



# Design of a Compact Pressure Sensor for Multi-Layer Insulation



Team 15: Sebastian Bellini, Jason Carvalho, Stephen Johnson, Michael Kiefer Sponsor: James Jim Martin (NASA Marshall Space and Flight Center) Faculty Advisor: Dr. Wei Guo

## Project Scope

The goal of this project is to design and implement a compact pressure sensor that can fit between the layers of Multi-Layer Insulation (MLI) and measure minute changes in pressure.

## Background

- Multi-Layer Insulation is a thermal insulation system used to protect cryogenic fluids and spacecraft
- The unwanted interstitial pressure within the MLI allows for additional energy transfer between layers via conduction and convection
- Working Conditions
  - Cold Welding
  - Out gassing

## Objectives

- Design a pressure sensor with minimal moving parts
- Minimize wiring and power consumption of the pressure sensor
- Minimize heat produced by the sensor

## Constraints

- Must read a minimum of  $10^{-2}$  Pa
- Must read one sample per second
- Minimally invasive to the MLI

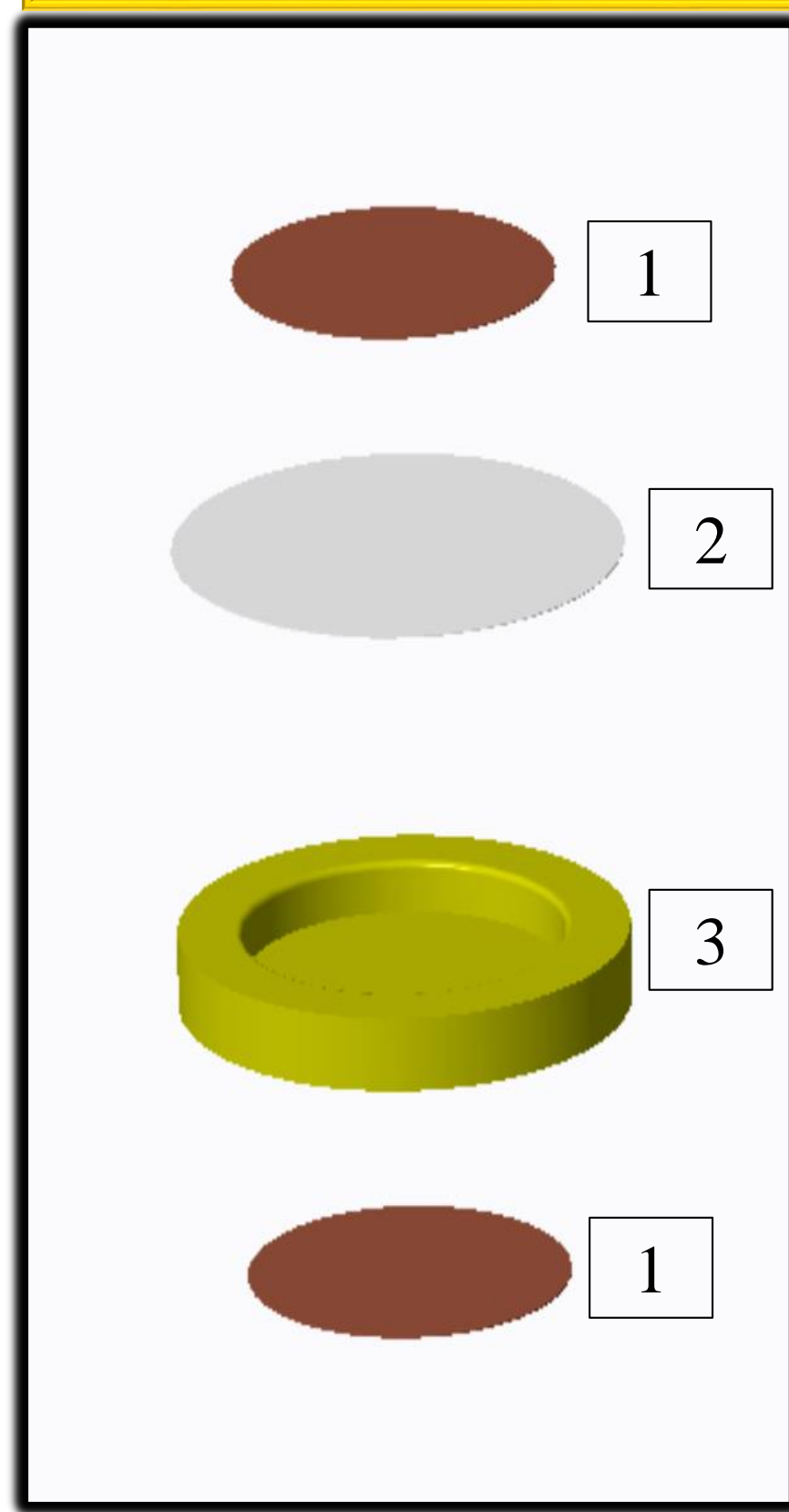
## Challenges

- Achieving a viable price point for a UV silicone adhesive
- Determining a suitable diaphragm substitute
- Manufacturing sensor in vacuum chamber

