



Compact Pneumatic UAV Launcher



FAMU - FSU College of Engineering
Sponsored By Eglin Air Force Base

Launch Team - Group 3

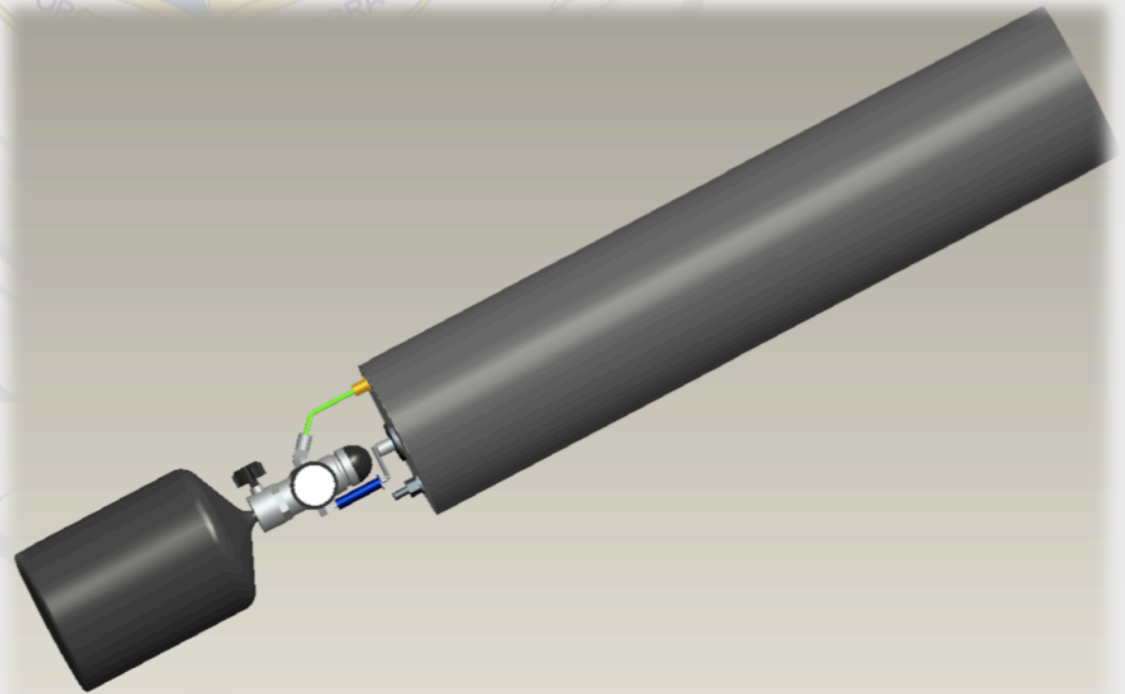
Enye' Blocker
Jared Rodriguez

Timothy Bartlett
Adewale Adelakun



Overview

- Problem Statement & Background
 - Design Specifications
 - Progress To Date
 - Interim Design
 - Anticipated Prototype
 - Experimental Plan
 - Final Design
-
- Assumptions
 - Theoretical Calculations





Explanation

Problem Statement

Eglin Air Force Base needs a safe, efficient, and effective method of launching their current UAV prototype into flight.

Needs Assessment

The objective of this design is to provide a means of propelling an Unmanned Aerial Vehicle (UAV) into flight, which will be provided by Eglin Air Force Base.



Launch Team - Group 3

Background

- Currently launched by hand, tubing, or other aircraft
- Methods do not meet the EAFB standards
- Better means of launch must be developed

