

# AUVSI DESIGN COMPETITION

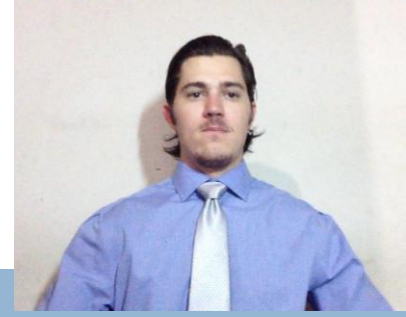
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*Instructor:* Dr. Gupta, Dr. Helzer

*Students:* David Hegg, Christopher Bergljung,  
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# Overview



"The goal of this project is work effectively as an international team to create the best possible aircraft for future success at the 2015 AUVSI SUAS Competition."

**FIPSE**- Fund for the Improvement of Postsecondary Education

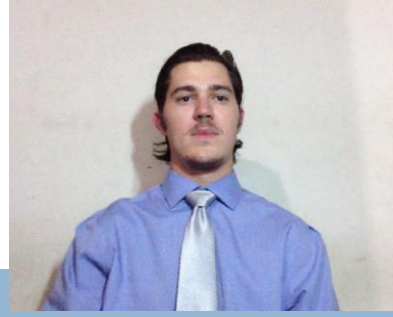
- Two members of Team 8 are currently studying in Itajuba, Brazil
- International experience
- Communication and teamwork skills

## **Tasks:**

- **Design** aircraft, optimizing for competition
- **Build** and modify existing Senior Telemaster plane
- **Program** aircraft for automated VTOL and navigation
- **Test** aircraft and adjust accordingly



# Progress



Team 8 has been working diligently to design the most effective aircraft design for the senior design project.

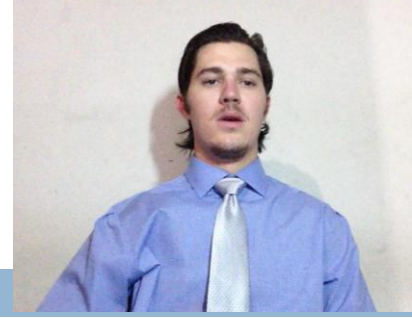


## ***Decision Matrix:***

1. Retrofit Last Year's Plane
  - Cost and Time effective
  - Long flight duration and High Payload
2. Build a Multi-Rotor
  - Great Opportunity to Learn
  - VTOL
  - Foundation for Future

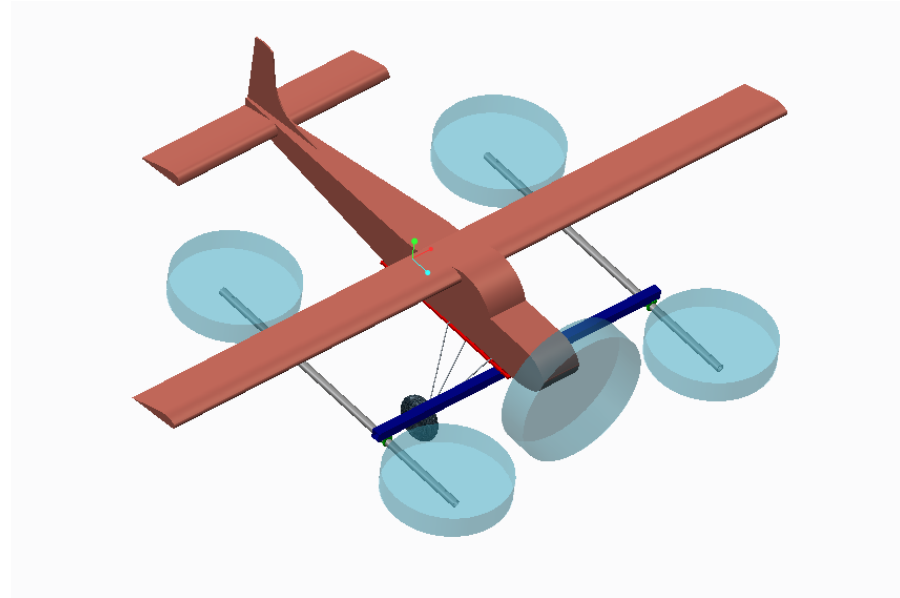
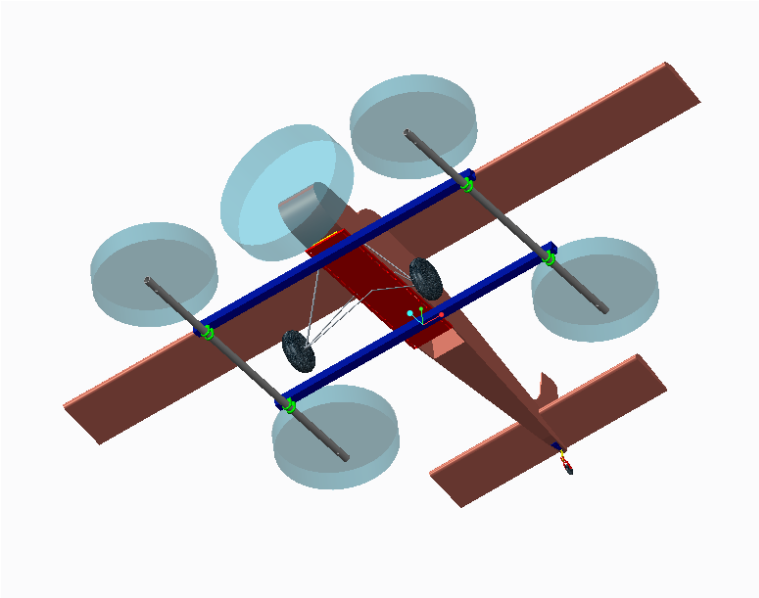
**Decided a *Hybrid Aircraft* would combine the best features of both designs.**