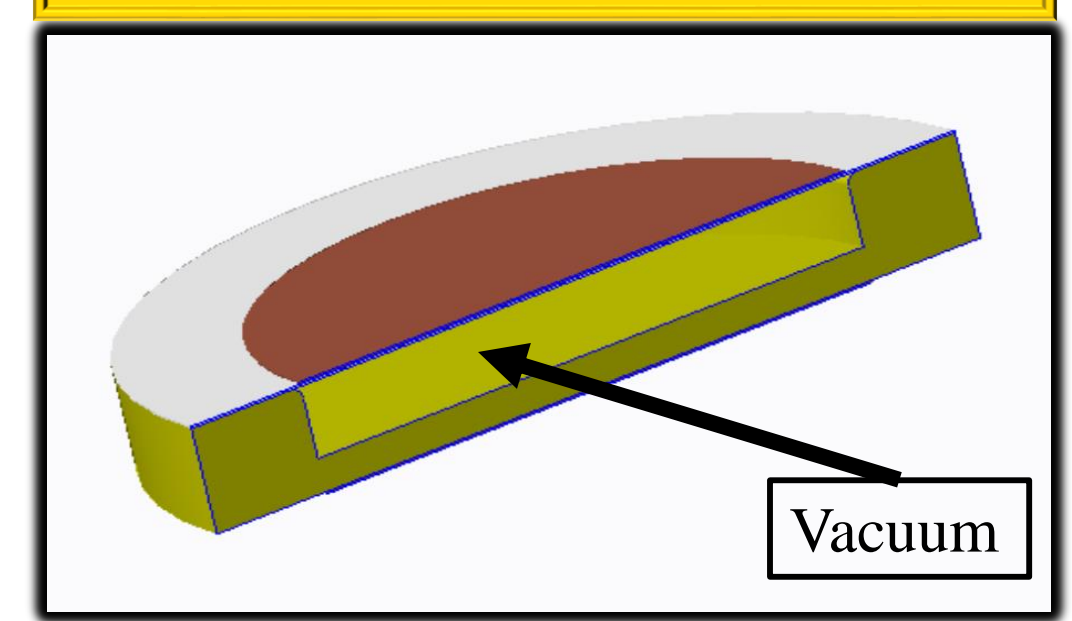
Sample of MLI



Capacitor Design Exploded View



Capacitor Design



This plot demonstrates how the natural frequency relates to the impedance of the capacitor