Thursday 12-04-2008



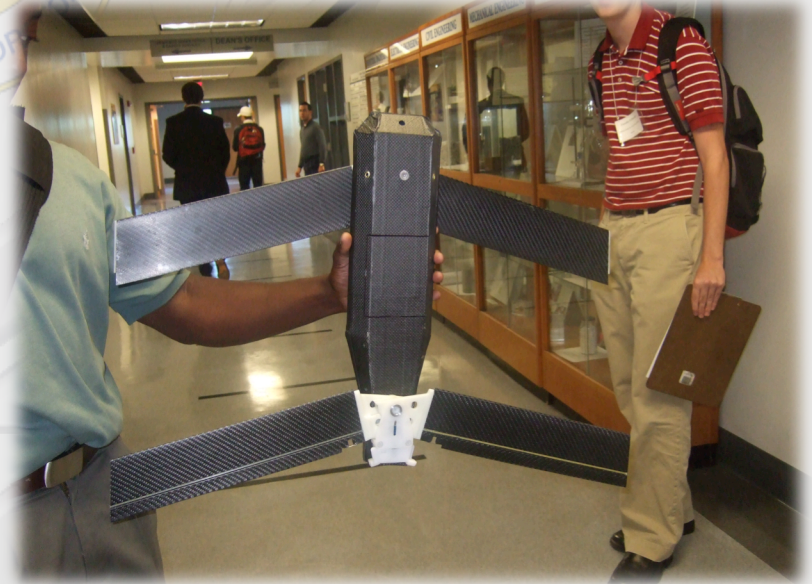
Design Specifications

Launcher

- 60 ft/sec min exit velocity
- Max 600Gs Instantaneous acc
- Launcher weight limit: 2.5 lbs -> 5.5lb (due to requisition and approval)
- Estimated 30-45 deg launch angle
- No energetic methods or accelerants
- Must be repeatable a min of 5x
- Customer prefers a tubular design
- Max dim 36" L x 4.5"W x 4.5"H square or 36" L x 5.5" OD
- Unimproved surface conditions

UAV

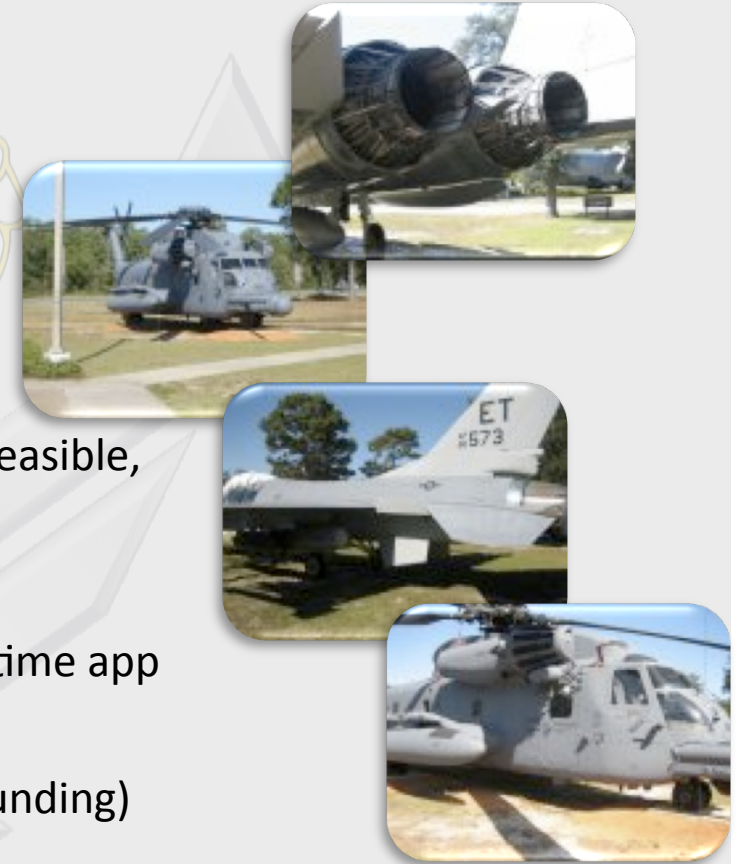
- Approx 3.5lb fully equipped
- Estimated 18" L x 4" W x 2.5" H in launch position





