



# Miniature Bomb Rack Unit



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- Cost Analysis
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- Conclusion

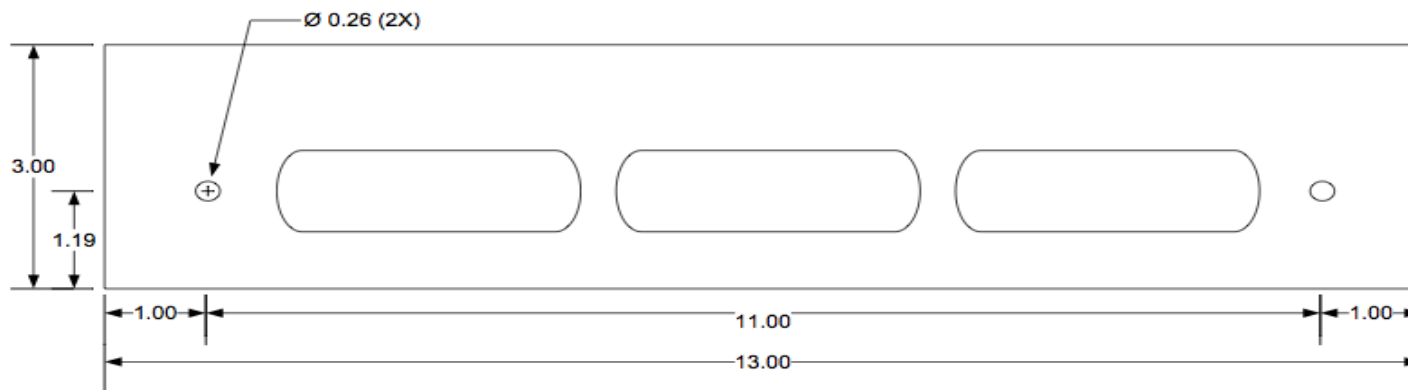
# Problem Statement

- Design and develop a Bomb Rack Unit (BRU) that is attached to the Tigershark UAV capable of housing and launching a cylindrical payload.
- BRU must contain an electrical interface that allows the user to go through a safety sequence before the payload is released
- Construct a working prototype

# Tigershark UAV Platform

## Specifications:

- Wing span 21 feet
- Empty airframe weight - 150 lbs.
- Gross take off weight - 300 lbs.
- Payload capacity – 50 lbs.
- One hard-point location per wing for launcher attachment

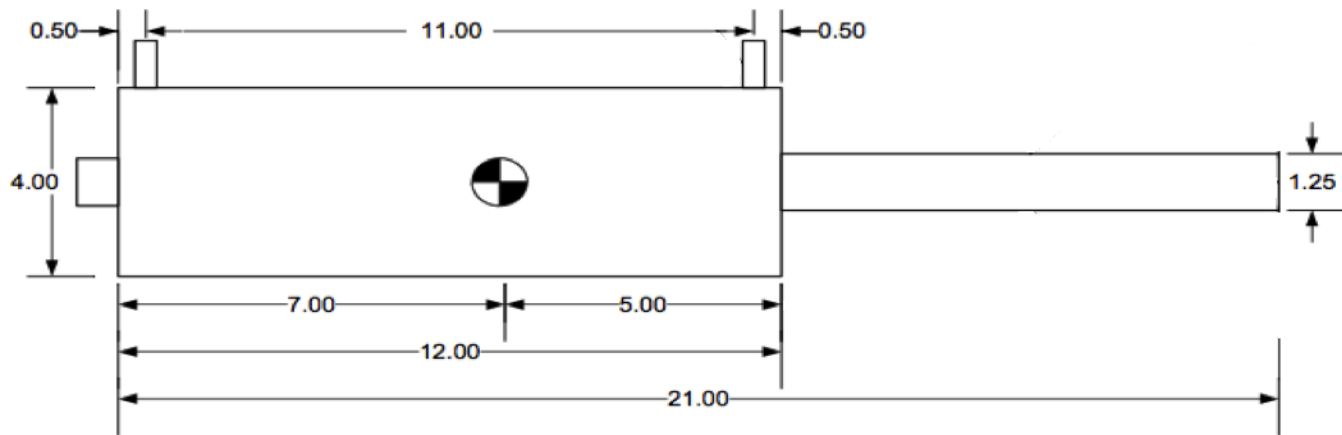


Pylon  
Location

Made from 1" thick Aluminum  
Tolerance +/- 0.05"

# Constraints

- BRU must not exceed 5 lbs
- Capable of holding a payload that is 10 lbs
- Achieve an ejection velocity of at least 4 ft/s
- Retain payload during aircraft maneuvers up to 2G lateral load and 1G landing shock.
- Use lanyard pull to arm payload
- Budget - \$2000



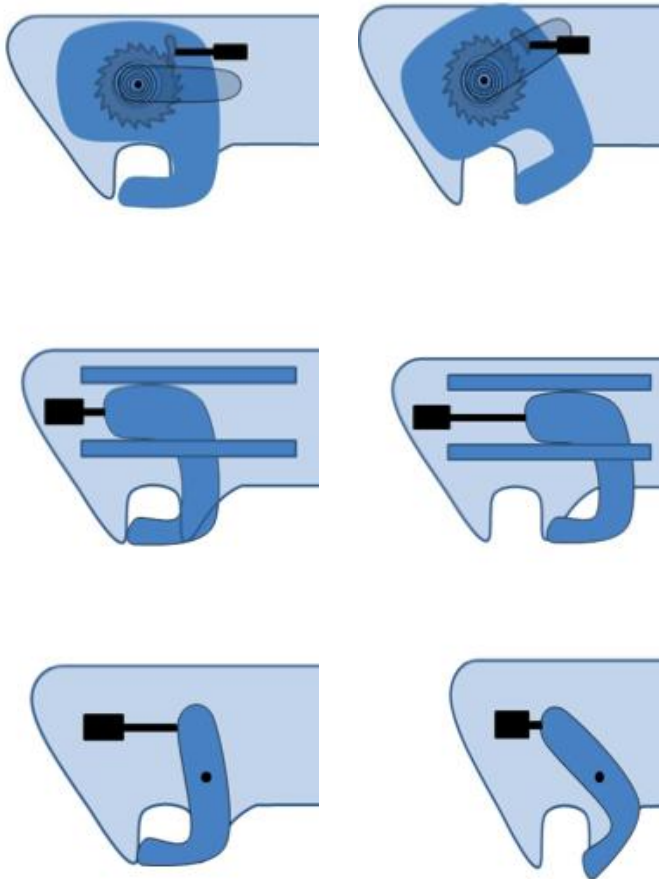
Weight = 10lbs

# Concept Generation

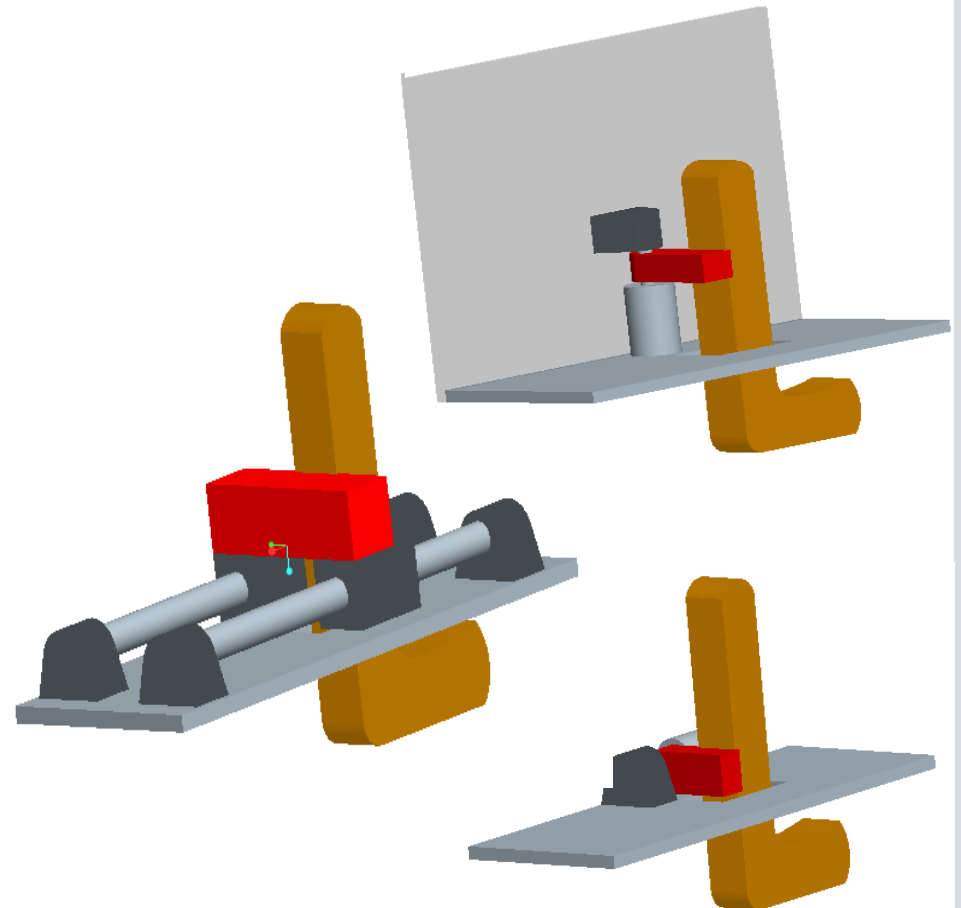
- Design was broken into 4 main subsystems
  - Hook Assembly
    - Carries/Releases Payload
  - Mechanical Safety
    - Prevents any misfires
  - Sway Brace
    - Stabilizes payload during airplane maneuvers
  - Ejector Mechanisms
    - Ejects payload safely from BRU