# Frame Design



## **Constraints:**

- Lightweight
- Strong
- Low Cost
- Simple
- Removable
- Vibration damping

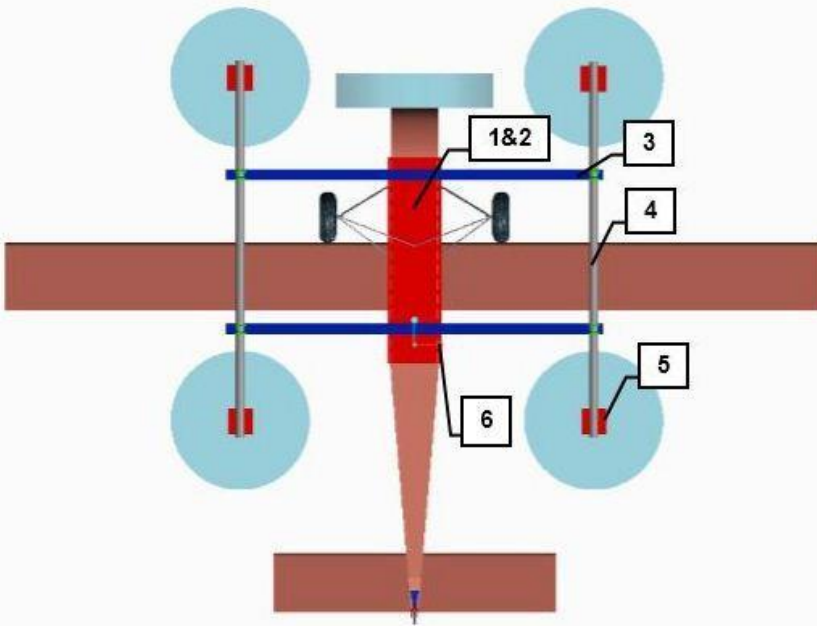


# Frame Design



## ***Material Selection:***

1. G-10 Garolite Base
2. Quick-Recovery Polyurethane Foam Spacer
3. 6061 Aluminum Cross Beams
4. High-Strength Rigid Carbon Fiber booms
5. G-10 Garolite Motor Mount Adapters
6. Industrial Strength Velcro



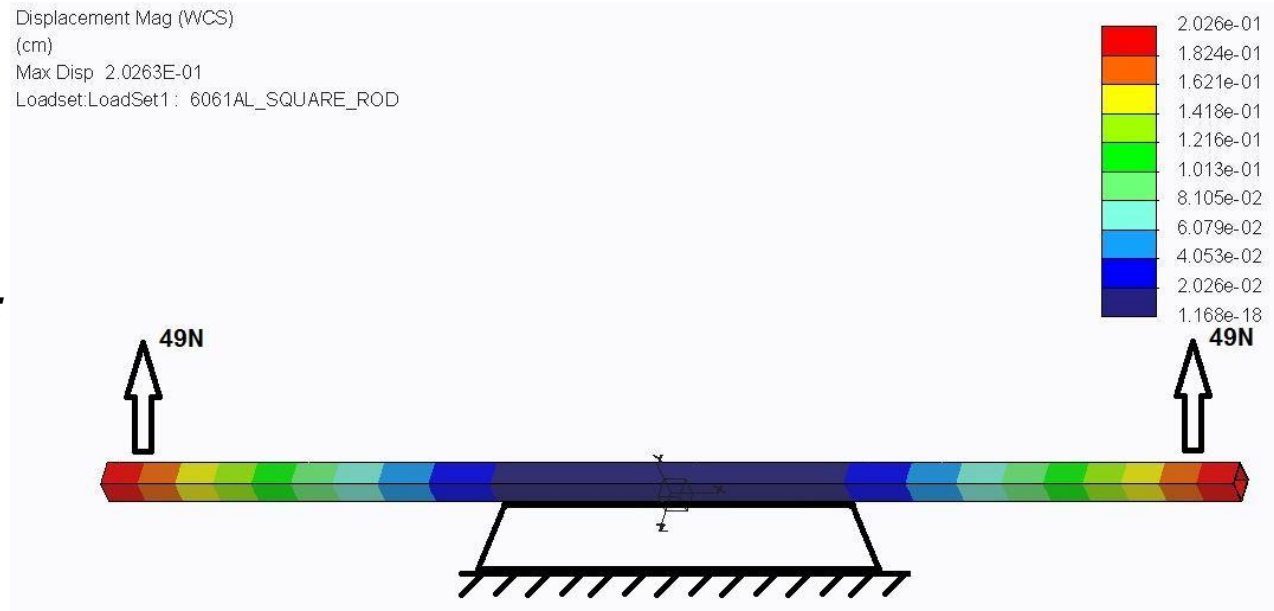
# Frame Design



## **Beam Displacement:**

Force of lift on both ends of the beam:  
4.45kg for each motor  $\approx$  5kg = 49N

Forces are applied 5cm from each end, with the width of the body fixed (15cm)



**Max Displacement:**  
**0.2026cm**

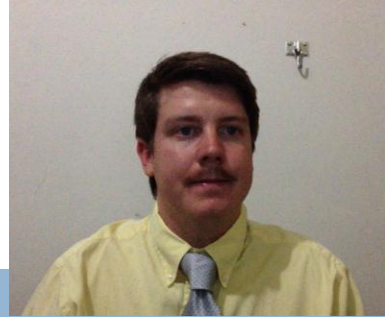
# Frame Design



**Frame Design: Weight and Cost Analysis**

Component	Description	Dimensions	Volume (in <sup>3</sup> )	Density (lb/in <sup>3</sup> )	Weight/Part (lb)	Price	Vendor	Qty.	Weight (lb)	Subtotal	Extras	
G10	Base Excellent Tensile and Impact Strength	23x6.1x.1575"	22.097	0.063	1.392	\$0.00	n/a	1	1.392	\$0.00		
G10	Motor Mount Adapter	2.36 x 2.36 x .1575"	0.879	0.063	0.055	\$0.00	n/a	4	0.221	\$0.00		
Carbon Fiber Tubes	Parallel Arms for holding the motors Excellent Tensile Strength	0.5ODx0.414 ID x 43.3"	10.690	0.067	0.716	\$35.87	McMaster-Carr Part #:2153T41	2	1.432	\$71.74	\$35.87	
6061 Al	Square Tubes Cross Bar Good/good : Tensile/Impact	1 x 1 x 43.3" 0.0625" thick	10.150	0.1	1.015	\$23.38	McMaster-Carr Part #:6546K53 6ft.	2	2.030	\$46.76	\$23.38	
Foam Spacer	Padding to Protect Plane and Decrease Vibration	24 x 24 x .25"	35.075	0.012	0.406	\$34.03	McMaster-Carr Part #:86375K252	1	0.406	\$34.03		
D.B. Orange	Double/Bubbe Orange Epoxy, 10 Pack High Peel Stgth.	n/a	n/a	n/a	0.000	\$16.00	theepoxysource.com	1	0.000	\$16.00		
Velcro	Industrial Strength Double Sided Velcro to Attach the Frame to the Plane	n/a	n/a	n/a	0.250	\$20.00	n/a	1	0.250	\$20.00		
Zip Ties	Zip ties to Secure the Carbon Fiber Tubes to the Cross Bars	n/a	n/a	n/a	0.000	\$10.00	n/a	1	0.000	\$10.00		
Hardware	Screws, Bolts, Etc.	n/a	n/a	n/a	0.000	\$20.00	n/a	1	0.000	\$20.00		
									Subtotal	5.732	\$218.53	\$58.87
									<b>Total</b>	<b>2.600 kg</b>	<b>\$277.40</b>	