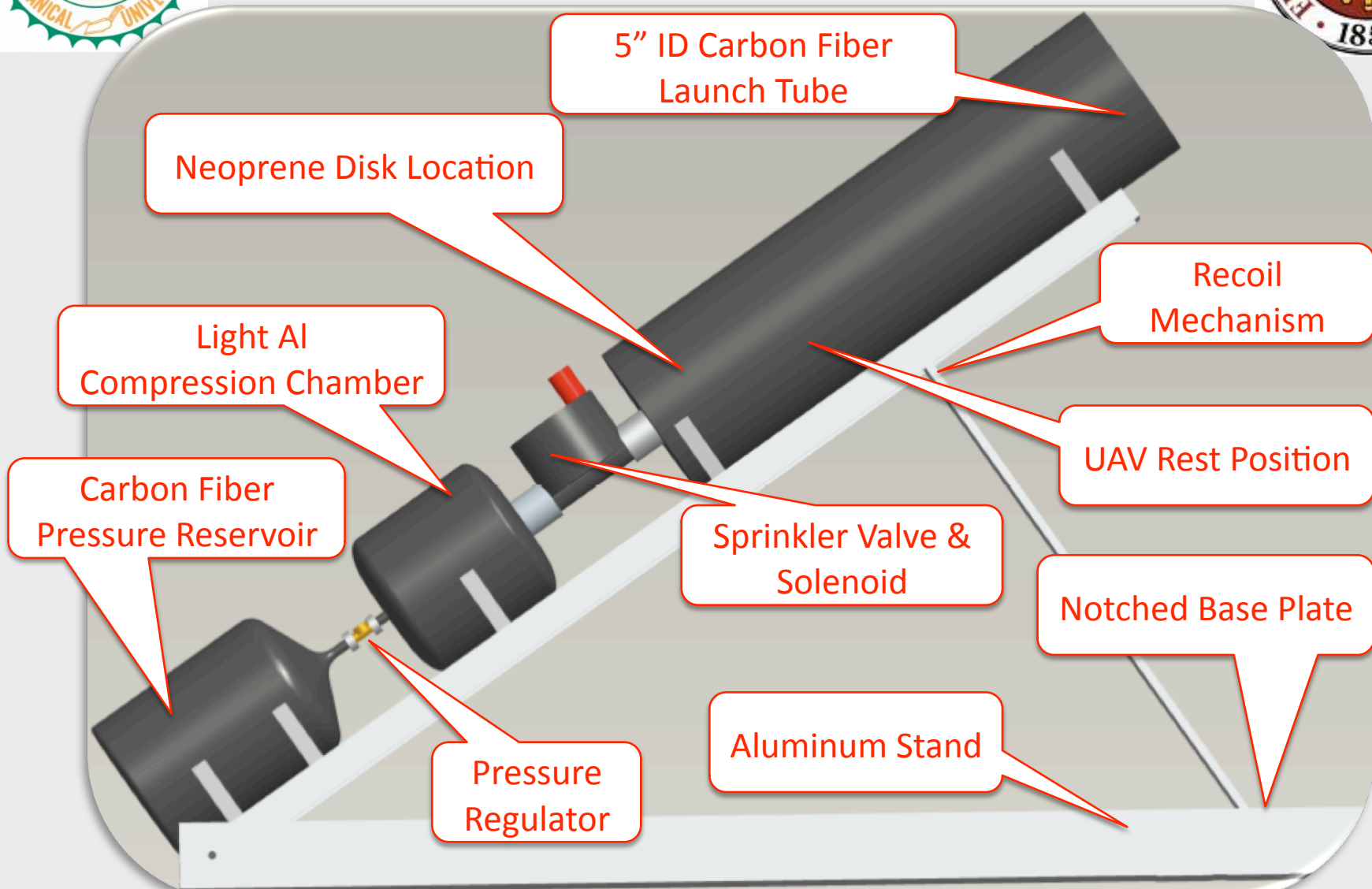
Progress to Date

- Thorough research
- Proper theoretical mathematical analysis
- Completed a feasible conceptual design
- Altered the conceptual design to a more efficient, feasible, & safe interim design
- Visited sponsor to assess situation & visualize real-time app
- Derived a feasible prototype for testing (awaiting funding)
- Optimized the interim design to more effectively meet requirements