# Concept Generation

## Hook Release

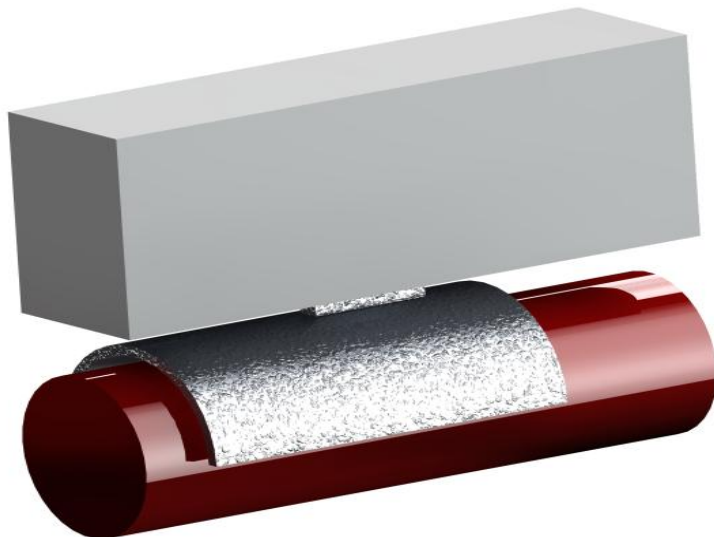


## Safety Block

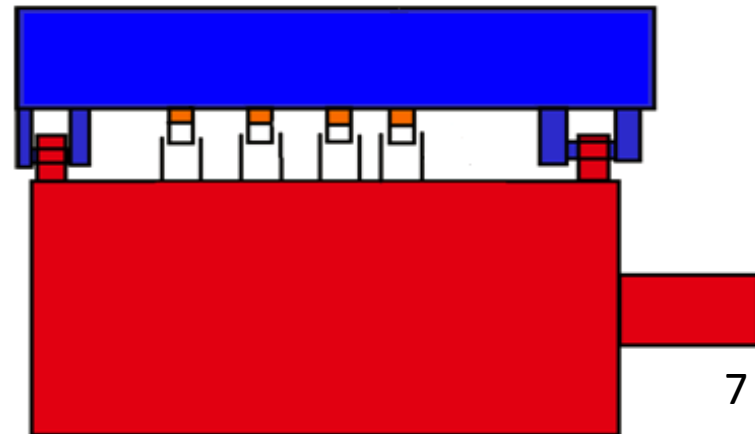
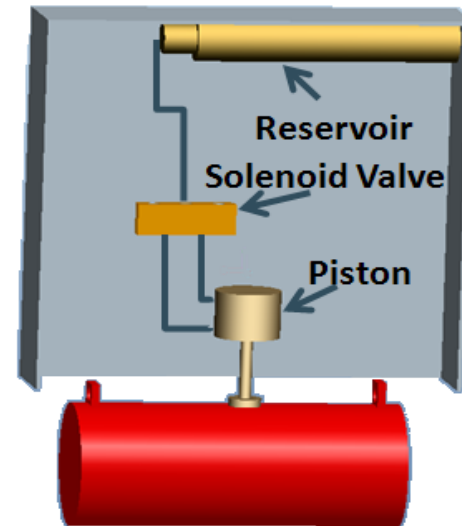


# Concept Generation

Sway Brace



Ejector Mechanism



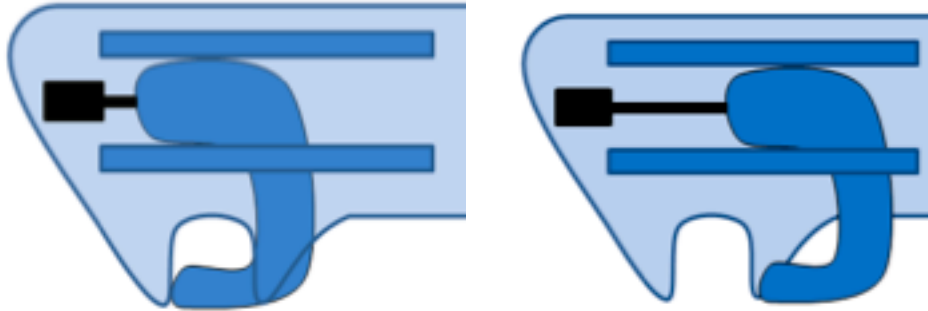


# Selection Criterion

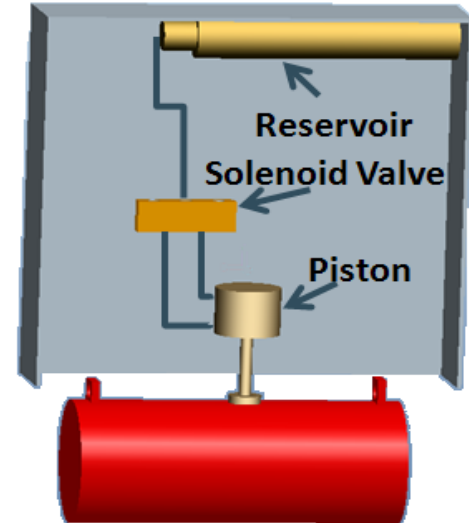
- **Hook Release**
  - Compactness
  - Weight
  - Strength
  - Operational Speed
- **Safety System**
  - Compactness
  - Weight
  - Strength
  - Speed
- **Sway Brace**
  - Weight
  - Load carrying
  - Size Flexibility
  - Simplicity
- **Ejector Mechanism**
  - Weight
  - Size
  - Cost
  - Safety
  - Simplicity