# Motor Selection

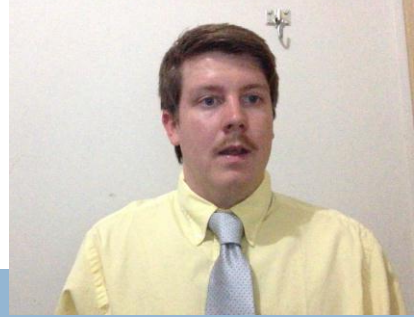


Equipment weight						Total
Frame	Battery (2)	Motor (4)	Prop (4)	ESC (4)	Misc.	kg
8.283	1.080	0.844	0.204	0.216	0.300	10.92

- Calculated weight  
10.92kg
- Safety Factor: 1.5
- Total Thrust Needed:  
16.38kg
- Thrust per motor:  
4.10kg

Calculated Weight \* Safety Factor = Total Thrust Needed





# Motor Selection



Cobra 4510 Multi-Rotor Motor



Tiger Motor MN4120



Tarrot 5008 Motor

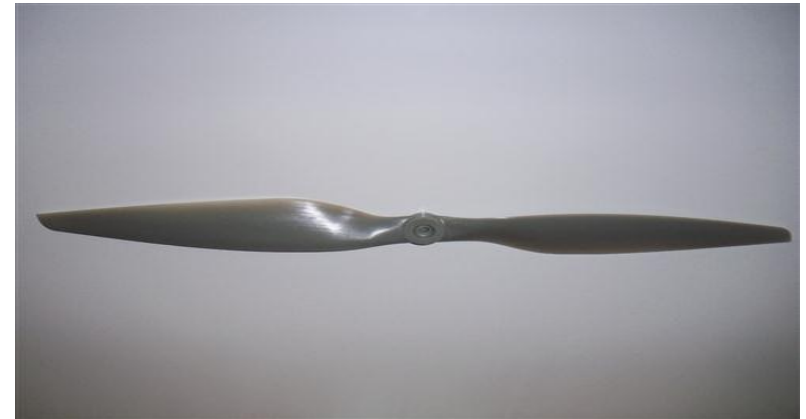
Motor	Cobra	T-Motor	Tarrot
Thrust	4468 g	4280 g	4100 g
Price	\$74.99	\$129.90	\$59.90
Weight	211g	253g	168g



# Motor Selection



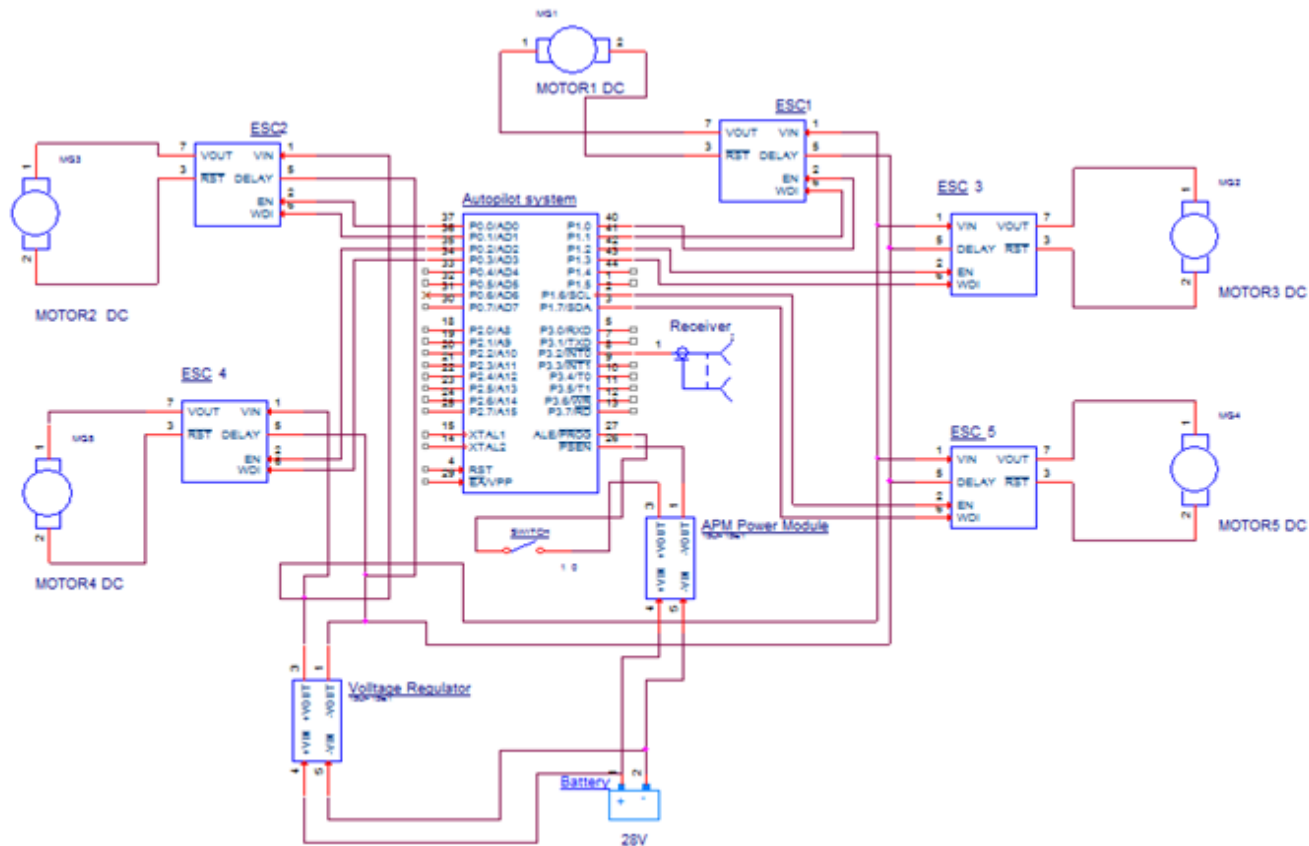
Cobra 4510, KV = 420, Current 35A, Weight 211g