Interim Aspiration



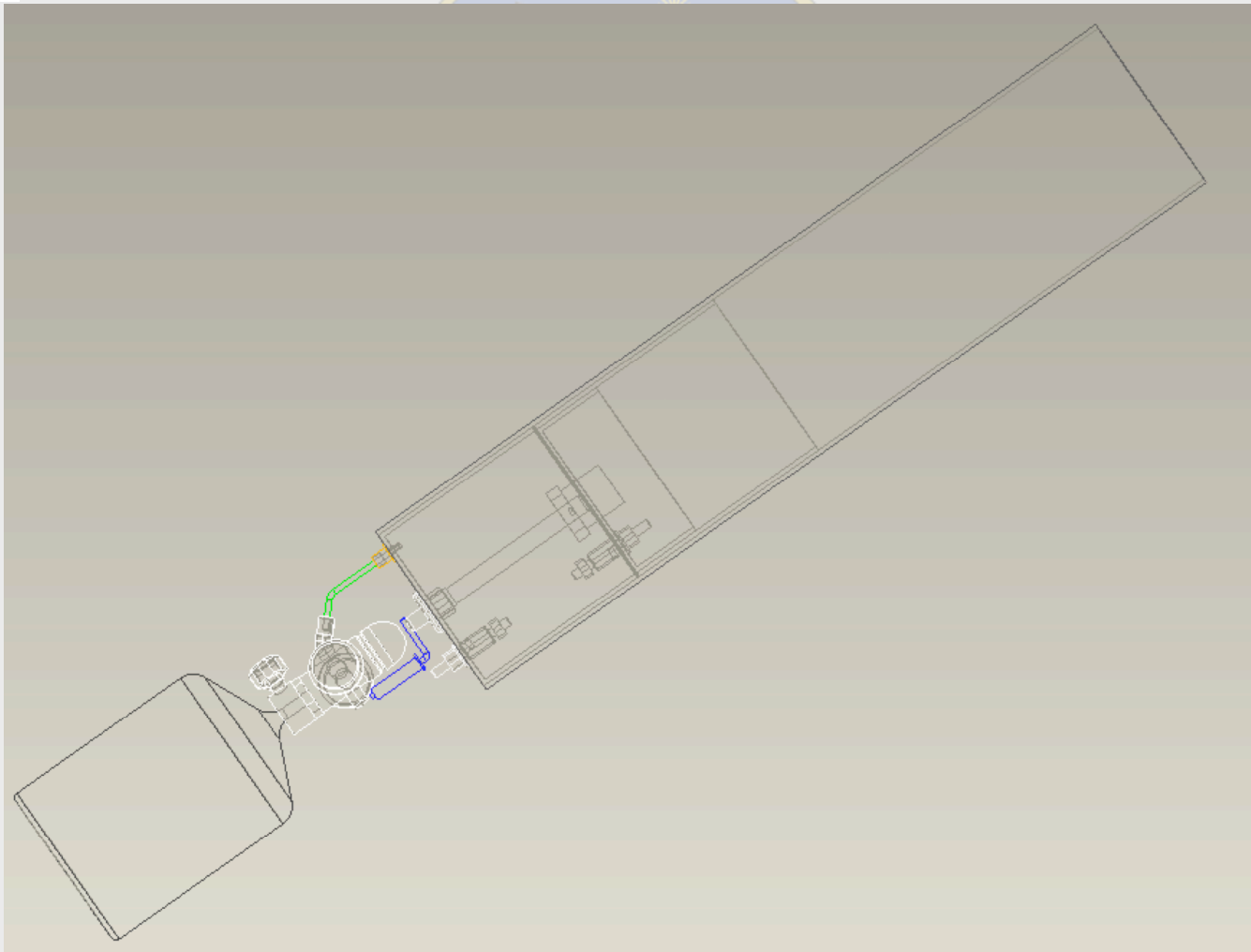


Final Design Changes

- Implemented
 - Rear positioned quick release push pin
 - Tethered cable stopping mechanism
 - Prefabricated bipod legs as a stand option
 - Internal charge chamber
- Removed
 - Valve component for simplicity
 - External compression chamber



Final Design



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Mathematical Assumptions

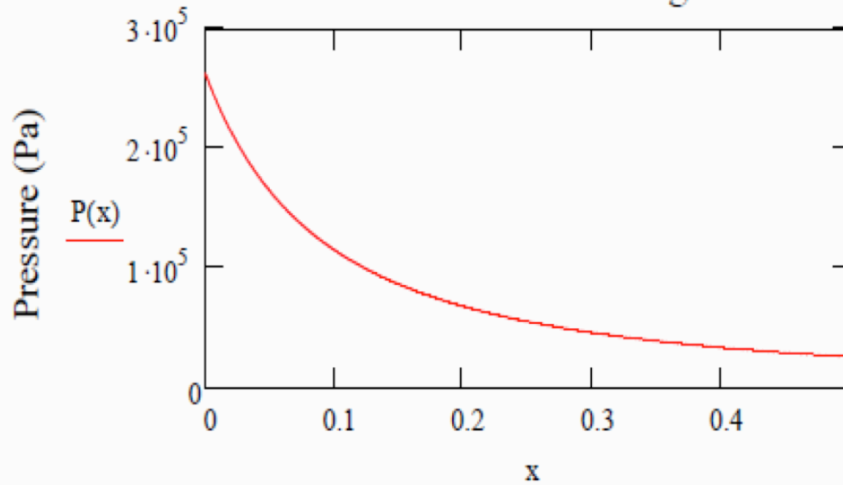


- