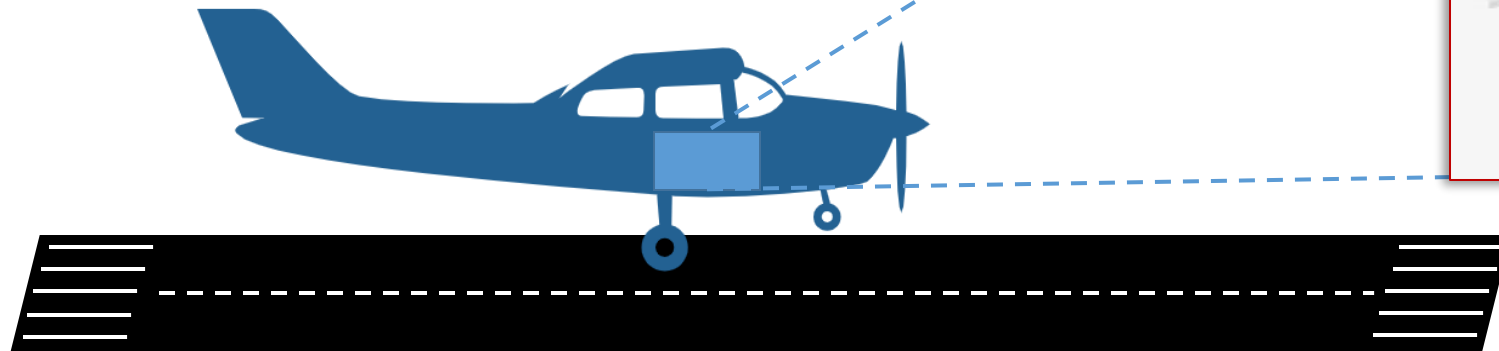
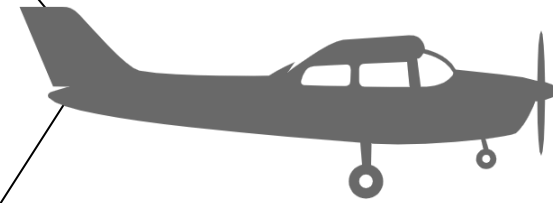
# Functional Decomposition: Maneuver in Flight



# Functional Decomposition: Land



# Functional Decomposition: Transport Payload



Secure Payload  
Load/unload Payload

# Concept Selection Backup

# House of Quality

| Relationships |   |
|---------------|---|
| Strong        | ● |
| Moderate      | ○ |
| Weak          | ▽ |

| Direction of Improvement |   |
|--------------------------|---|
| Maximize                 | ▲ |
| Target                   | ◇ |
| Minimize                 | ▼ |

| Row #                       | Weight Chart | Relative Weight | Customer Importance | Customer Requirements (Explicit and Implicit) | Column #                 | 1            | 2          | 3          | 4                                 | 5                           | 6             | 7                          | 8          | 9                        | 10 |
|-----------------------------|--------------|-----------------|---------------------|---|--------------------------|--------------|------------|------------|-----------------------------------|-----------------------------|---------------|----------------------------|------------|--------------------------|----|
|                             |              |                 |                     |   | Direction of Improvement | ▼            | ▲          | ▲          | ◇                                 | ▲                           | ◇             | ▼                          | ▼          | ▼                        | ▼  |
|                             |              |                 |                     |   | Weight(Lb)               | Thrust (lbf) | Lift (lbf) | Drag (lbf) | Acceleration (ft/s <sup>2</sup> ) | Weight Distribution (lbf/x) | Wingspan (in) | Time to Unload Cargo (sec) | Price (\$) | Manufacturing Time (sec) |    |
| 1                           | ■            | 16%             | 10                  | Fly   | ●                        | ●            | ●          | ●          | ●                                 | ▽                           | ●             | ▽                          | ▽          | ○                        |    |
| 2                           | ■            | 10%             | 6                   | Carry Payload                                 | ●                        | ●            | ●          | ●          | ●                                 | ●                           | ○             | ○                          | ▽          | ○                        |    |
| 3                           | ■            | 8%              | 5                   | Takeoff Distance                              | ●                        | ●            | ●          | ●          | ●                                 | ▽                           | ●             | ○                          | ○          | ●                        |    |
| 4                           | ■            | 8%              | 5                   | Landing                                       | ●                        | ●            | ○          | ●          | ●                                 | ▽                           | ▽             | ▽                          | ○          | ●                        |    |
| 5                           | ■            | 8%              | 5                   | Cost  | ▽                        | ●            | ▽          | ▽          | ○                                 | ○                           | ○             | ▽                          | ●          | ○                        |    |
| 6                           | ■            | 15%             | 9                   | 3-D Printed                                   | ●                        | ▽            | ▽          | ▽          | ●                                 | ○                           | ●             | ●                          | ●          | ●                        |    |
| 7                           | ■            | 10%             | 6                   | Flight Stability                              | ○                        | ▽            | ●          | ●          | ○                                 | ●                           | ●             | ●                          | ▽          | ○                        |    |
| 8                           | ■            | 10%             | 6                   | Payload Accessibility                         | ▽                        | ▽            | ▽          | ●          | ○                                 | ●                           | ▽             | ●                          | ▽          | ●                        |    |
| 9                           | ■            | 16%             | 10                  | Safety  | ●                        | ●            | ▽          | ▽          | ○                                 | ○                           | ○             | ●                          | ○          | ▽                        |    |
| Technical Importance Rating |              |                 |                     |   | 700                      | 629          | 464.5      | 590.3      | 638.7                             | 409.7                       | 554.8         | 535.5                      | 345.2      | 509.7                    |    |
| Relative Weight             |              |                 |                     |   | 13%                      | 12%          | 9%         | 11%        | 12%                               | 8%                          | 10%           | 10%                        | 6%         | 9%                       |    |
| Weight Chart                |              |                 |                     |   | ■                        | ■            | ■          | ■          | ■                                 | ■                           | ■             | ■                          | ■          | ■                        |    |

*Initial Pugh Selection Chart*

|                           |                  | <b>Concepts</b> |          |          |          |          |          |          |
|---------------------------|------------------|-----------------|----------|----------|----------|----------|----------|----------|
| <b>Selection Criteria</b> | <b>Concept 7</b> | <b>1</b>        | <b>2</b> | <b>3</b> | <b>4</b> | <b>5</b> | <b>6</b> | <b>8</b> |
| Weight                    | <b>DATUM</b>     | +               | -        | -        | S        | S        | +        | +        |
| Drag                      |                  | +               | S        | -        | S        | -        | S        | +        |
| Wingspan                  |                  | +               | +        | +        | S        | S        | +        | +        |
| Time to Unload            |                  | +               | -        | S        | S        | -        | +        | -        |
| Manufacturing Time        |                  | -               | -        | +        | S        | S        | +        | -        |
| Cost                      |                  | -               | +        | S        | +        | +        | S        | +        |
| <b># of pluses</b>        |                  | <b>4</b>        | <b>2</b> | <b>2</b> | <b>1</b> | <b>1</b> | <b>4</b> | <b>4</b> |
| <b># of minuses</b>       |                  | <b>2</b>        | <b>3</b> | <b>2</b> | <b>0</b> | <b>2</b> | <b>0</b> | <b>2</b> |

**Pugh Chart 1**

Eliminated Concept 2 & 5. Concept 6 becomes new datum.



## Final Pugh Selection Chart

| Selection Criteria  | Concept 6 | Concepts |          |          |          |
|---------------------|-----------|----------|----------|----------|----------|
|                     |           | 1        | 3        | 4        | 8        |
| Weight              | DATUM     | +        | -        | S        | +        |
| Drag                |           | +        | -        | -        | S        |
| Wingspan            |           | S        | S        | -        | S        |
| Time to Unload      |           | S        | -        | +        | -        |
| Manufacturing Time  |           | -        | S        | +        | -        |
| Cost                |           | -        | S        | S        | +        |
| <b># of pluses</b>  |           | <b>2</b> | <b>0</b> | <b>2</b> | <b>2</b> |
| <b># of minuses</b> |           | <b>2</b> | <b>3</b> | <b>2</b> | <b>2</b> |

## Pugh Chart 2

Eliminated Concept 3 & 8. Concept 1, 4, and 6 transfer to AHP.