# Final Concepts

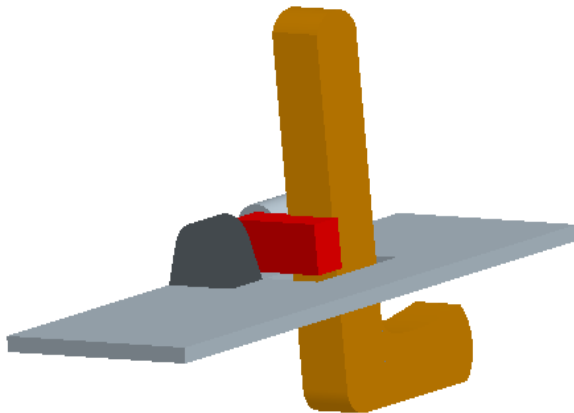
Hook Release



Ejector Mechanism



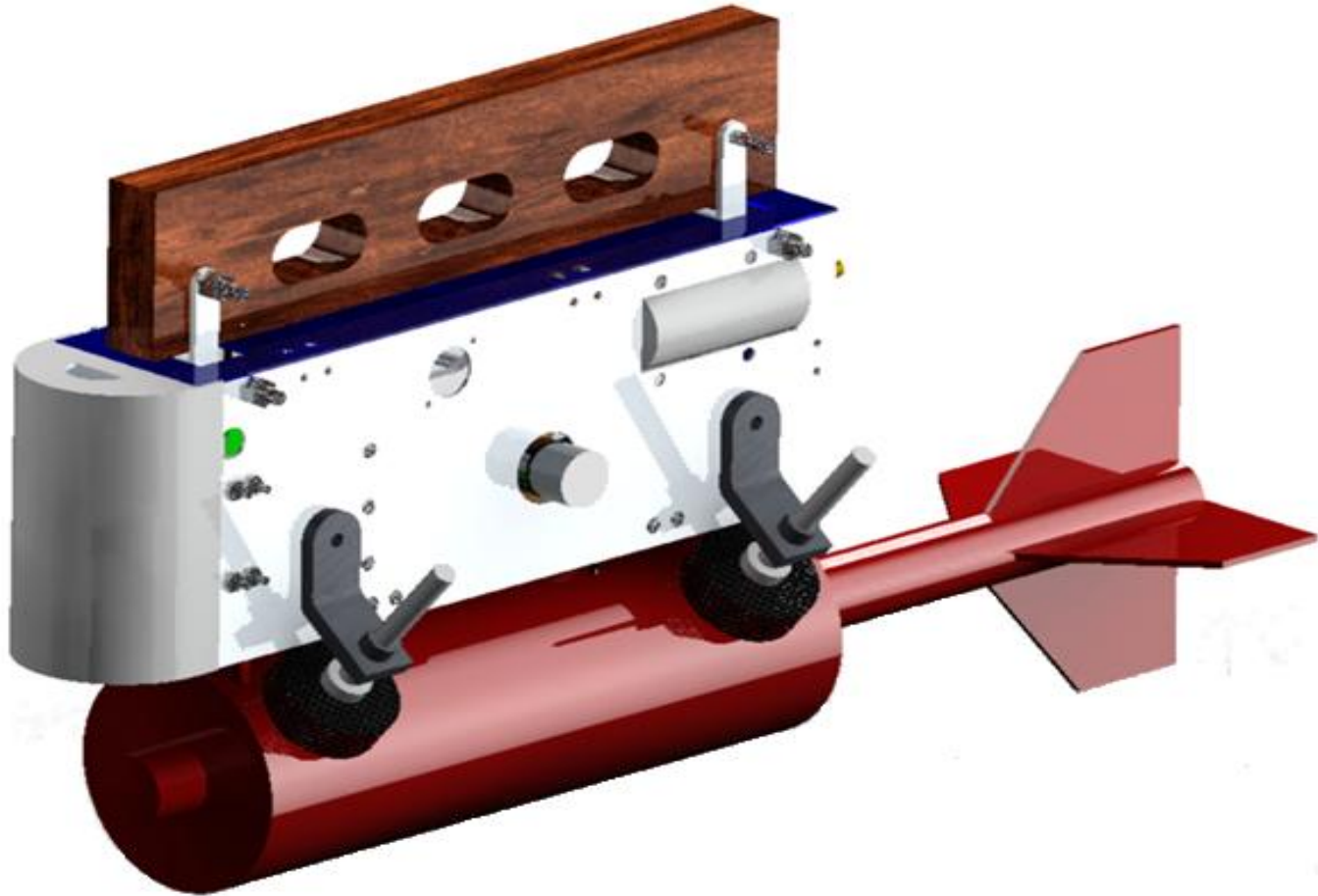
Safety Block



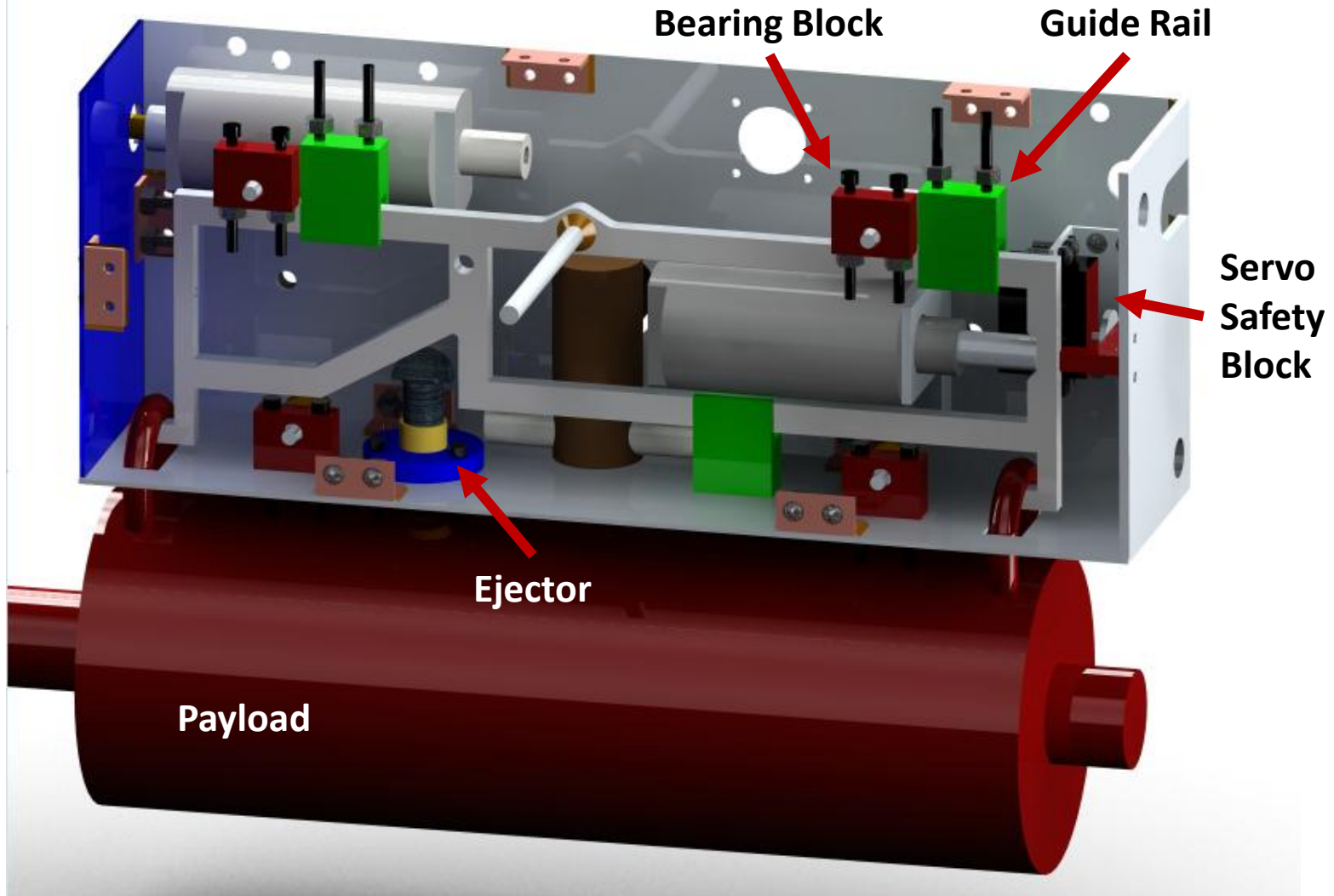
Sway Brace



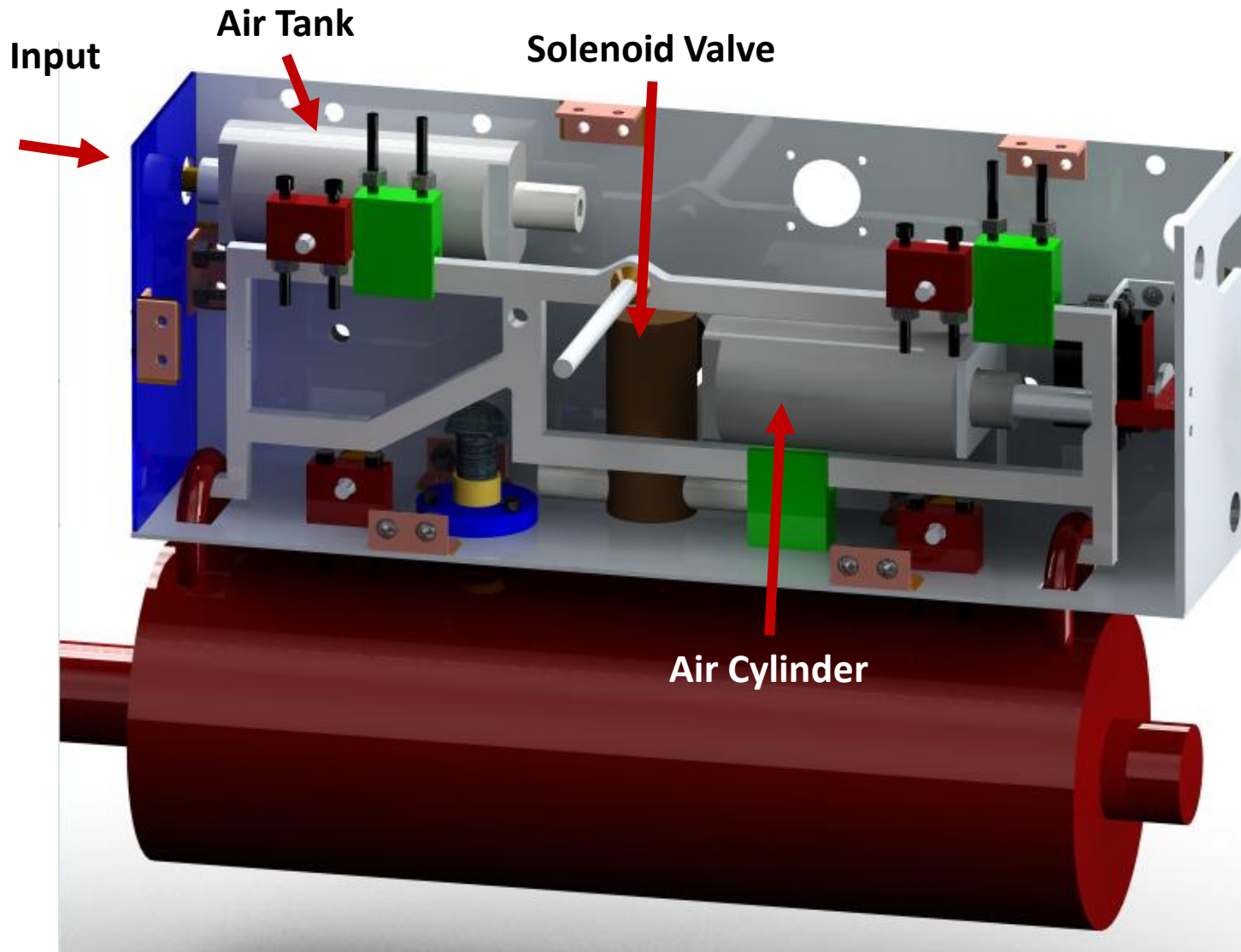
# Final Design



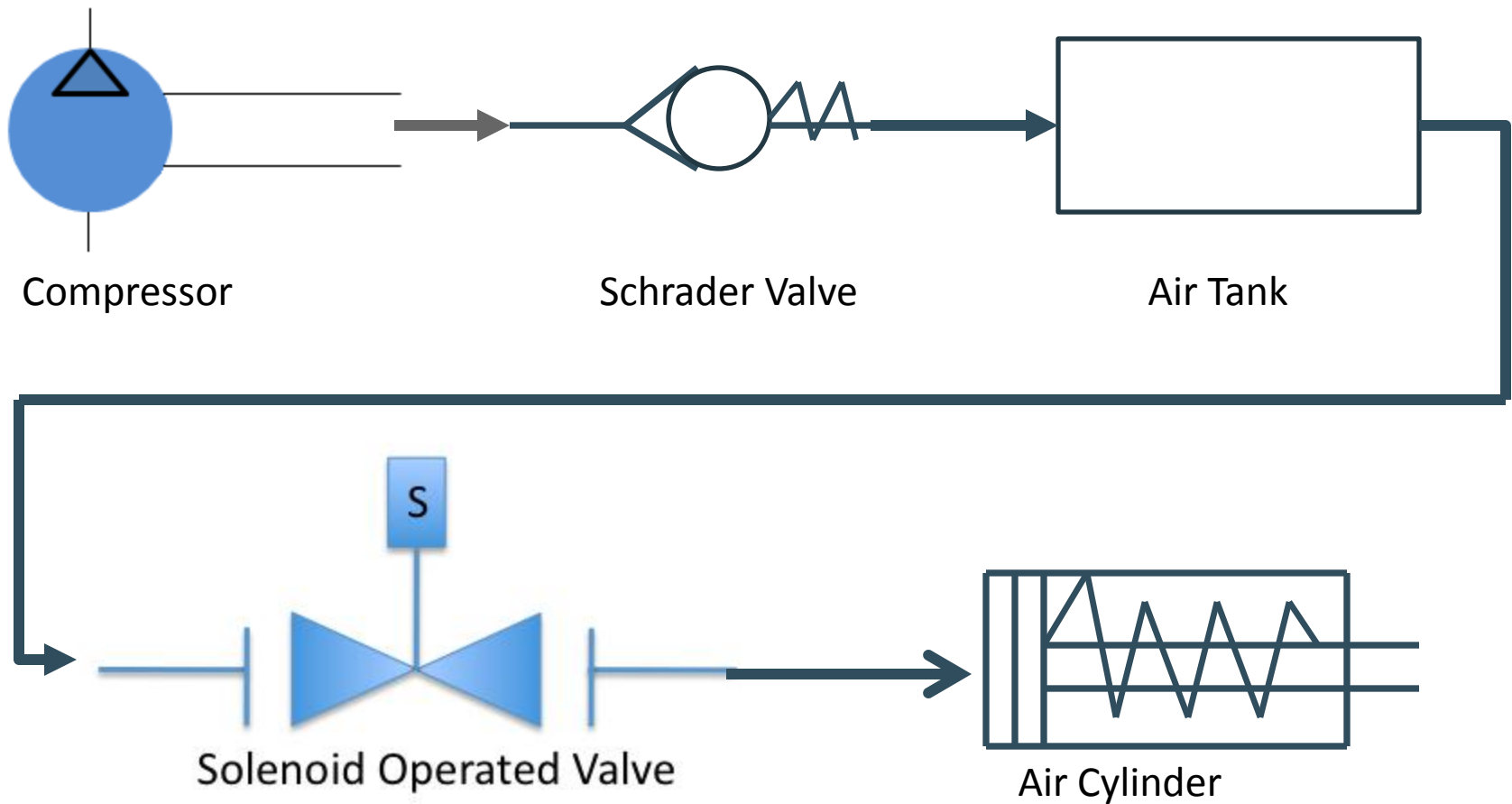
# Final Design (Mechanics)