APC 18 x 5.5" Multi-Rotor Propeller

Prop Size	Li-Po Cells	Input Voltage	Motor Amps	Input Watts	Prop RPM	Pitch Speed in MPH	Thrust Grams	Thrust Eff. Ounces	Thrust Eff. Grams/W
18x5.5-MR	6	22.2	38.76	860.5	6,414	33.4	4468	157.60	5.19

# Circuit schematic



# Autopilot Selection

## ***Autopilot selection:***

Ardupilot 2.5

## ***System Features:***

- Fully autonomous waypoint navigation for multi-rotor vehicle
- Failsafe programming options if device loses signal
- Relay real-time telemetry data to ground system



# ESC Selection

## **ESC Selection for Quad-Rotor:** Cobra 60A opto multirotor ESC

### **Design features :**

- Permits device to operate with minimal radio interference at high currents
- If the autopilot system loses signal, the system will automatically switch to idle



# Electrical Power Calculations

**Remaining battery capacity if aircraft land and takeoff for 40s**

$$= (\text{battery capacity} - (\text{discharge time} * \text{current drawn}))$$

$$= (10A - (.0111\text{hours} * 155.04A)) = \mathbf{8.28 A}$$

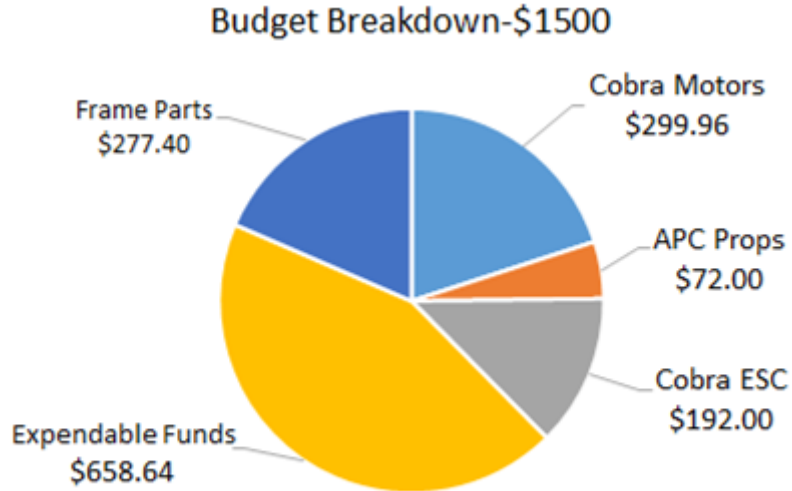
**Hover time = (battery capacity)/(current drawn)**