*Criteria 1 – Drag*

|                       | <b>Weighted Sum Vector</b> | <b>Criteria Weight</b>  | <b>Consistency (Con)</b> |
|-----------------------|----------------------------|-------------------------|--------------------------|
| <b>Concept 1</b>      | 1.460                      | 0.480                   | 3.044                    |
| <b>Concept 4</b>      | 0.346                      | 0.115                   | 3.010                    |
| <b>Concept 6</b>      | 1.230                      | 0.405                   | 3.033                    |
| <b>Avg Con: 3.029</b> | <b>Con Index: 0.015</b>    | <b>Con Ratio: 0.028</b> | <b>Consistent?: Yes</b>  |

*Criteria 2 – Weight*

|                       | <b>Weighted Sum Vector</b> | <b>Criteria Weight</b>  | <b>Consistency (Con)</b> |
|-----------------------|----------------------------|-------------------------|--------------------------|
| <b>Concept 1</b>      | 1.230                      | 0.405                   | 3.033                    |
| <b>Concept 4</b>      | 0.346                      | 0.115                   | 3.010                    |
| <b>Concept 6</b>      | 1.460                      | 0.480                   | 3.044                    |
| <b>Avg Con: 3.029</b> | <b>Con Index: 0.015</b>    | <b>Con Ratio: 0.028</b> | <b>Consistent?: Yes</b>  |

## Analytic Hierarchy Process

Overview of drag and weight criteria

### Criteria 3 – Wingspan

|                       | <b>Weighted Sum Vector</b> | <b>Criteria Weight</b>  | <b>Consistency (Con)</b> |
|-----------------------|----------------------------|-------------------------|--------------------------|
| <b>Concept 1</b>      | 0.790                      | 0.260                   | 3.033                    |
| <b>Concept 4</b>      | 0.320                      | 0.106                   | 3.011                    |
| <b>Concept 6</b>      | 1.946                      | 0.633                   | 3.072                    |
| <b>Avg Con: 3.039</b> | <b>Con Index: 0.019</b>    | <b>Con Ratio: 0.037</b> | <b>Consistent?: Yes</b>  |

### Criteria 4 – Time to Unload

|                       | <b>Weighted Sum Vector</b> | <b>Criteria Weight</b>  | <b>Consistency (Con)</b> |
|-----------------------|----------------------------|-------------------------|--------------------------|
| <b>Concept 1</b>      | 0.790                      | 0.260                   | 3.033                    |
| <b>Concept 4</b>      | 0.320                      | 0.106                   | 3.011                    |
| <b>Concept 6</b>      | 1.946                      | 0.633                   | 3.072                    |
| <b>Avg Con: 3.039</b> | <b>Con Index: 0.019</b>    | <b>Con Ratio: 0.037</b> | <b>Consistent?: Yes</b>  |

## Analytic Hierarchy Process

Overview of wingspan and time to unload criteria

### Criteria 5 – Manufacturing Time

|                       | <b>Weighted Sum Vector</b> | <b>Criteria Weight</b>  | <b>Consistency (Con)</b> |
|-----------------------|----------------------------|-------------------------|--------------------------|
| <b>Concept 1</b>      | 0.273                      | 0.091                   | 3.000                    |
| <b>Concept 4</b>      | 1.364                      | 0.455                   | 3.000                    |
| <b>Concept 6</b>      | 1.364                      | 0.455                   | 3.000                    |
| <b>Avg Con: 3.000</b> | <b>Con Index: 0.000</b>    | <b>Con Ratio: 0.000</b> | <b>Consistent?: Yes</b>  |

### Criteria 6 – Cost

|                       | <b>Weighted Sum Vector</b> | <b>Criteria Weight</b>  | <b>Consistency (Con)</b> |
|-----------------------|----------------------------|-------------------------|--------------------------|
| <b>Concept 1</b>      | 1.853                      | 0.574                   | 3.230                    |
| <b>Concept 4</b>      | 0.427                      | 0.140                   | 3.049                    |
| <b>Concept 6</b>      | 0.897                      | 0.286                   | 3.133                    |
| <b>Avg Con: 3.137</b> | <b>Con Index: 0.069</b>    | <b>Con Ratio: 0.132</b> | <b>Consistent?: No</b>   |

## Analytic Hierarchy Process

Overview of manufacturing time and cost criteria

*AHP Criteria Weights and Consistency Check*

|                           | <b>Weighted Sum Vector</b> | <b>Criteria Weight</b>  | <b>Consistency (Con)</b> |
|---------------------------|----------------------------|-------------------------|--------------------------|
| <b>Drag</b>               | 2.840                      | 0.369                   | 7.697                    |
| <b>Wingspan</b>           | 1.387                      | 0.212                   | 6.554                    |
| <b>Time to Unload</b>     | 0.607                      | 0.097                   | 6.268                    |
| <b>Weight</b>             | 1.044                      | 0.156                   | 6.711                    |
| <b>Manufacturing Time</b> | 0.962                      | 0.143                   | 6.739                    |
| <b>Cost</b>               | 0.159                      | 0.024                   | 6.591                    |
| <b>Avg Con: 6.760</b>     | <b>Con Index: 0.152</b>    | <b>Con Ratio: 0.122</b> | <b>Consistent?: No</b>   |

## Analytic Hierarchy Process

Overview of criteria weights

| [Final Rating Matrix] <sup>T</sup> |           |           |           |
|------------------------------------|-----------|-----------|-----------|
| Selection Criteria                 | Concept 1 | Concept 4 | Concept 6 |
| Drag                               | 0.480     | 0.115     | 0.405     |
| Weight                             | 0.405     | 0.115     | 0.480     |
| Wingspan                           | 0.260     | 0.106     | 0.633     |
| Time to Unload                     | 0.260     | 0.106     | 0.633     |
| Manufacturing Time                 | 0.091     | 0.455     | 0.455     |
| Cost                               | 0.574     | 0.140     | 0.286     |

X

| Criteria Weights {W} |        |
|----------------------|--------|
|                      | Weight |
| Drag                 | 0.369  |
| Weight               | 0.212  |
| Wingspan             | 0.097  |
| Time to Unload       | 0.156  |
| Manufacturing Time   | 0.143  |
| Cost                 | 0.024  |

=

|           | Alternative Value |
|-----------|-------------------|
| Concept 1 | 0.355             |
| Concept 4 | 0.162             |
| Concept 6 | 0.483             |

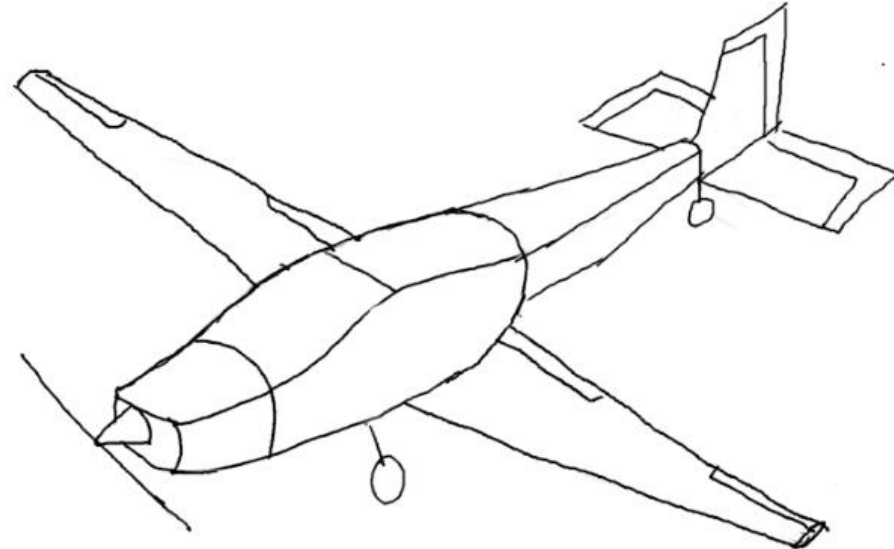
## Analytic Hierarchy Process

Overview of final selection matrix

# Selected Concept: Concept Six

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- ✈ Tapered Wings
- ✈ Low wing location
- ✈ Tricycle with tail wheel
- ✈ LW PLA
- ✈ Conventional tail