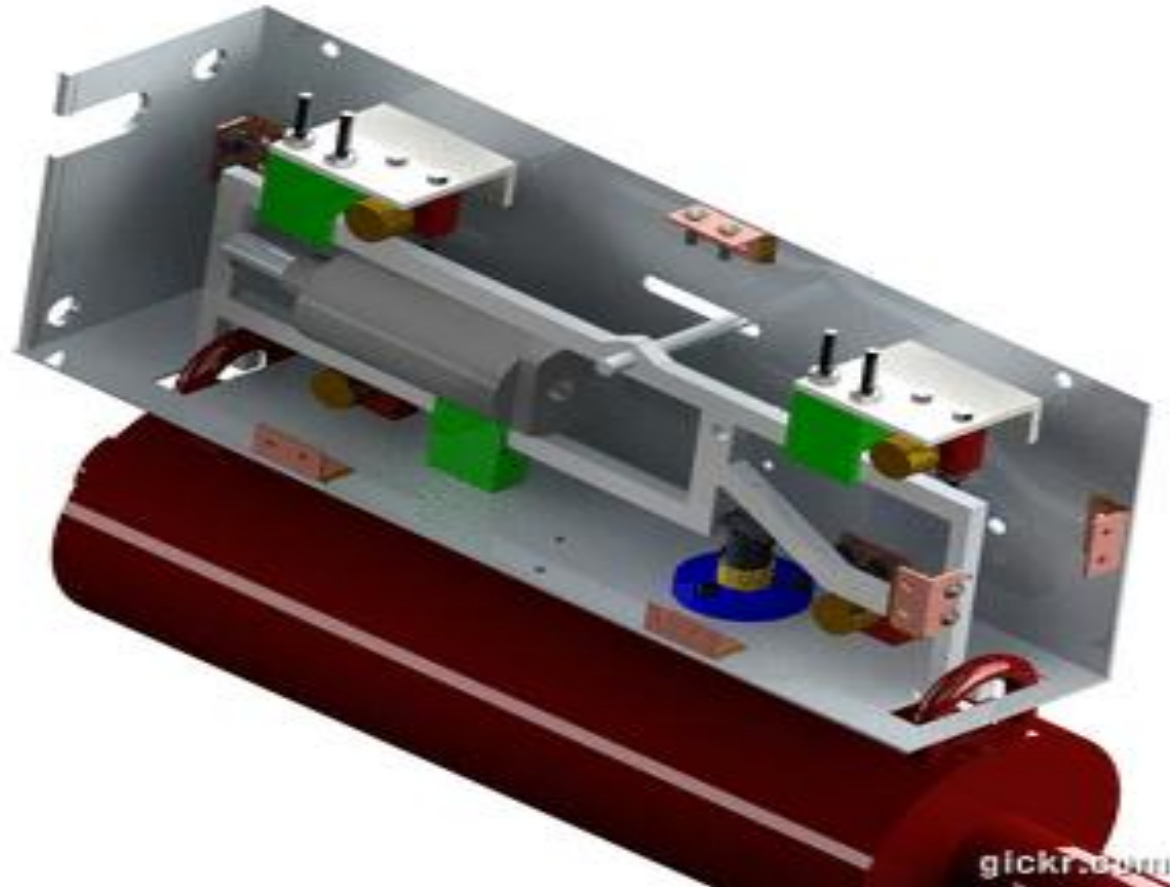
# Final Design (Pneumatics)