$$= (8.28 A / 96) * 60\text{min} = \mathbf{5.175 min}$$

$$\text{Total flight time} = 5.175 \text{ min} + .66\text{min} = \mathbf{5.84 min}$$

$$\text{Recommended flight time} = \mathbf{4.6733 min}$$

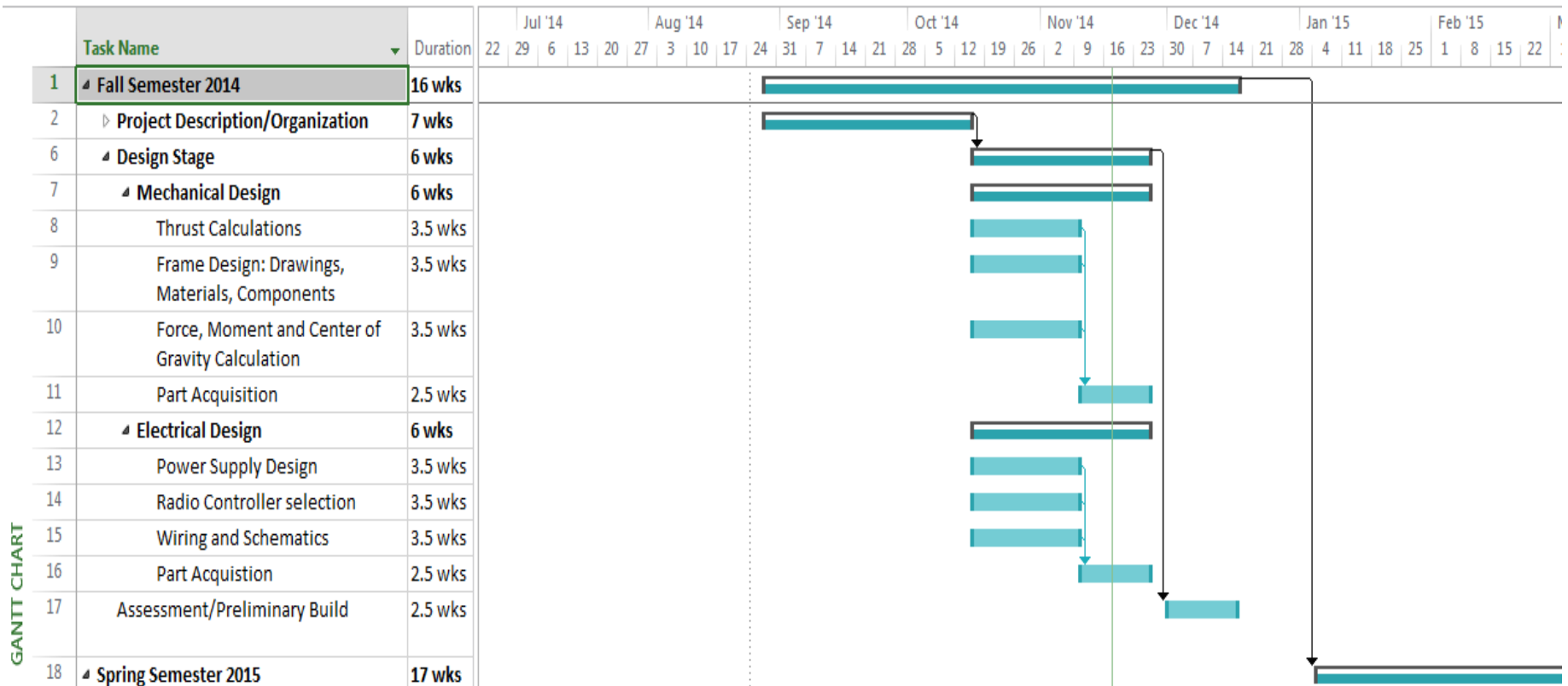
# Cost Analysis



## ***Analysis:***

1. Utilized 56% (\$841.36) of budget
2. Efficient Spending
3. Surplus Added

# Schedule/Gantt Chart





# Future Work

## ***Fall Semester:***

1. Part Acquisition
2. Frame Drawings

## ***Spring Semester:***

1. Manufacturing/Preliminary Build
2. Test Flight/Troubleshoot (Horizontal)
3. Finalized Build
4. Test Flight/Troubleshoot (Vertical)



# Final Summary



1. Improved multidisciplinary and international communication skills
2. Selection of hybrid design
3. Frame design and material selection
4. Motor, Prop, and ESC selection
5. Electrical Components/Power Design
6. Budget/Schedule

# References

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ANY  
QUESTIONS  
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