# Concept Generation Backup

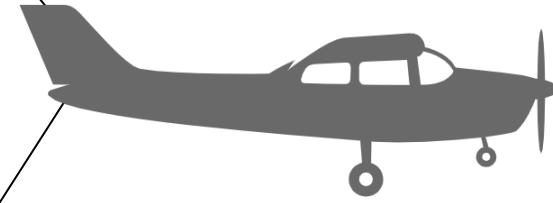
| Modular Connections       | 3D Materials | Propeller Size | Propeller Pitch | Number of Blades | Landing Gear              | Landing Gear Mechanism | Landing Gear Suspension | Wings       | Wing Location | Wing Orientation     | Aileron/Flaps         | Motor          | Fuselage                      | Electronics additions    | Battery               | Tail                |
|---------------------------|--------------|----------------|-----------------|------------------|---------------------------|------------------------|-------------------------|-------------|---------------|----------------------|-----------------------|----------------|-------------------------------|--------------------------|-----------------------|---------------------|
| Compression               | PLA          | Large Prop     | Large Pitch     | 2 Blade          | Tricycle with Front Wheel | Fixed                  | Fixed                   | Elliptical  | Low Wing      | Uniform Leading Edge | Plain                 | Low kv Rating  | Flying boat                   | Speed Densor             | High Battery Capacity | Conventional        |
| Formfit                   | ABS          | Small Prop     | Small Pitch     | 3 Blade          | Tricycle with Tail-Wheel  | Retractable            | 3D Printed Flexible     | Tapered     | Mid Wing      | Swept                | Split                 | High kv Rating | Double booms                  | Gyroscope                | Low Batter Capacity   | T-Tail              |
| Glue                      | LW-PLA       |                |                 | 4 Blade          | Four Wheels               |                        | Metal Fleible           | Rectangular | High Wing     |                      | Slotted               |                | Symmetric from side view      | Camera                   | Higher Ampacity       | Cruciform           |
| Fasteners                 | TPU          |                |                 |                  | Ski-Plane                 |                        | Shocks                  | Inverted    |               |                      | Fowler                |                | SubSonics                     | Illumination             | Appropriate C rating  | Dual                |
| Japanese glue free joints | pp           |                |                 |                  |                           |                        |                         | Winglets    |               |                      | Double-Slotted Fowler |                | Super Sonic                   | Extra Battery            |                       | Triple              |
| T-joint glued form fit    |              |                |                 |                  |                           |                        |                         | Triangular  |               |                      | Junkers               |                | High capacity sub sonic       | Special Speed Controller |                       | V                   |
| Soldering                 |              |                |                 |                  |                           |                        |                         |             |               |                      | Gouge                 |                | High manurability super sonic |                          |                       | Inverted V          |
|                           |              |                |                 |                  |                           |                        |                         |             |               |                      | Fairey-Youngman       |                |                               |                          |                       | Inverterd Y         |
|                           |              |                |                 |                  |                           |                        |                         |             |               |                      | Zap                   |                |                               |                          |                       | Twin                |
|                           |              |                |                 |                  |                           |                        |                         |             |               |                      | Krueger               |                |                               |                          |                       | Boom                |
|                           |              |                |                 |                  |                           |                        |                         |             |               |                      | Gurney                |                |                               |                          |                       | High Boom           |
|                           |              |                |                 |                  |                           |                        |                         |             |               |                      | Leading Edge Droop    |                |                               |                          |                       | Multiple-plane tail |
|                           |              |                |                 |                  |                           |                        |                         |             |               |                      | Handley-page          |                |                               |                          |                       |                     |

## Concept Generation

Excel table which combined morphological chart and crap shoot method to generate 100 concepts



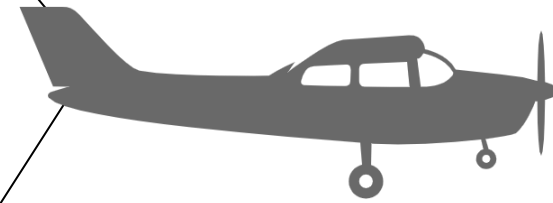
# Concept Generation



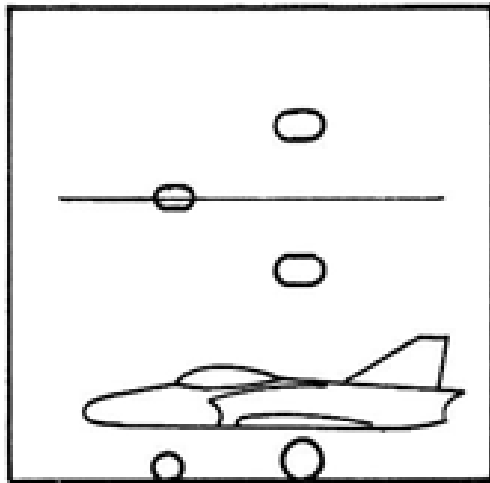
*Concept Generation Chart*

| 3D Material | Landing Gear              | Wings       | Wing Location | Aileron/Flaps | Fuselage               | Tail         |
|-------------|---------------------------|-------------|---------------|---------------|------------------------|--------------|
| PLA         | Tricycle with Front Wheel | Elliptical  | Low Wing      | Plain         | Flying boat            | Conventional |
| ABS         | Tricycle with Tail-Wheel  | Tapered     | Mid Wing      | Split         | Double booms           | T-Tail       |
| LW-PLA      | Four Wheels               | Rectangular | High Wing     | Slotted       | Subsonic               | Cruciform    |
|             |                           |             |               |               | High Capacity Subsonic | Triple       |
|             |                           |             |               |               |                        | Twin         |
|             |                           |             |               |               |                        | Boom         |
|             |                           |             |               |               |                        | High Boom    |