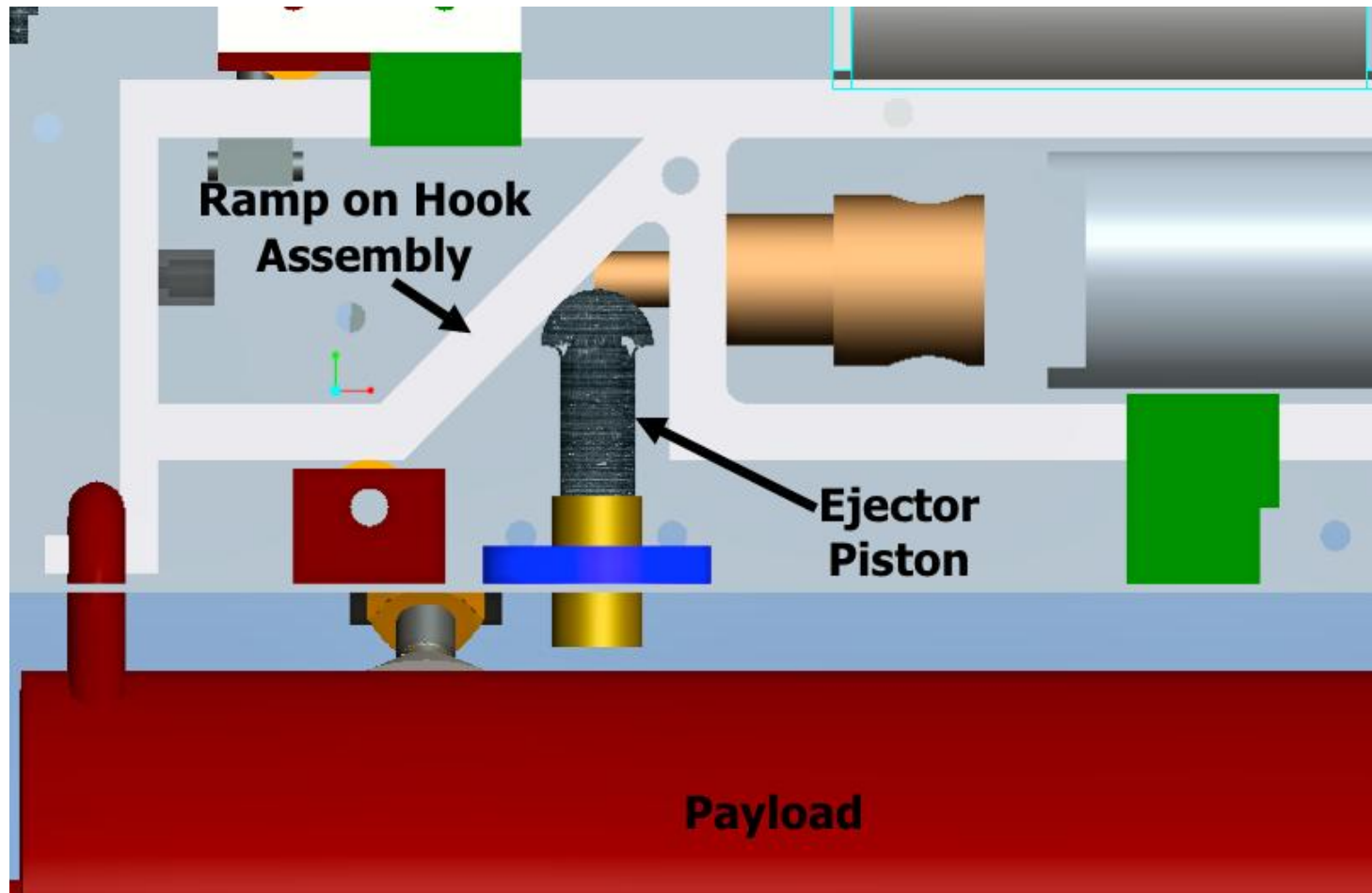
# Pneumatic System



# Hook Assembly

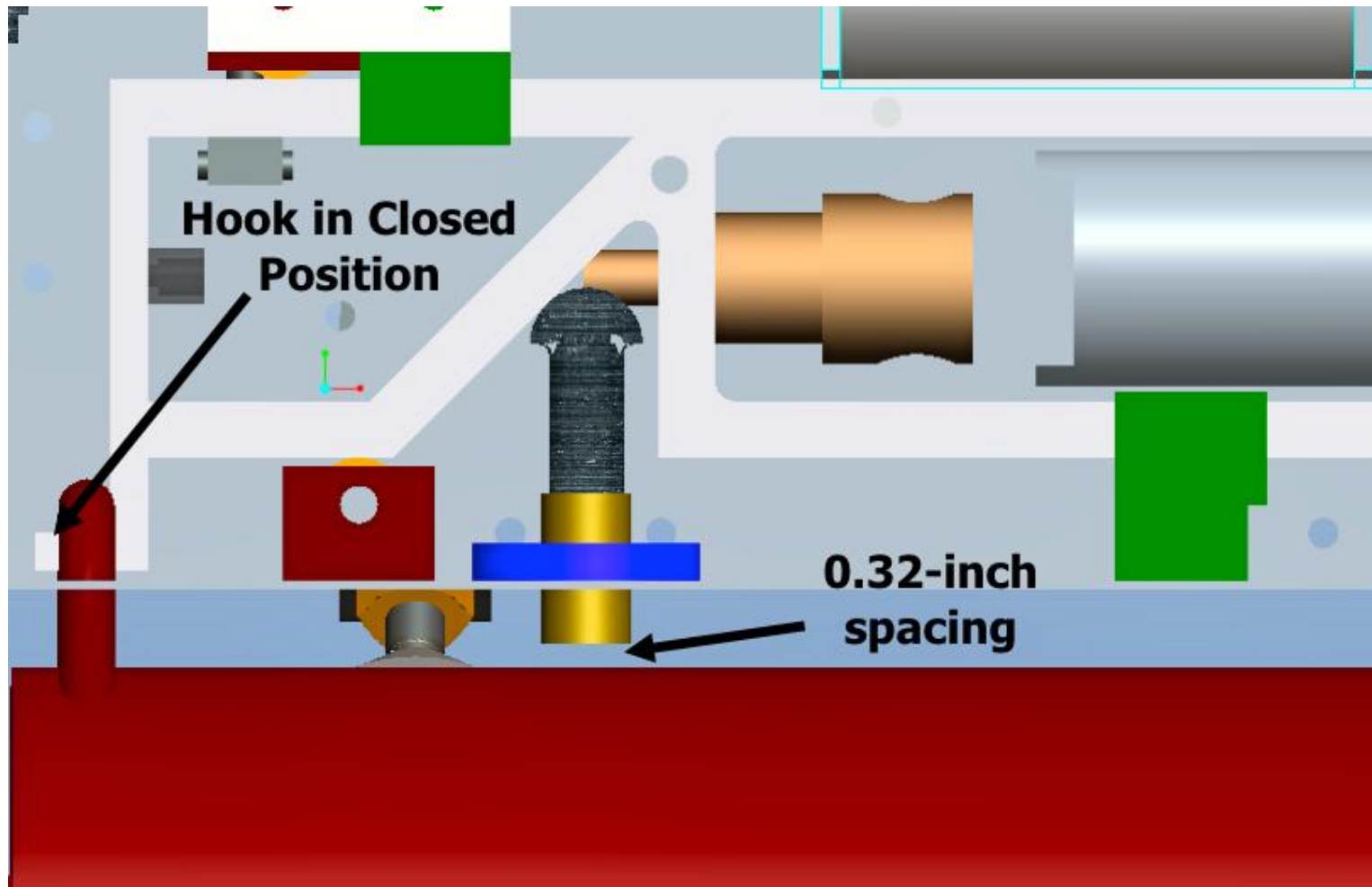


# Ejector System

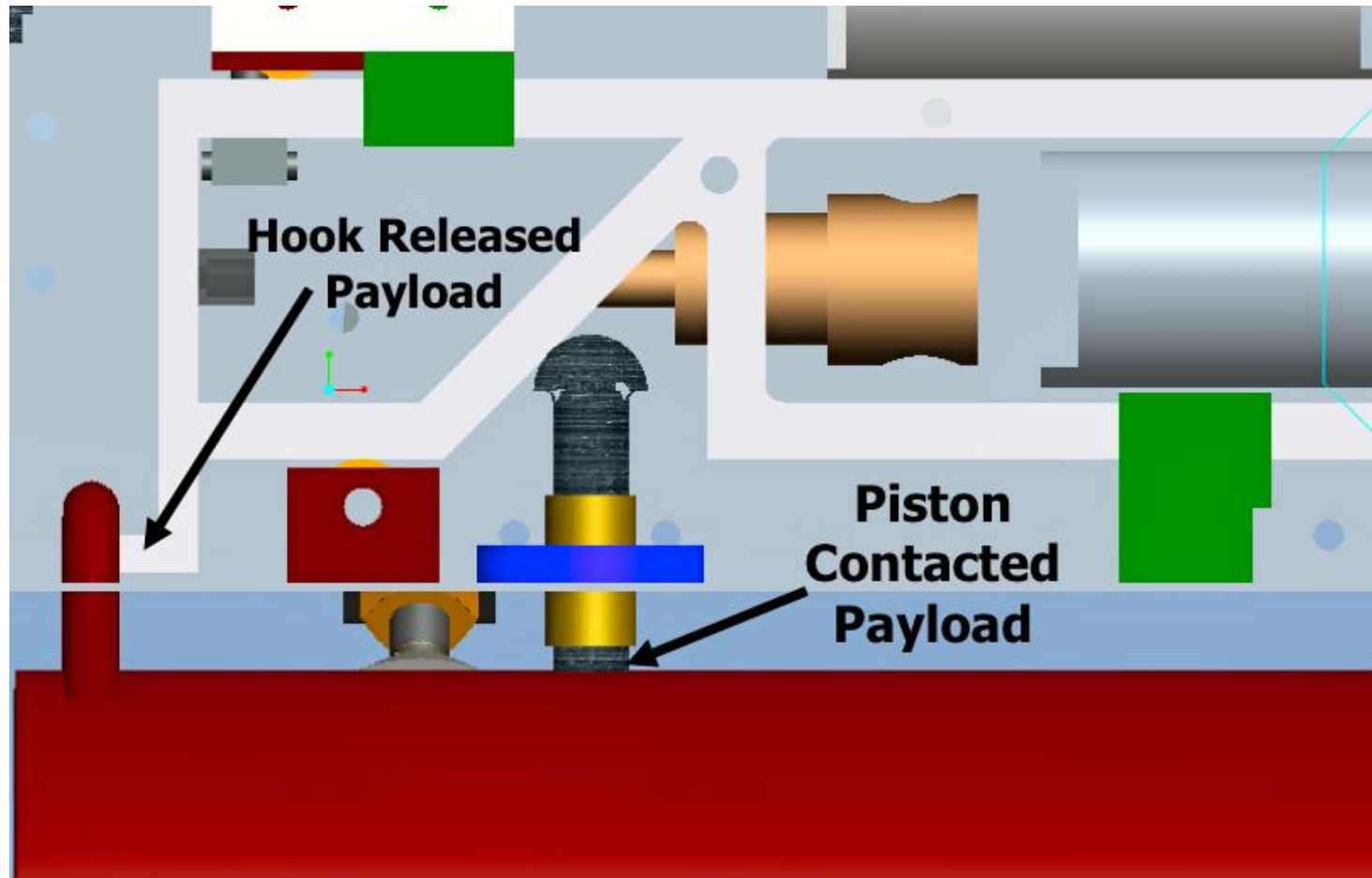




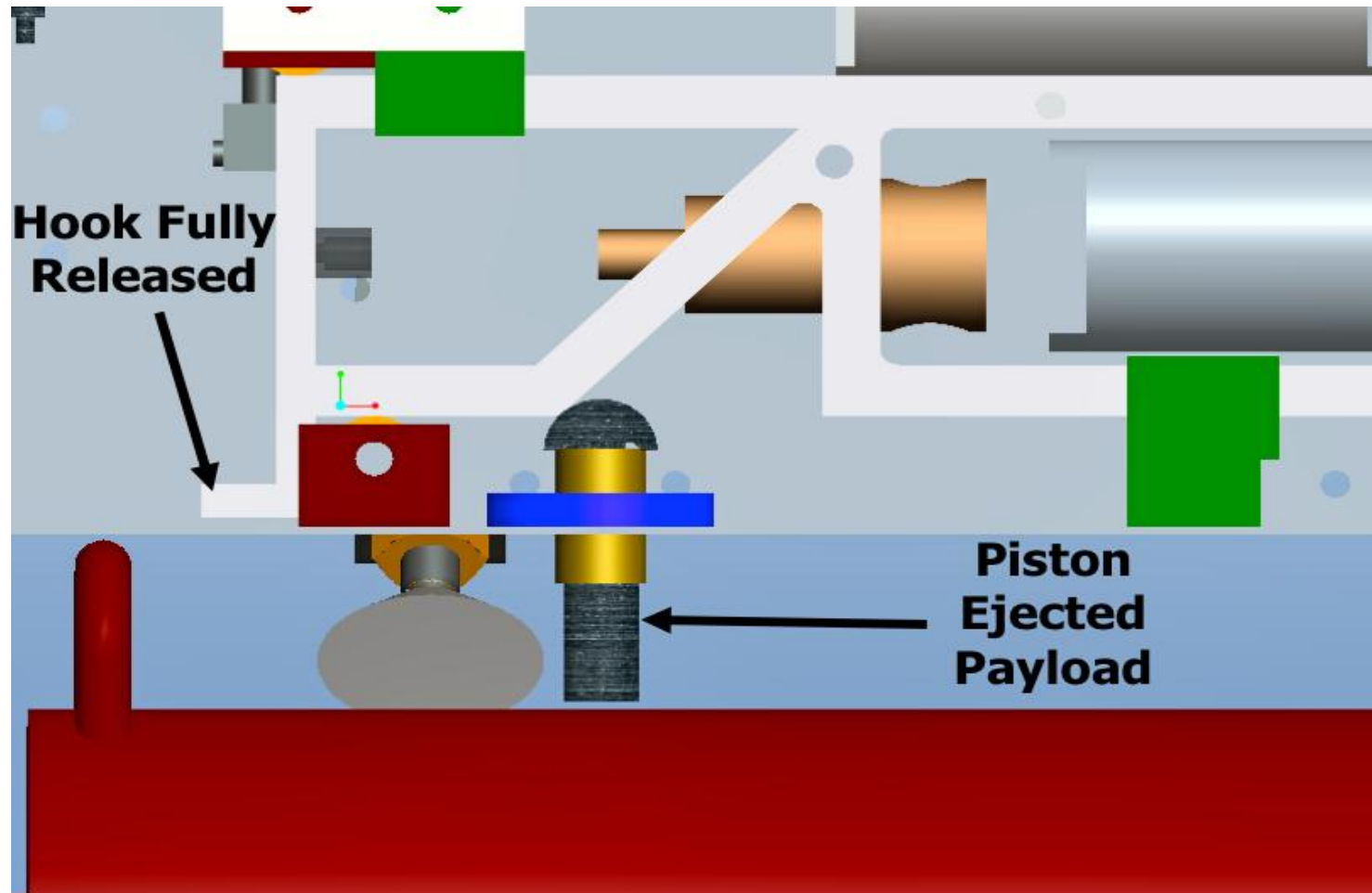
# Ejector System



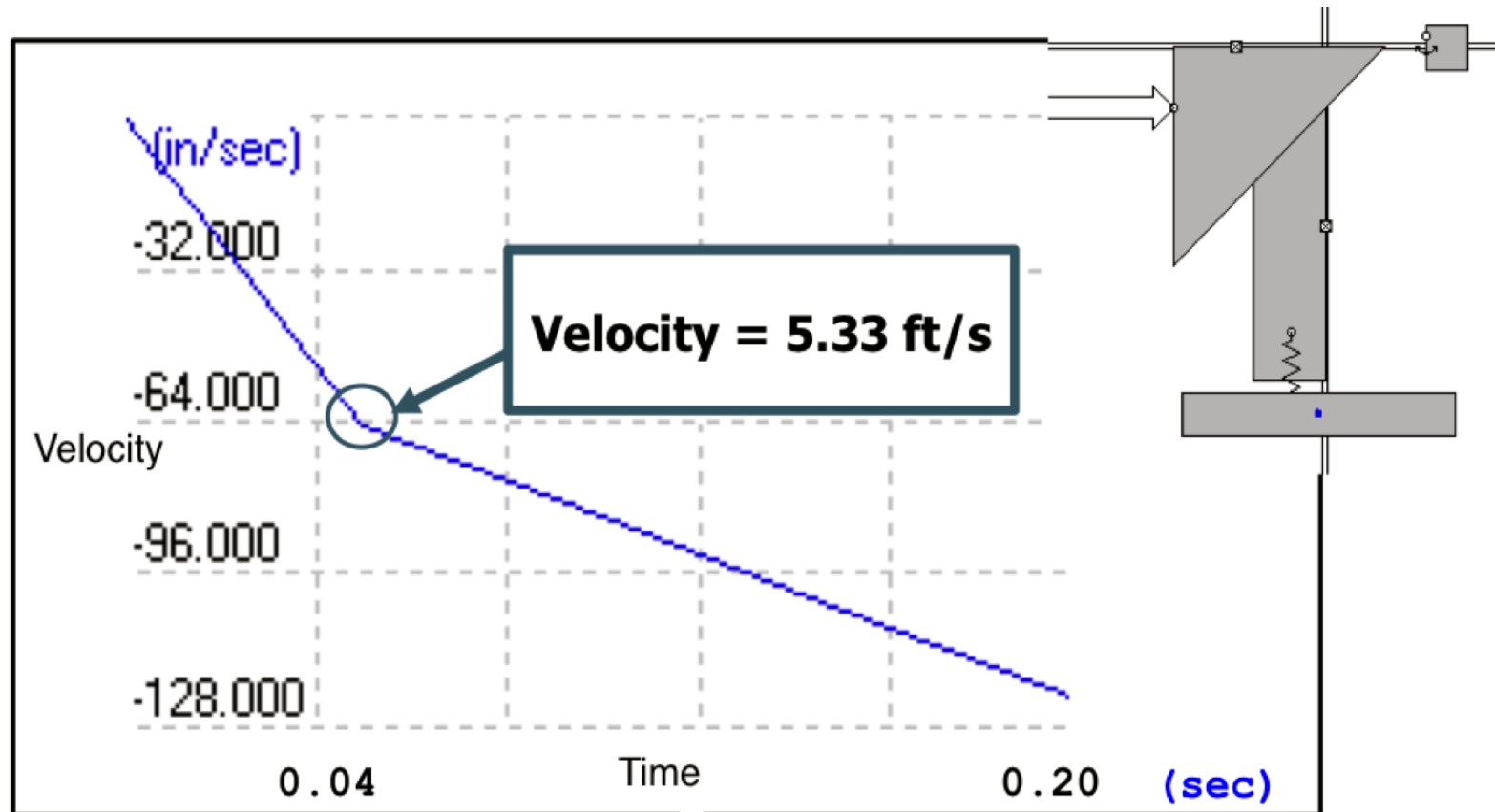
# Ejector System



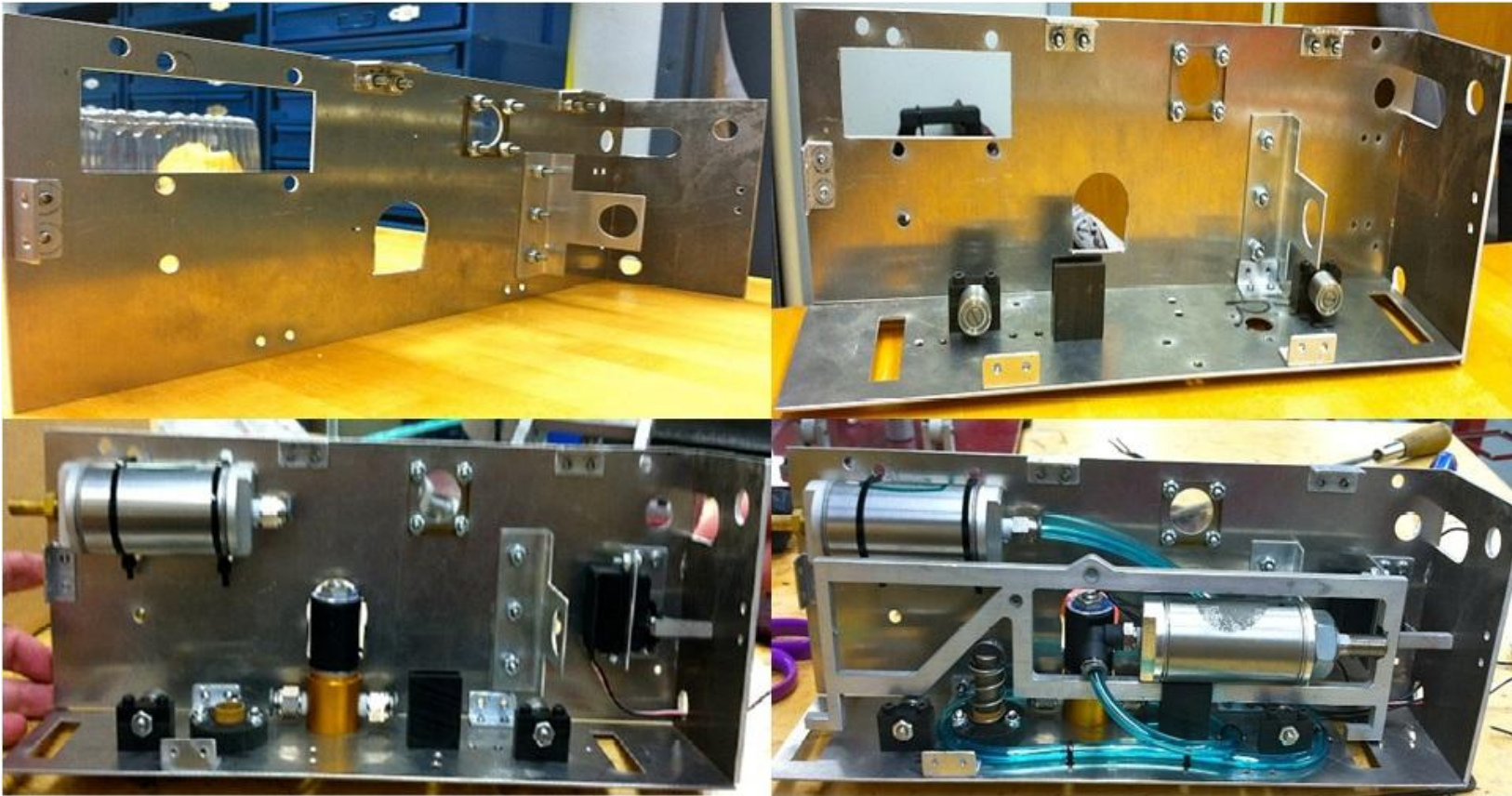
# Ejector System



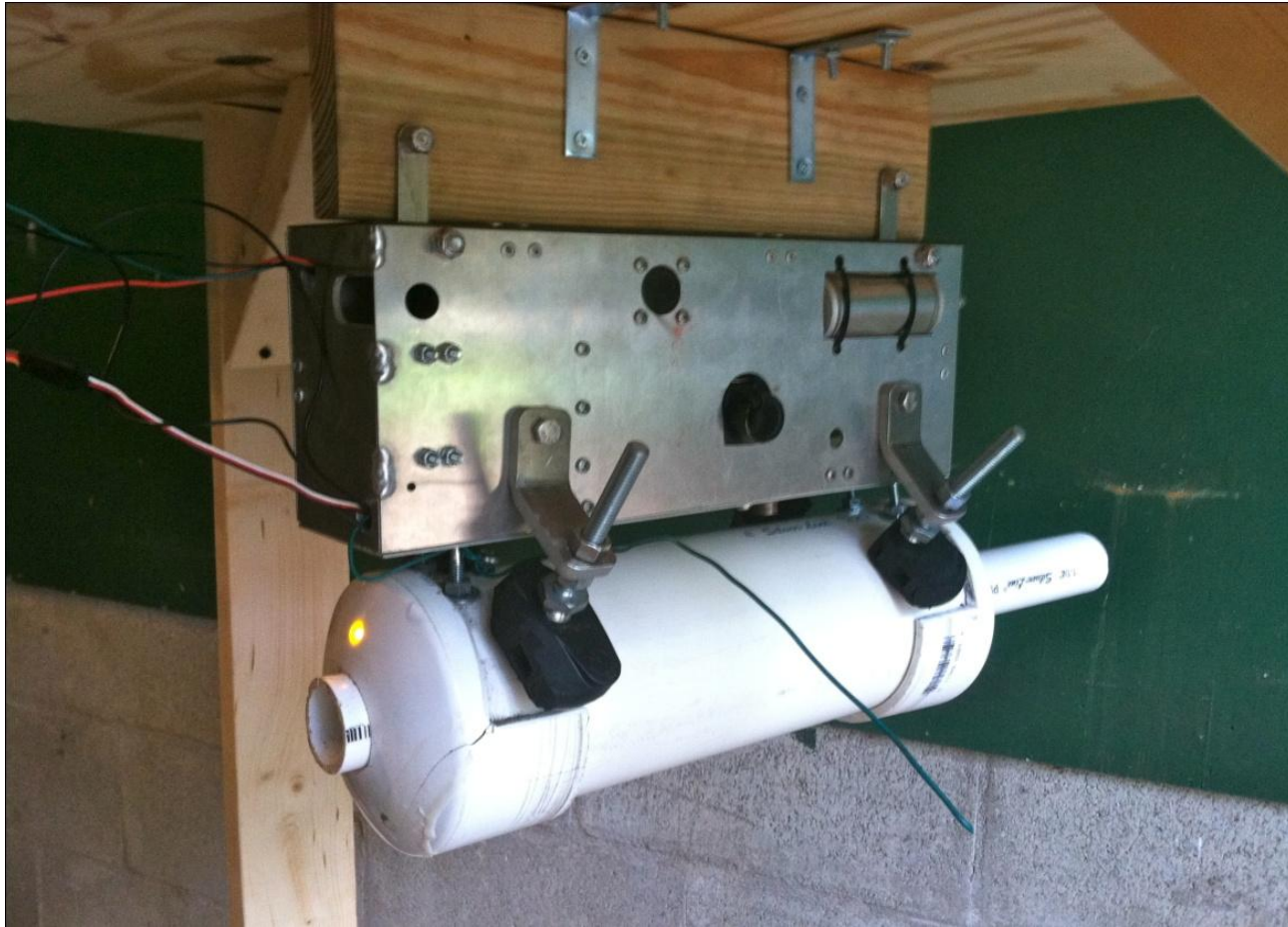
# Ejector Mechanism Simulation



# Prototype Assembly



# Final Prototype



# Payload Release



# Velocity Analysis

- Experimental velocity calculated 5.33 ft/s