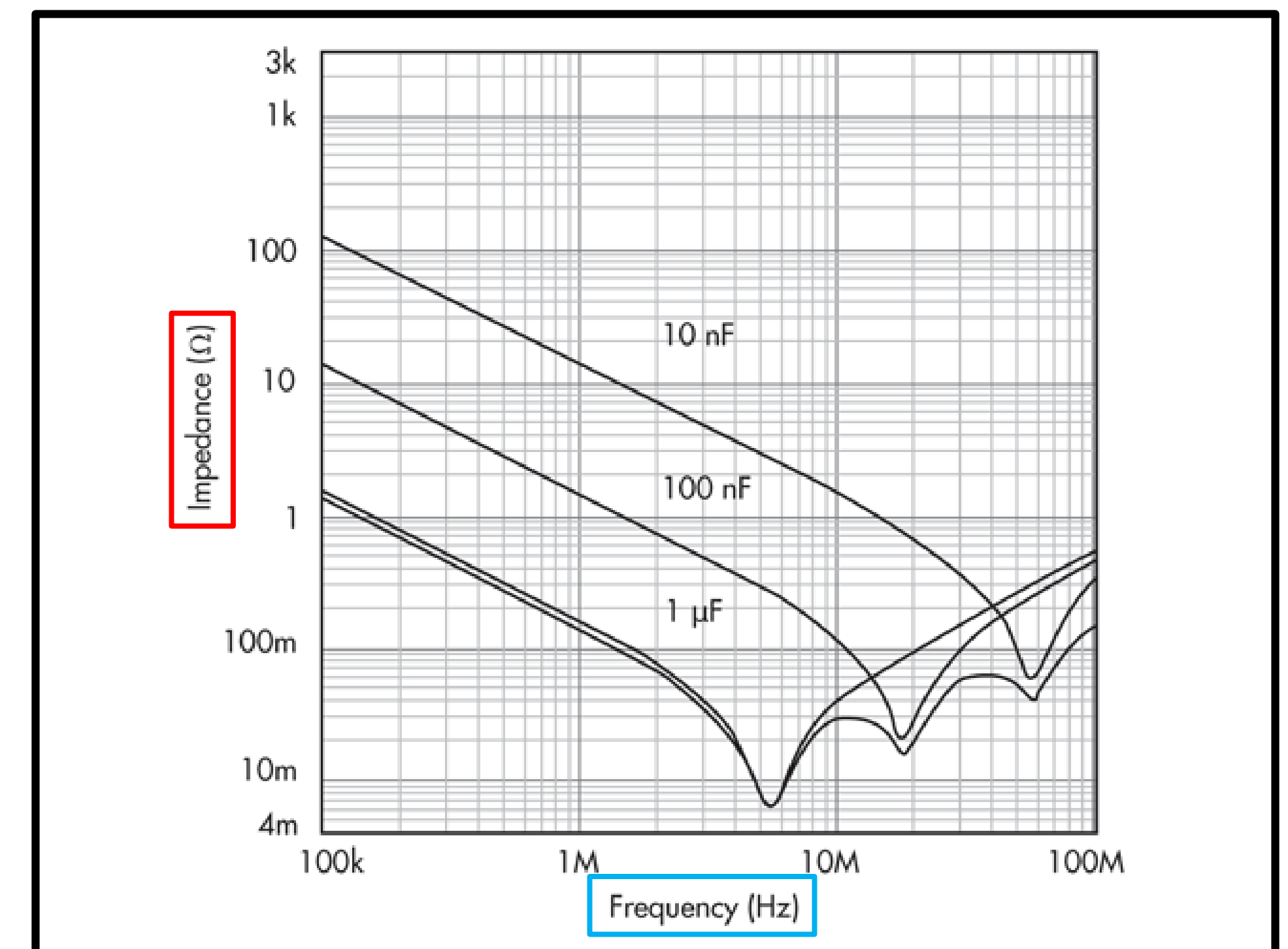
$$V = I * R$$

$$\text{Voltage} = f(\text{frequency})$$

$$\text{Frequency}_R = f(\text{deflection})$$

$$\text{Pressure} = f(\text{deflection})$$

$$\text{Pressure} = f(\text{frequency}_R)$$



## Capacitor Design

- Palladium-Gold sputtered capacitance tract
- Silicone diaphragm
- Silica capacitor base

## Future Work

- Purchase Masterbond UV10 epoxy
- Finish sensor assembly
- Test and calibrate sensor

## Acknowledgements

We would like to thank Dr. Wei Guo for his guidance throughout the entire senior design process.

## Reference

- Ashrafi, Ashkan. *Research Gate*. Research Gate, Aug. 1999. Web.
- "Multi layer insulation, multilayer films for MLI insulation - Dunmore corporation

$$h = \sqrt{\frac{3\pi r^2 * P}{4\pi\sigma_y}}$$

$$w_{max} = - \frac{3\pi r^2 P \left( \left( \frac{1}{\mu} \right)^2 - 1 \right) r^2}{16\pi E h \left( \frac{1}{\mu} \right)^2}$$

$$p = \frac{0.855}{(1 - \mu^2)^{\frac{3}{4}}} * \frac{E\sqrt{\gamma}}{\left( \frac{r}{t} \right)^{\frac{5}{2}} * \frac{l}{r}}$$

$$\gamma = 1 - 0.901(1 - e^{-\phi})$$

$$\phi = \frac{1}{16} \sqrt{\frac{r}{t}}$$

Diaphragm Thickness (ID)	Critical Diaphragm Thickness	Critical Diaphragm Pressure	Maximum Pressure Exerted (Takeoff)	Maximum Deflection @ 150 kPa	Shell Thickness	Critical Body Pressure
125μm	0.50μm	600 kPa	150 kPa	28μm	20μm	400 MPa

$\mu$  = Poisson's Ratio  
 $E$  = Young's Modulus  
 $r$  = diaphragm radius  
 $l$  = sensor length  
 $t$  = shell thickness