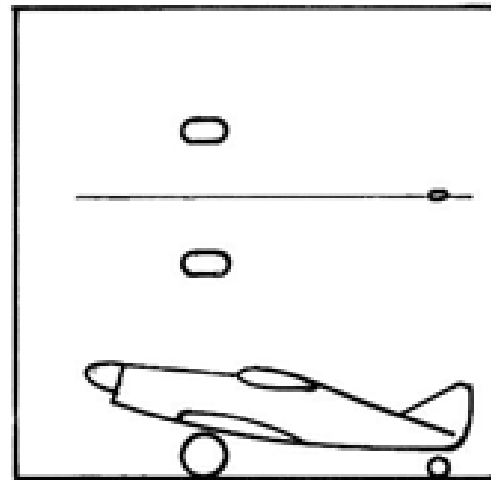
# Concept Generation



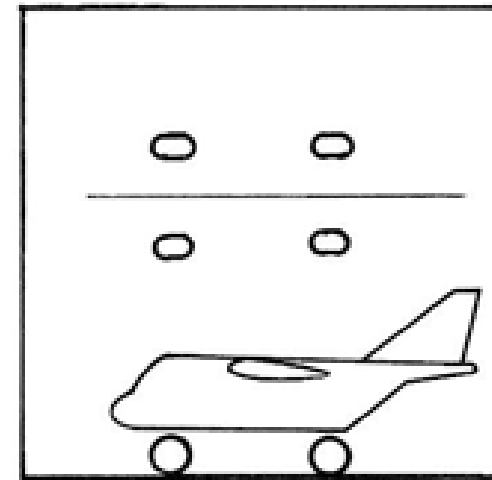
## Landing Gear Configuration



Tricycle with  
Front Wheel

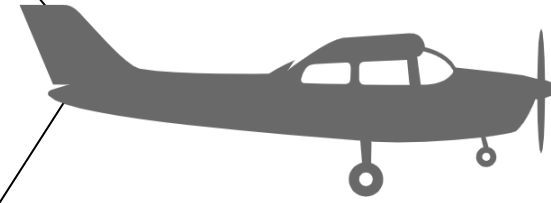


Tricycle with  
Tail Wheel

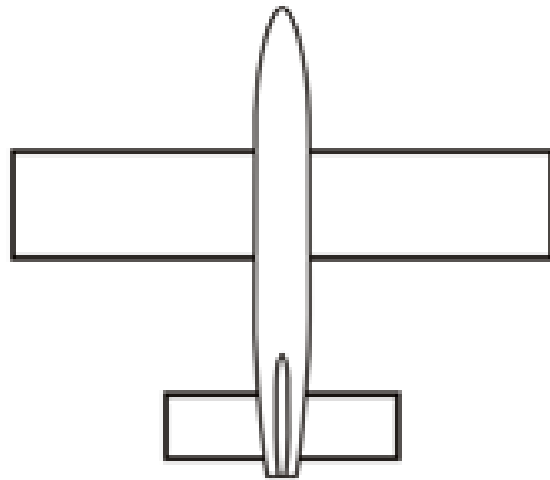


Four Wheel

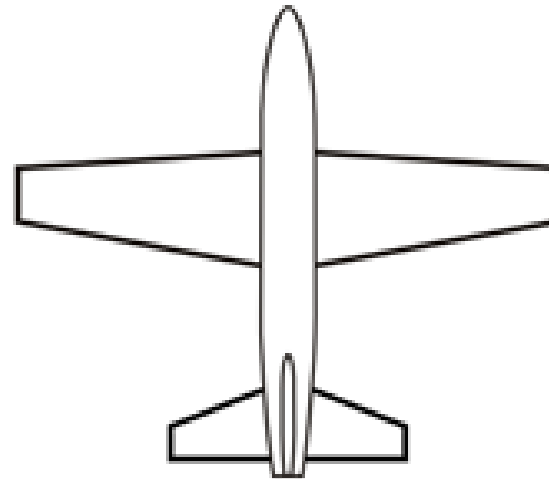
# Concept Generation



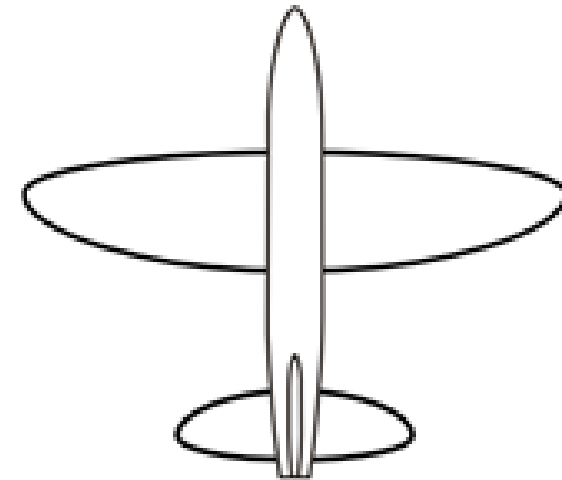
## Wing Planform



Rectangular

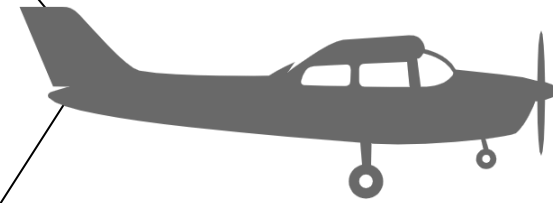


Tapered

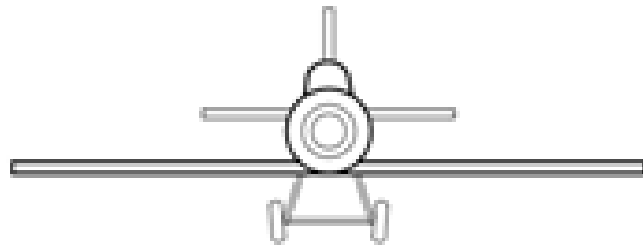


Elliptical

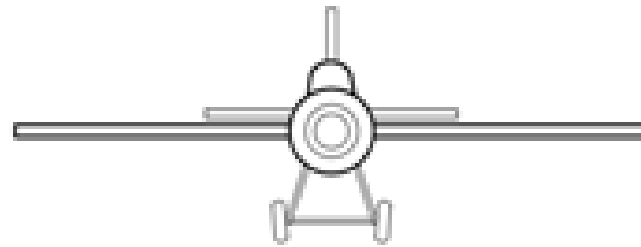
# Concept Generation



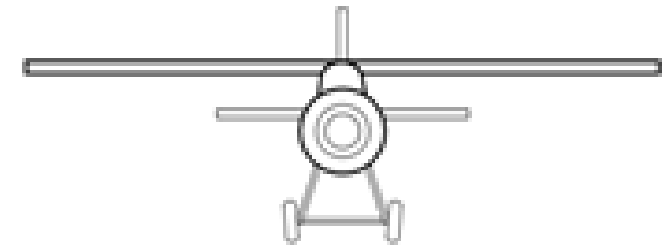
## Wing Location



Low Wing

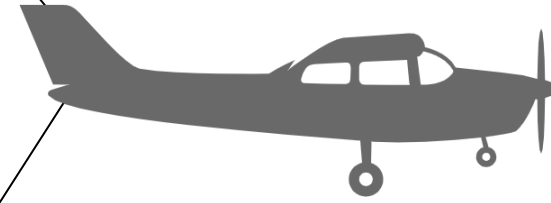


Mid Wing



High Wing

# Concept Generation



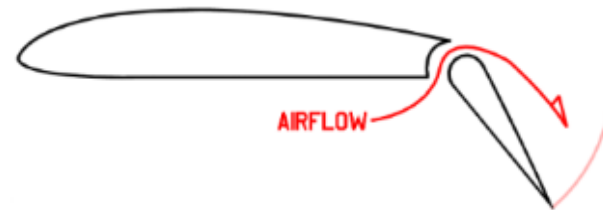
## Aileron and flaps



Plain Flap

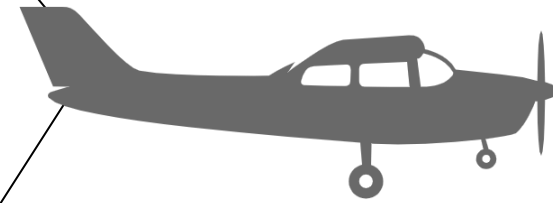


Split



Slotted

# Concept Generation



## Fuselage



Subsonic



High Capacity Subsonic

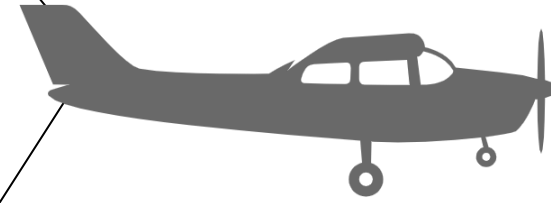


Flying Boat

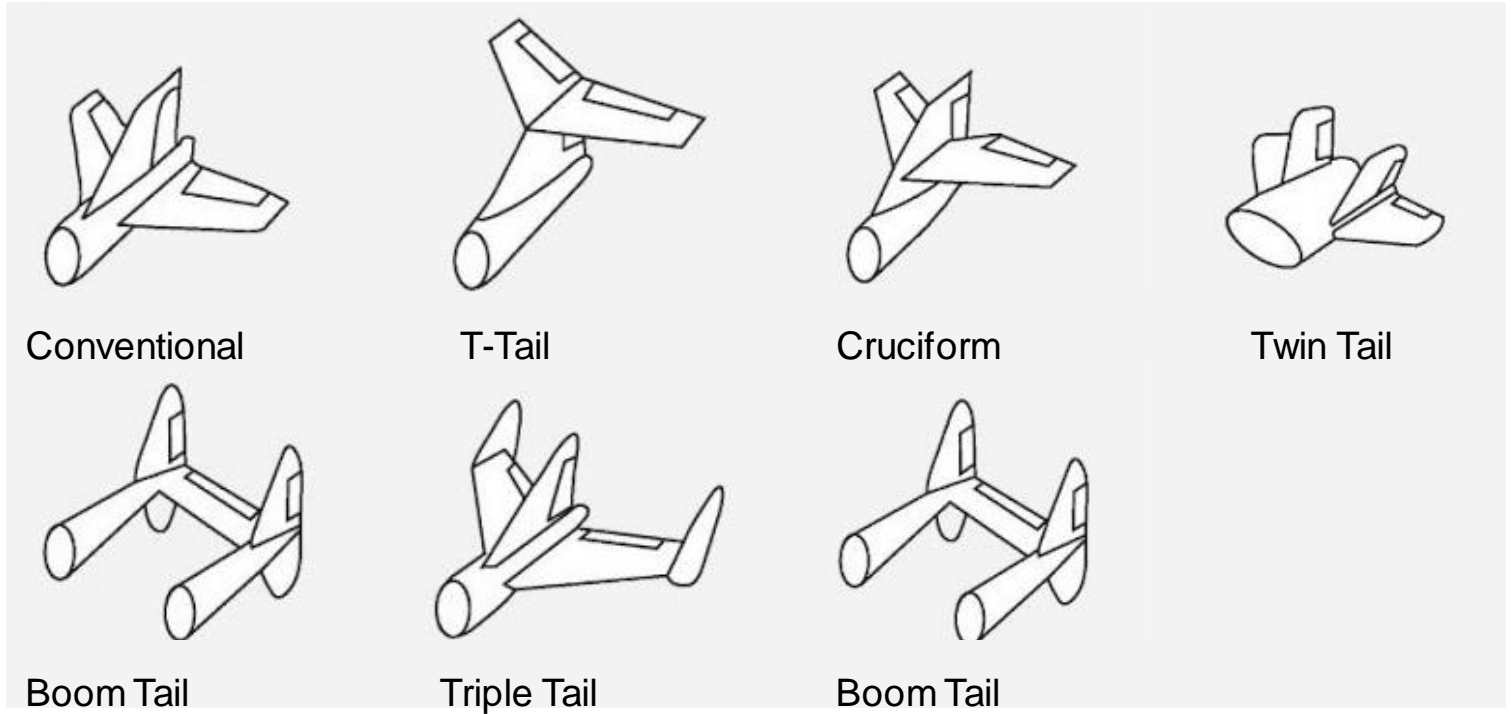


Double Boom

# Concept Generation



## Tail Configuration



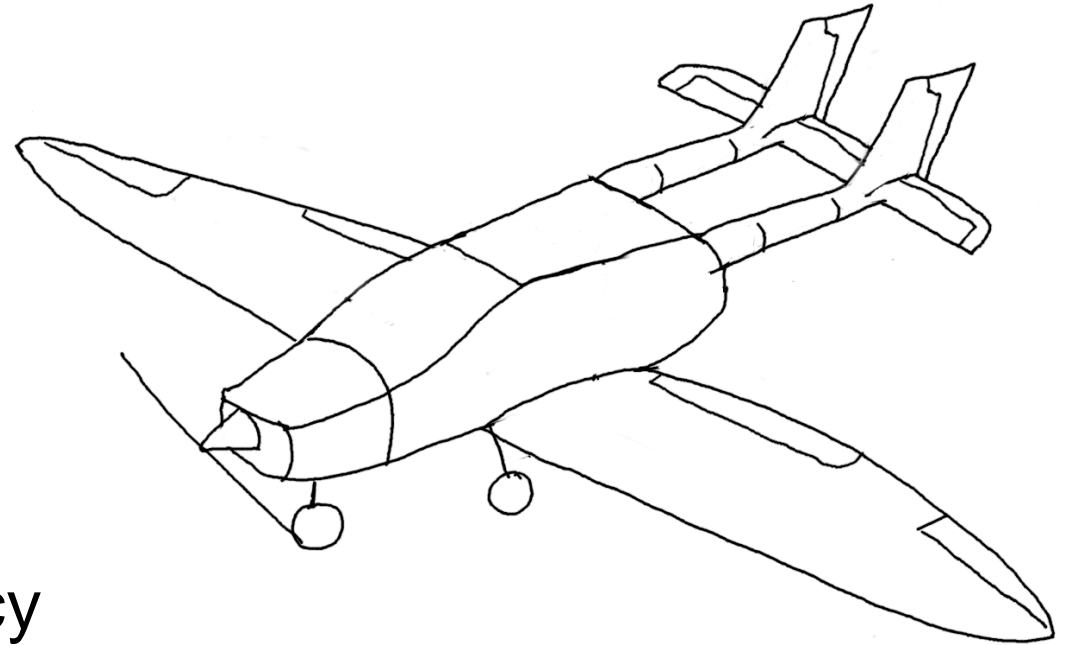


# Detailed Concept Backup

# Concept One

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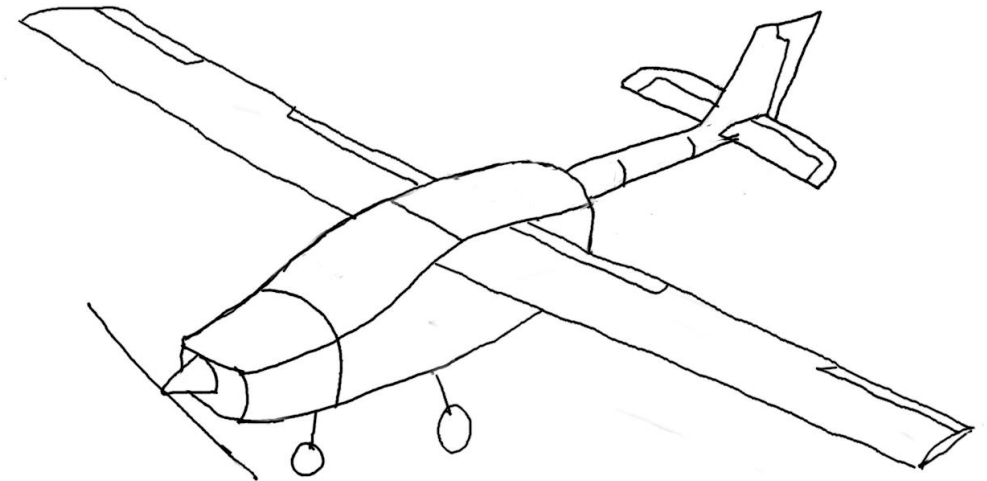
- ✈ Elliptical shaped wings most efficient
- ✈ Reduce wing load