- Initial velocity of 4.31 ft/s
- Pneumatic system filled between 60 – 70 psi
- High friction in ejector
- Measuring instruments
  - Apple iPhone 4
  - iMovie

<b>Velocity Test</b>	
	V.Initial (ft/s)
Test 1	3.50
Test 2	4.72
Test 3	4.72
Average	4.31

$$d = v_i t + \frac{1}{2} a t^2$$



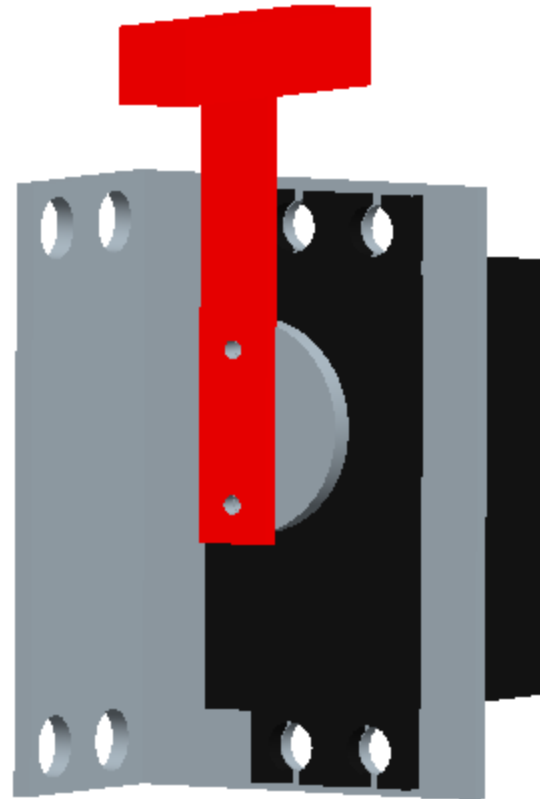
# Sway Brace

- Withstand lateral and vertical loads
- Bracket machined AL6061
- Safety Factor of 1.5



# Safety System

- Required Torque= 0.0066 lb-in
- Output Torque= 4.75 lb-in
- Block made of Al6061
- Servo weighs 15/16 oz
- Requires 4.8V input



# Safety System



# Landing Shock Safety Block

- Tests were conducted, with no empirical data recorded
- With proper test rig a quantitative G-force can be found

$$A_{stop} = \frac{V^2}{x}$$

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## Safety Block

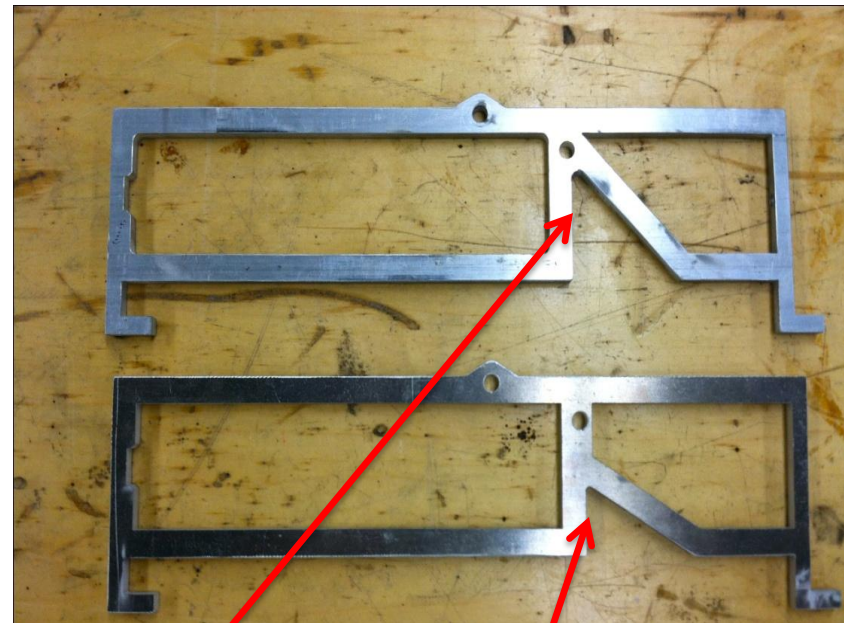
- Safety block easily rotates in out of position
- Stops BRU from any misfires

# Weight Analysis

- Final weight of Prototype – 5.4 lb
  - Pneumatic system used stainless steel
  - Aluminum frame
- To lower weight for future designs
  - Composite materials
  - Increase overall budget

# Issues and Solutions

- Ejector piston jamming
  - Slot angle reduction
- Minor hole tolerances
  - Increased with drill
- Safety block touching side walls
  - Filed stop block
- Assembly difficulties
  - Difficult to place nuts on screws
  - Suggest using threaded holes



**45 Degrees**

**30 Degrees**

# Budget Analysis

- Overall spent \$1304.55 out of \$2000 budget
- Excluding testing and unused components total cost of BRU = **\$688.74**
- Major Purchases
  - Raw Materials - \$102.54
  - Pneumatic System - \$147.22
  - Compressor/Tooling - \$205.12
  - RC Controller -\$107

# Conclusion

- Prototype was an overall success
  - Achieved required ejection velocity
  - Met safety constraints
- Stayed well under provided budget
- Suggested future improvements
  - Composite materials
  - Bearing ejector





# Acknowledgement

- Special Thanks to...
  - Eglin Air Force Research Lab
  - Mr. Russell Roberts (Sponsor)
  - Dr. Clark (Faculty Advisor)
  - Dr. Shih (Course Instructor)
  - Dr. Kosaraju (Course Instructor)
  - Dr. Dalban-Canassy (Course Instructor)

# Questions ????

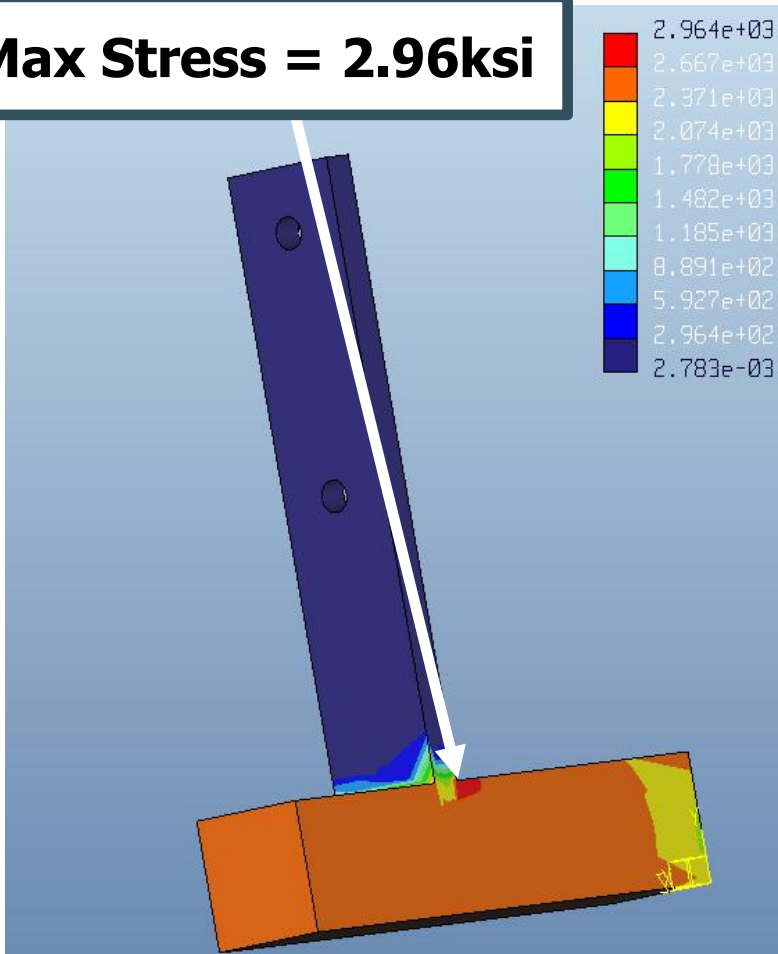


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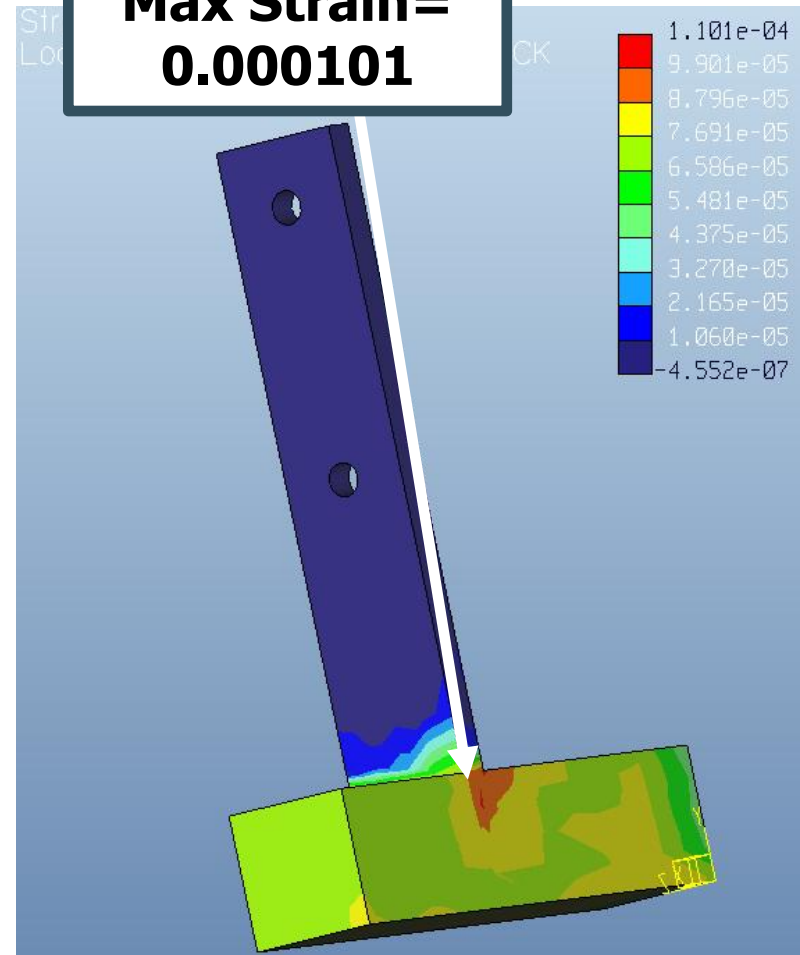
# Safety System

**Max Stress = 2.96ksi**



Von Mises Stress

**Max Strain = 0.000101**

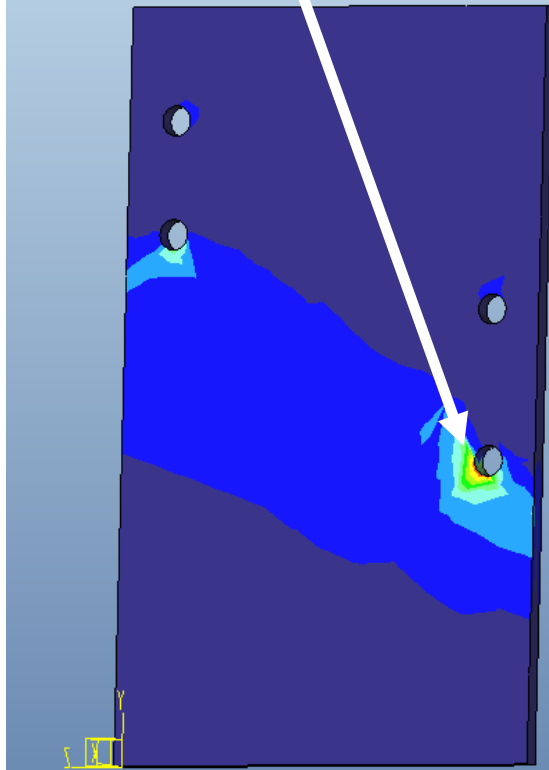
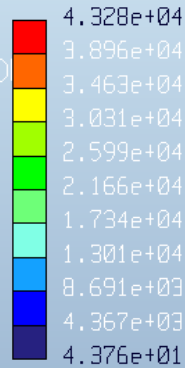