- ✈ Tricycle with singular wheel
- ✈ Split ailerons give more redundancy
- ✈ Boom tail reduces fuselage weight



## Concept Four

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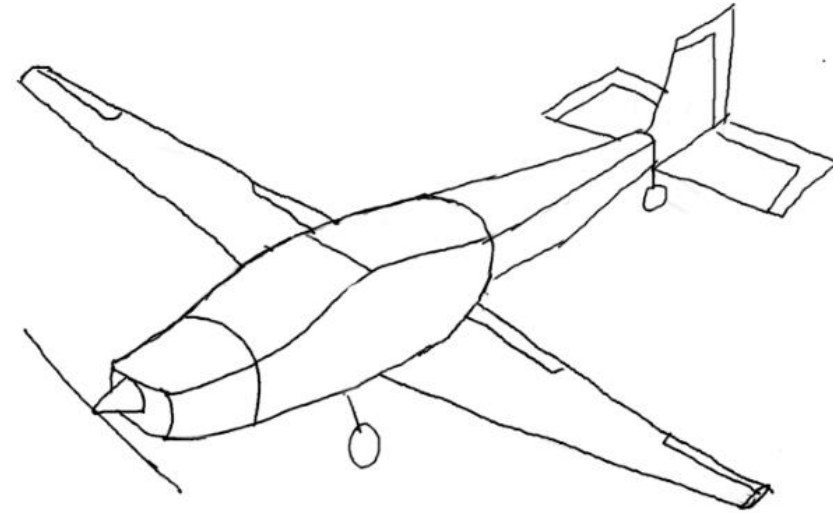
- ✈ Light weight PLA
- ✈ Slotted flap increases lift and decreases drag
- ✈ Rectangular wing is the least efficient design



## Concept Six

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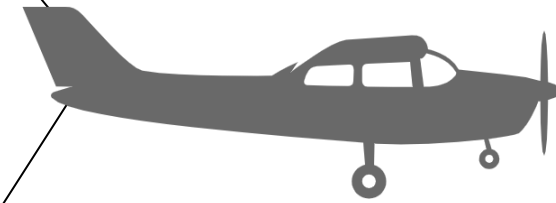
- ✈ Positive angle of attack
- ✈ Increase the lift
- ✈ Conventional Tail
- ✈ Light weight  
PLA



# Detailed Math Backup



# Takeoff Calculation



Takeoff Ground Distance: 
$$S_G = \int_0^{V_{TO}} \frac{VdV}{a} = \frac{1}{2} \int_0^{V_{TO}} \frac{dV^2}{a}$$

Takeoff Velocity: 
$$V_{TO} = 1.2 V_{stall} = 1.2 \sqrt{\frac{W_{TO}}{S_{ref}} \frac{2}{\rho C_{L_{max}}}}$$

Command Window

For an airplane with 15.000 lb weight, 72.000 in wingspan, and 14.000 in chord length yields 7.000 ft<sup>2</sup> wing area, 5.143 aspect ratio, and 34.286 oz/in<sup>2</sup> wing loading

The required velocity for take off is 34.441793 ft/s or 23.483034 mph

The required ground distance for take off considering thrust is 32.506 ft

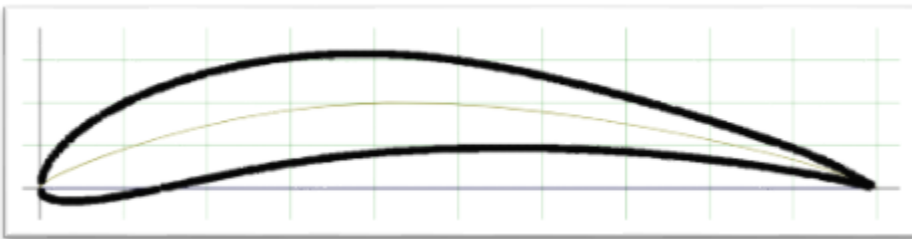
The above doesnt include drag, and thrust is a rough estimate at 8.500000 in the calculation

The required ground distance traveled for take off considering lift and drag is 49.665579 ft

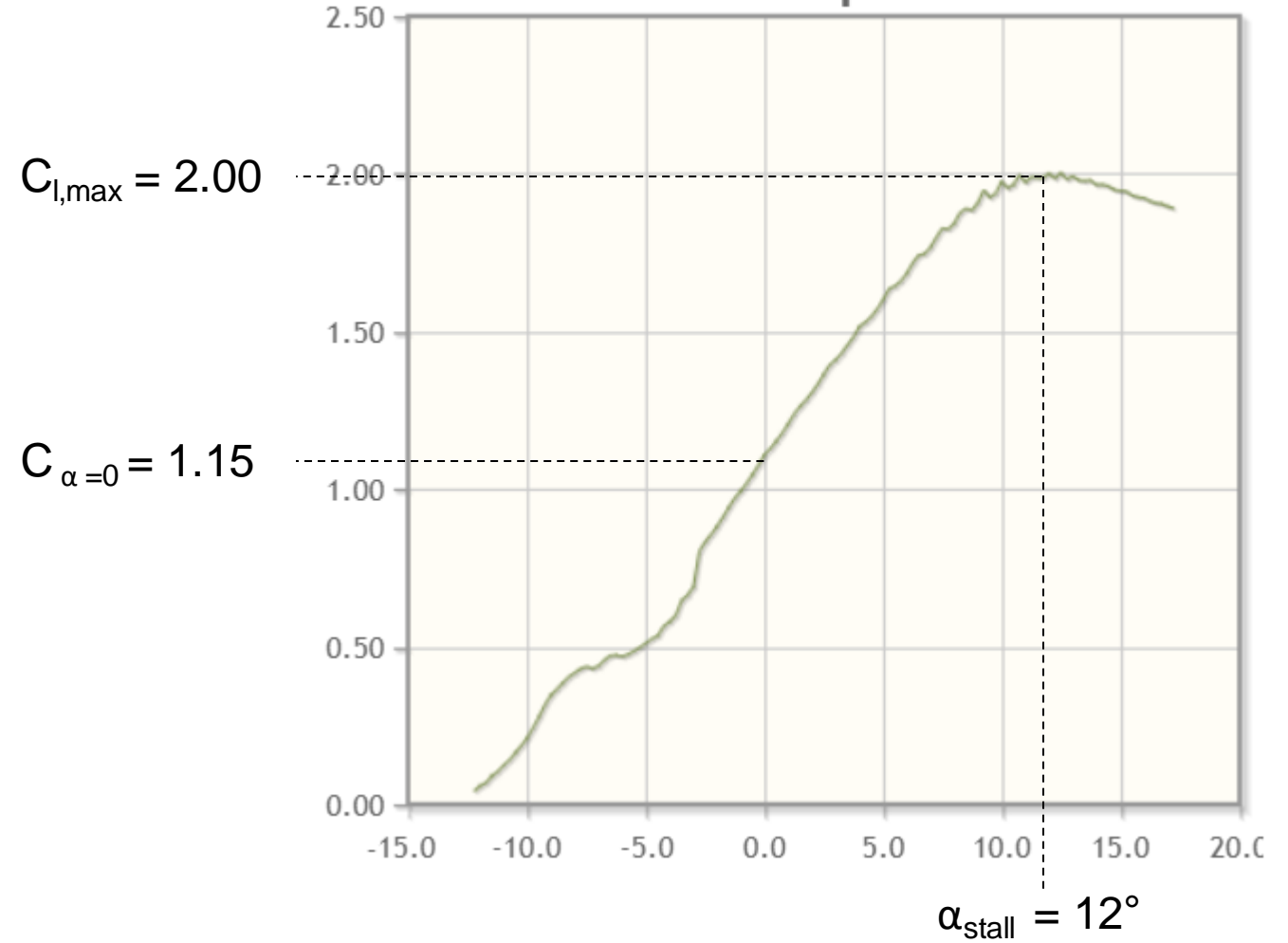
fx >>

# Airfoil Selection

- ✈ Selected the Eppler E423
- ✈ Satisfies targets for lift coefficient and stall angle of attack
- ✈ Designed as a heavy lift UAV airfoil

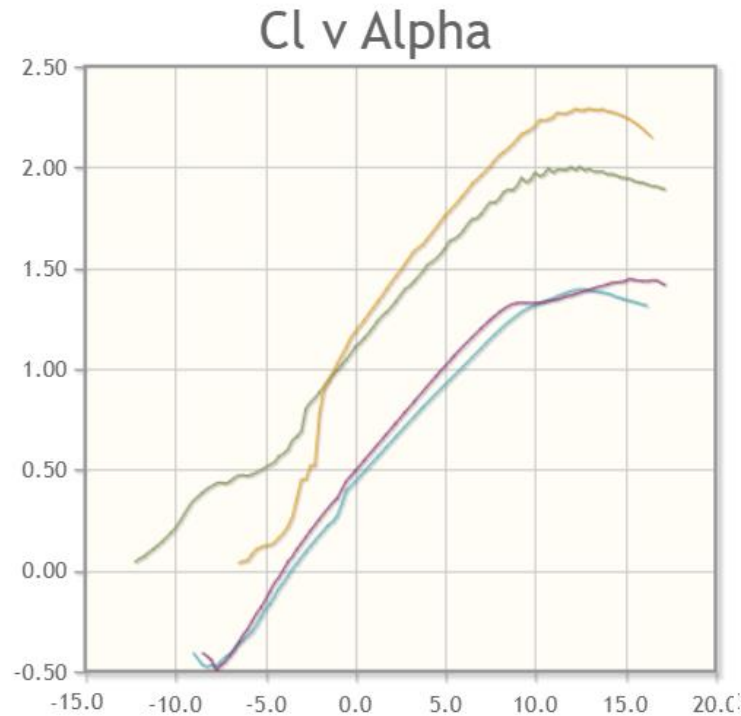
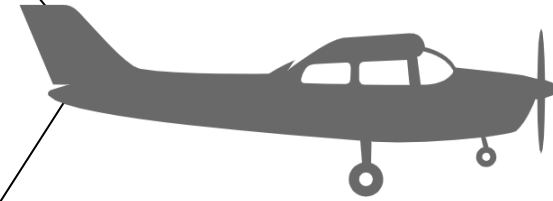


Lift Coefficient ( $C_l$ ) vs Angle of Attack ( $\alpha$ )

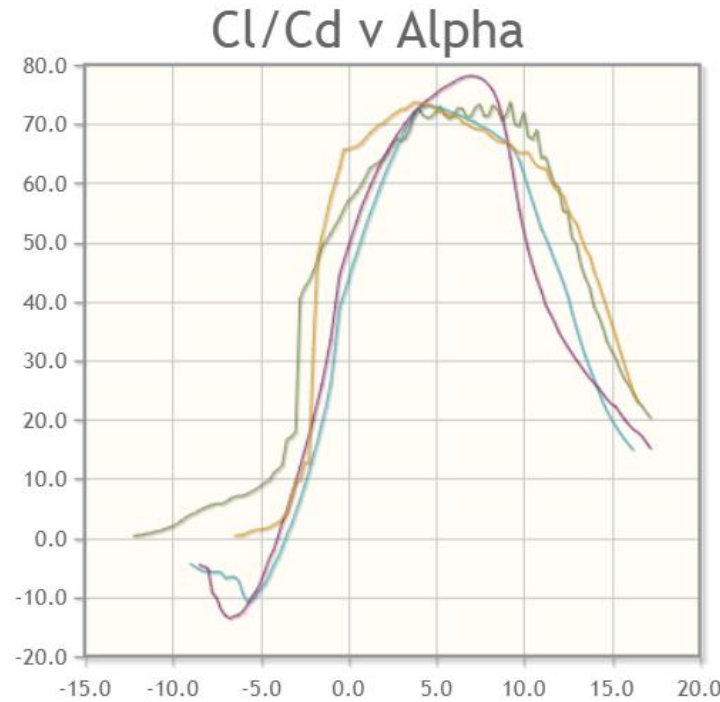




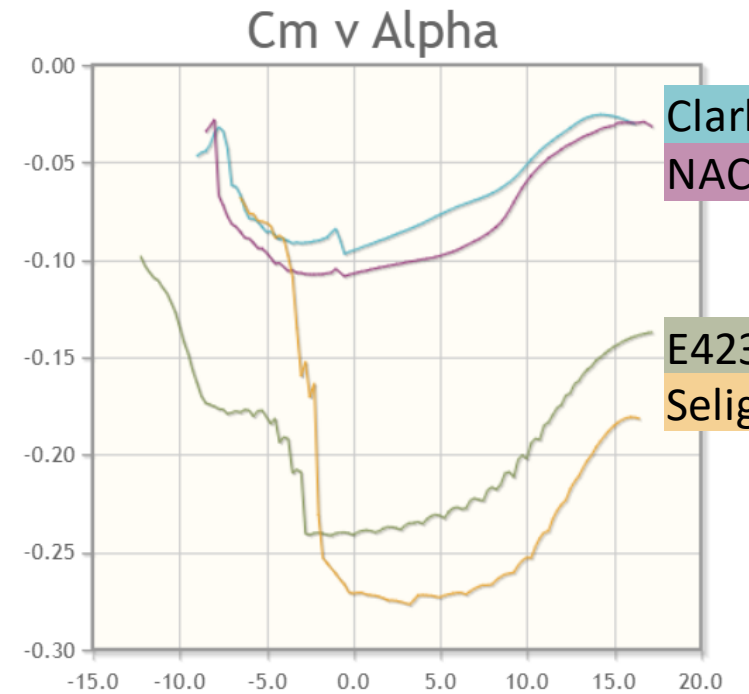
# Airfoil Selection



**Compare lift coefficient**



**Compare lift-to-drag ratio**

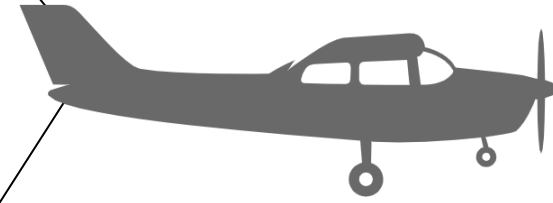


Clark Y  
NACA 4412

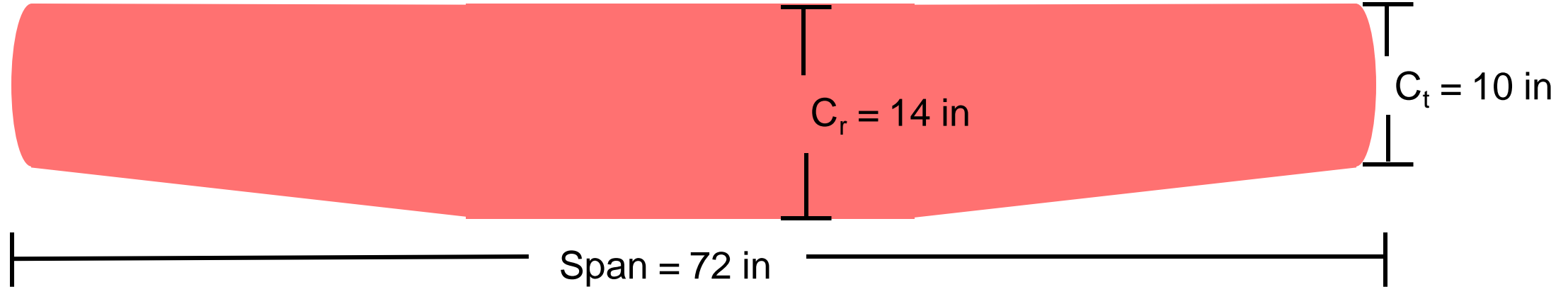
E423  
Selig 1223

**Compare moment coefficient**

# Wing Sizing



Flight Path ↑



✈ Produces takeoff distance of 49.6 ft at 23.4 mph

✈ Wing loading of 34.3 oz/in<sup>2</sup> and aspect ratio of 5.1

# Bill of Materials Backup



| Item No. | Items Purchased with Grant Money                      | \$ / Unit  | Qty | Price      | Price w/ Tax |
|----------|---|------------|-----|------------|--------------|
| 1        | SAE Competition Registration                          | \$1,100.00 | 1   | \$1,100.00 | \$ 1,100.00  |
| 2        | Onxy 22.2V 63 4000mAh 30C LiPo Battery, EC5 connector | \$ 76.39   | 2   | \$ 152.78  | \$ 164.24    |
| 3        | Futaba R2006GS 6-Channel S-FHSS Receiver              | \$ 39.99   | 1   | \$ 39.99   | \$ 42.99     |
| 4        | Onyx LiPo Charge Protection Bag                       | \$ 8.99    | 1   | \$ 8.99    | \$ 9.66      |
|          | Item 2, 3, 4 with \$30 off discount                   |            |     |            | \$ (30.00)   |
| 5        | E-flite Prop Adapters: Power 90                       | \$ 17.09   | 1   | \$ 17.09   | \$ 18.20     |
| 6        | E-flite X-Mount/Hardware: Power 90/110/160            | \$ 14.24   | 1   | \$ 14.24   | \$ 15.17     |
| 7        | APC 18x8 Thin Electric Propeller                      | \$ 11.13   | 1   | \$ 11.13   | \$ 11.85     |
|          | Item 5, 6, 7, shipping and handling                   |            |     |            | \$ 15.40     |
| 8        | Al 6061 Spar 0.37"x0.035"x0.305"                      | \$ 9.94    | 2   | \$ 19.88   | \$ 39.76     |
| 9        | Al 6061 Spar 0.25"x0.035"x0.18"                       | \$ 9.73    | 2   | \$ 19.46   | \$ 38.92     |
|          | Item 8 and 9 shipping and handling                    |            |     |            | \$ 21.96     |
| 10       | Colorfabb Light Weight PLA 0.75kg roll                | \$ 30.91   | 6   | \$ 185.46  | \$ 185.46    |
|          | Item 10 Shipping                                      |            |     |            | \$ 43.20     |
| 11       | Colorfabb Light Weight PLA 0.75kg roll                | \$39.60    | 5   | \$198.00   | \$198.00     |
|          |   |            |     |            |              |
|          |   |            |     | Total      | \$ 1,874.81  |

## Purchasing Log





|             | Items   | Category   | Qty | Retail Price | Price   | Total Wt. [oz] | % Complete | Completed |
|-------------|---|------------|-----|--------------|---------|----------------|------------|-----------|
| Electronics | FlightLine RC 5055-390kV Brushless Motor            | Propulsion | 1   | \$59.99      | \$0.00  | 14.460         | 100.00%    | YES       |
|             | E-Flite Power 90 Brushless Outrunner Motor 325Kv    | Propulsion | 1   | \$129.99     | \$0.00  | 15.800         | 100.00%    | YES       |
|             | ADMIRAL 6s, 4000 mAh, 40 C, 22.2 V Battery          | Power      | 1   | \$79.99      | \$0.00  | 21.090         | 100.00%    | YES       |
|             | Spare Battery                                       | Power      | 1   | \$79.99      | \$79.99 | 21.090         | 92.31%     | NO        |
|             | Prop Adapters: Power 90                             | Fastener   | 2   | \$17.09      | \$17.09 |                | 76.92%     | NO        |
|             | X-Mount/Hardware: Power 90, Motor mounting hardware | Fastener   | 1   | \$14.24      | \$14.24 | 0.120          | 92.31%     | NO        |
|             | Futaba 6J 6-Channel S-FHSS System                   | Control    | 1   | \$179.99     | \$0.00  | -              | 100.00%    | YES       |
|             | Futaba R2106GF 6-Channel S-FHSS Micro Receiver      | Control    | 1   | \$29.99      | \$29.99 | 0.140          | 92.31%     | NO        |
|             | ZTW GECKO 85A ESC WITH 8A SBEC WITH XT-60 CONNECTOR | Control    | 1   | \$49.36      | \$0.00  | 2.650          | 100.00%    | YES       |
|             | Hitec HS-311 Plastic Gear Standard Servo            | Control    | 7   | \$62.93      | \$0.00  | 10.570         | 92.31%     | NO        |
|             | Red Arming Plug                                     | Safety     | 1   | \$6.47       | \$6.47  | 0.130          | 92.31%     | NO        |
|             | SAE 2019 Power Limiter V2 regular class 1000W       | Safety     | 1   | \$75.00      | \$0.00  | 0.720          | 100.00%    | YES       |
|             | Cell Meter Battery Capacity Checker                 | Safety     | 1   | \$8.99       | \$8.99  | 1.760          | 100.00%    | YES       |

## Bill of Materials: Electronics



|             | Items                 | Category | Qty | Retail Price | Price   | Total Wt. [oz] | % Complete | Completed |
|-------------|-----------------------|----------|-----|--------------|---------|----------------|------------|-----------|
| 3D Printing | Ailerons              | Wing     | 2   |              | \$0.00  | 0.000          | 69.23%     | NO        |
|             | Flap                  | Wing     | 2   |              | \$0.00  | 0.000          | 61.54%     | NO        |
|             | Hinges                | Wing     | 4   |              | \$0.00  | 0.000          | 61.54%     | NO        |
|             | Support Spar          | Wing     | 2   |              | \$23.58 | 0.000          | 61.54%     | NO        |
|             |                       |          |     |              |         |                | 0.00%      |           |
|             | Cargo Bay             | Fuselage | 1   |              | \$0.00  | 0.000          | 61.54%     | NO        |
|             | Nose Cone             | Fuselage | 1   |              | \$0.00  | 0.000          | 61.54%     | NO        |
|             | Electronics Bay       | Fuselage | 1   |              | \$0.00  | 0.000          | 61.54%     | NO        |
|             | Hinges                | Fuselage | 4   |              | \$0.00  | 0.000          | 61.54%     | NO        |
|             |                       |          |     |              |         |                | 0.00%      | NO        |
|             | Elevator              | Tail     | 2   |              | \$0.00  | 0.000          | 61.54%     | NO        |
|             | Rudder                | Tail     | 1   |              | \$0.00  | 0.000          | 61.54%     | NO        |
|             | Vertical Stabilizer   | Tail     | 1   |              | \$0.00  | 0.000          | 61.54%     | NO        |
|             | Horizontal Stabilizer | Tail     | 1   |              | \$0.00  | 0.000          | 61.54%     | NO        |
|             | Hinges                | Tail     | 4   |              | \$0.00  | 0.000          | 61.54%     | NO        |
|             |                       |          |     |              |         |                |            |           |
|             |                       |          |     |              |         |                |            |           |

## Bill of Materials: 3D Printing



|              | Items                           | Category | Qty | Retail Price | Price  | Total Wt. [oz] | % Complete | Completed |
|--------------|---------------------------------|----------|-----|--------------|--------|----------------|------------|-----------|
| Landing Gear | Dubro Super Lite Wheels 3"      | Wheel    | 2   | \$8.99       | \$0.00 | 0.488          | 100.00%    | YES       |
|              | Sullivan SkyLite Wheel          | Wheel    | 2   | \$38.66      | \$0.00 | 4.460          | 100.00%    | YES       |
|              | Dubro Axle Shaft                | Fastener | 2   | \$6.79       | \$0.00 | 2.400          | 100.00%    | YES       |
|              | Dubro Tail Wheel Assembly       | Fastener | 1   | \$3.99       | \$3.99 | 0.176          | 92.31%     | NO        |
|              | Dubro Tail Wheel 1"             | Wheel    | 1   | \$2.48       | \$2.48 | 0.680          | 92.31%     | NO        |
|              |                                 |          |     |              |        |                |            |           |
|              | [Shocks, if needed]             |          |     |              |        |                | 15.38%     |           |
|              | [Shocks accessories, if needed] |          |     |              |        |                | 15.38%     |           |

## Bill of Materials: Landing Gear





|       | Items                 | Category | Qty | Retail Price | Price   | Total Wt. [oz] | % Complete | Completed |
|-------|-----------------------|----------|-----|--------------|---------|----------------|------------|-----------|
| Cargo | Size Five Soccer Ball | Cargo    | 1   | \$15.00      | \$15.00 | 15.000         | 92.31%     | NO        |
|       | Velcro Bands          | Fastener | 4   | \$2.10       | \$2.10  | 0.800          | 92.31%     | NO        |
|       | Steel Plates          | Cargo    | 6   | \$8.35       | \$0.00  | 96.000         | 100.00%    | YES       |
|       | Metal Screw           | Fastener | 6   | \$1.18       | \$1.18  | 0.423          | 92.31%     | NO        |

### Bill of Materials: Cargo

|         | Items                                 | Category    | Qty | Retail Price | Price   | Total Wt. [oz] | % Complete | Completed |
|---------|---------------------------------------|-------------|-----|--------------|---------|----------------|------------|-----------|
| Testing | Flite Test Water-Resistant Foam Board | Prototype   | 10  | \$2.99       | \$29.90 | 40.212         | 92.31%     | NO        |
|         | Gorilla Glue Hot Glue Sticks          | Prototype   | 30  | \$3.97       | \$3.97  | 4.480          | 92.31%     | NO        |
|         | Polylactic Acid                       | Filament    | 3   | \$20.99      | \$0.00  | 317.700        | 100.00%    | YES       |
|         | Acrylonitrile Butadiene Styrene       | Filament    | 5   | \$18.99      | \$0.00  | 882.500        | 100.00%    | YES       |
|         | Flexible                              | Filament    | 2   | \$26.99      | \$0.00  | 141.200        | 100.00%    | YES       |
|         | Light Weight Polylactic Acid          | Filament    | 2   | \$57.79      | \$54.00 | 105.600        | 100.00%    | YES       |
|         | Loctite Gel Control 4g Super Glue     | Fastener    | 2   | \$2.98       | \$2.98  | 0.560          | 92.31%     | NO        |
|         | APC Electric Propeller 16x8E          | Propulsion  | 1   | \$8.42       | \$8.42  | 1.830          | 100.00%    | YES       |
|         | APC Electric Propeller 18x8E          | Propulsion  | 1   | \$11.13      | \$11.13 | 3.030          | 92.31%     | NO        |
|         | APC Electric Propeller 18x10E         | Propulsion  | 1   | \$11.13      | \$0.00  | 2.570          | 100.00%    | YES       |
|         | Door Hinge                            | Thrust Test | 1   | \$1.34       | \$1.34  | 0.700          | 100.00%    | YES       |
|         | Poplar Board                          | Thrust Test | 1   | \$4.71       | \$0.00  | 17.000         | 100.00%    | YES       |
|         |                                       |             |     |              |         |                |            |           |
|         |                                       |             |     |              |         |                |            |           |

## Bill of Materials: Testing



| Sum total from each column   | Value      |
|--|------------|
| Total # of parts   | 130        |
| Total retail value of parts  | \$1,052.99 |
| Total expense to T513 (some parts were sponsored or reused from last | \$316.84   |
| Total weight of parts (units: lb)                                    | 107.896    |
| Total weight of electronics (units: lb)                              | 5.533      |
| Total weight of airplane components so far (units: lb)               | 4.847      |
|  |            |
|  |            |
| BoM Progress Tracking  | Value      |
| # of parts left to order and/or 3d print                             | 27         |
| # of parts at 100% completion  | 19         |
| # of parts in BoM  | 49         |
| Total BoM % completion   | 39%        |

## Bill of Materials: Project Progress



# Targets and Metrics Backup

| Function        | Metric                            | Target        | Method of Validation                 | Tools for Validation  |
|-----------------|-----------------------------------|---------------|--------------------------------------|---|
| Accelerate      |                                   |               |                                      |   |
| Generate Thrust | Force                             | 10 lbf        | Experimental                         | Force Gauge/ Scale  |
|                 | Propeller Size                    | 14in - 18in   | Physical Experiment and Computations | Test sized propellers to determine maximum thrust and compare against DriveCalc program |
|                 | Electric Motor Rating Kv Rating   | 390 Kv Rating | Given by Manufacture                 | Manufacture Validated   |
|                 | Electric Motor Maximum Power      | 950W          | Experimental                         | Apply current and measure voltage with a voltmeter                                      |
|                 | Propulsion System Battery Voltage | 22.2 V        | Experimental                         | Voltmeter   |
| Taxi on Runway  | Angular Steering for Front Wheel  | -60° to 60°   | Experimental                         | Attach to front wheel, test total rotation, and record time                             |

## Targets and Metrics

| Function       | Metric                             | Target           | Method of Validation         | Tools for Validation                             |
|----------------|------------------------------------|------------------|------------------------------|--|
| Accelerate     |                                    |                  |                              |  |
| Apply Throttle | Velocity for Takeoff               | 30 mph           | Theoretical Calculations     | MATLAB, PropCal 3.0                              |
|                | Ground Distance for Takeoff        | Less than 100 ft | Theoretical and Experimental | MATLAB and flight testing                        |
|                | Propulsion system battery capacity | 4000 mAh         | Given by manufacturer        | Manufacturer Validated                           |
|                | Propulsion System battery duration | 10 minutes       | Theoretical Calculations     | Determined by current drawn by propulsion system |
|                | Power limiter top limit            | 1000 W           | Competition Requirement      | Manufacturer Validated                           |

## Targets and Metrics



## Targets and Metrics

| Lift                           |                       |                           |  |  |
|--------------------------------|-----------------------|---------------------------|--|--|
| Generate Pressure Differential | Angle of Attack       | 2-5 Degrees               | Database Comparative Analysis                      | xlfr5  |
|                                | Coefficient of Lift   | Greater than 1.0          | Theoretical Calculations                           | MATLAB   |
|                                | Coefficient of Drag   | Less than 1.0             | Theoretical Calculations                           | MATLAB   |
|                                | Wingspan              | 60 – 120 in               | Experimental and Theoretical Calculations          | Prototyping, Solid works simulations, and MATLAB |
|                                | Wing Loading          | 10 –20 oz/ft <sup>2</sup> | Finite Element Analysis                            | MATLAB, SOLIDWORKS Simulation                    |
| Structure                      | Gross-take-off weight | Less than 55 lbs          | Theoretical Calculations, Physical Experimentation | SOLIDWORKS Simulation, digital scale             |
| Inhibit Stall                  | Stall Speed           | Greater than 30mph        | Theoretical Calculation                            | MATLAB simulation                                |
|                                | Stall Angle of Attack | Greater than 25 Degrees   | Experimentation                                    | Flight testing and XLFR5                         |



| Function             | Metric               | Target          | Method of Validation                         | Tools for Validation                                    |
|----------------------|----------------------|-----------------|--|---|
| Decelerate           |                      |                 |  |   |
| Reduce throttle      | Velocity for Landing | Less than 30mph | Theoretical calculations and experimentation | MATLAB, Prop Calc 3.0, testing motor and flight testing |
| Engage Flaps         | Time to deploy       | 1 Second        | Experimental                                 | Stopwatch   |
|                      | Angle of flaps       | 0°- 30°         | Computer simulation                          | SOLIDWORKS Simulations                                  |
| Stabilize approach   |                      |                 |  |   |
| Absorb Landing Force | Force                | 2x Weight (lbf) | Theoretical                                  | MATLAB and FEA  |

## Targets and Metrics





| Function           | Metric                    | Target                  | Method of Validation | Tools for Validation  |
|--------------------|---------------------------|-------------------------|----------------------|---|
| Maneuver in Flight |                           |                         |                      |   |
| Servo Motors       | Servo Motor Angular Speed | 0.17 sec per 60 degrees | Given by Manufacture | Manufacturer Validated  |
|                    | Angular Pitch Position    | -60° to 60°             | Experimentally Test  | Attach to control surface, test total rotation, and record time |
|                    | Angular Roll Position     | -60° to 60°             | Experimentally Test  | Attach to control surface, test total rotation, and record time |
|                    | Angular Yaw Position      | -60° to 60°             | Experimentally Test  | Attach to control surface, test total rotation, and record time |

## Targets and Metrics



| Function             | Metric   | Target    | Method of Validation     | Tools for Validation                           |
|----------------------|--|-----------|--------------------------|--|
| Secure Cargo         |  |           |                          |  |
| Load/Unload Payload  | Time   | 2 Minutes | Human                    | Load/unload payload from cargo area with hands |
| Carry Payload        | Force  | 5 lbf     | Experimental             |  |
|                      | Radio System Battery Current Capacity          | 1000 mAh  | Rule Requirement         | Manufacturer Validated                         |
|                      | Radio System Battery Time Duration             | 6 min     | Theoretical Calculations | Determined by current drawn by controller      |
| Controller           |  |           |                          |  |
| Radio Control System | Wavelength Frequency                           | 2.4 GHz   | Competition Requirement  | Manufacturer Validated                         |
|                      | Electronic speed controller continuous current | 85 A      | Given by Manufacturer    | Manufacturer Validated                         |

## Targets and Metrics