Strain

# Safety System

Stress von Mises (WCS)

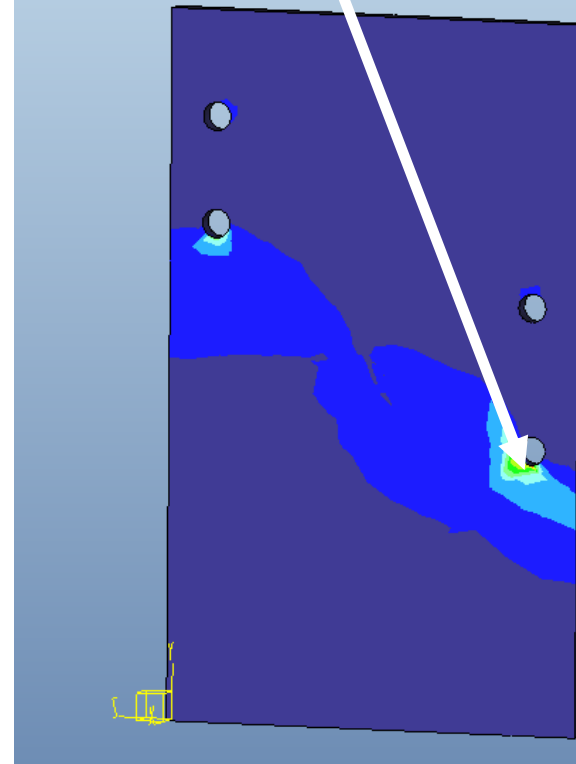
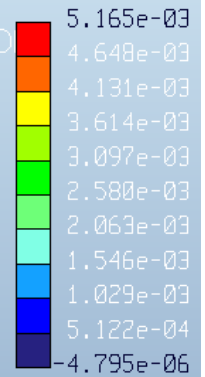
**Max Stress = 43.3 ksi**



Von Mises Stress

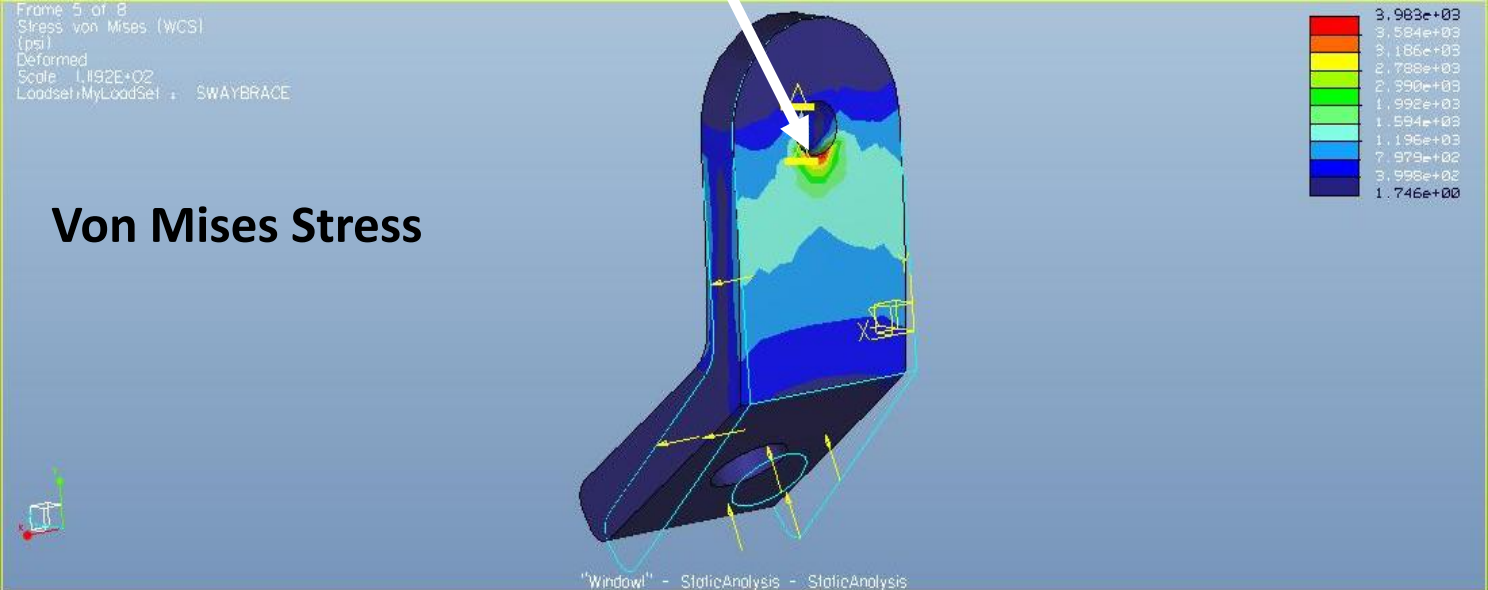
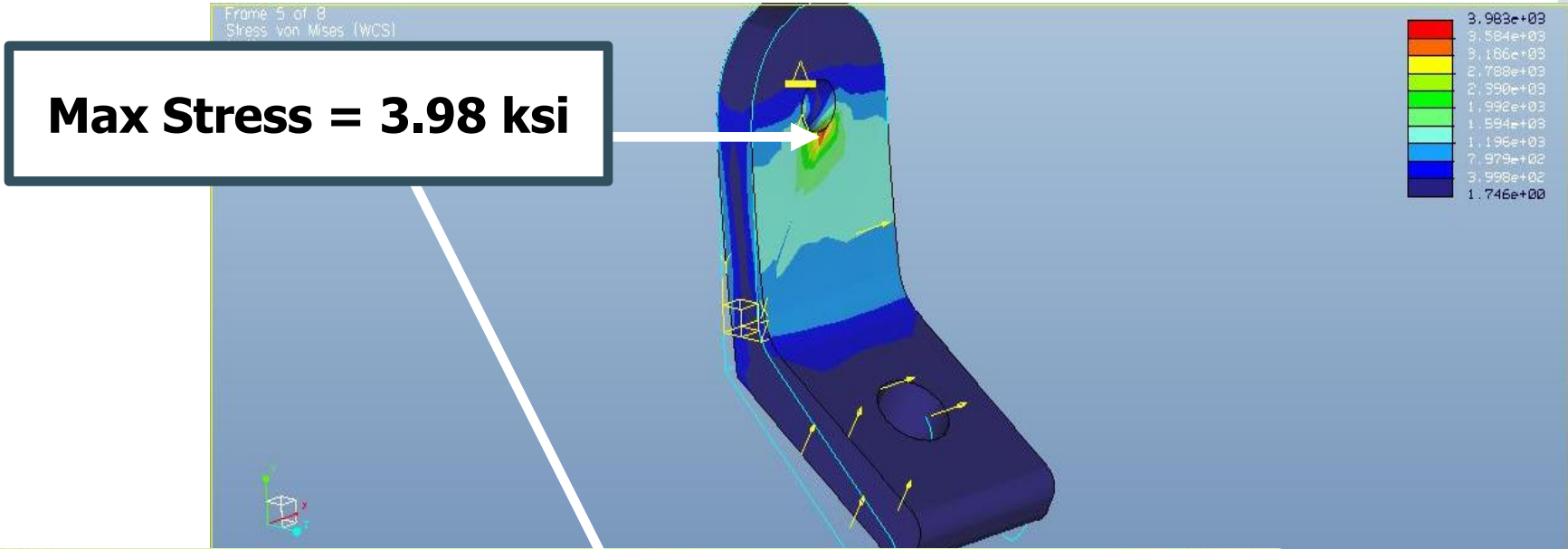
Strain von Mises (WCS)

**Max Strain = 0.005165**

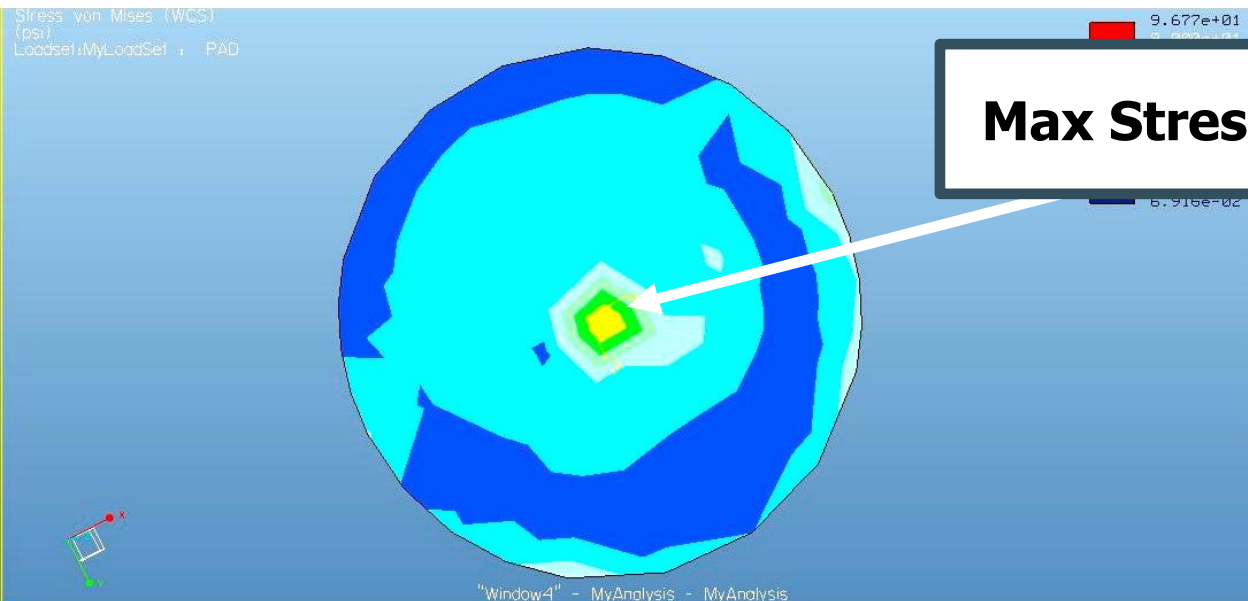


Strain

# Sway Brace (Bracket)



# Sway Brace (Pad)



Von Mises Stress

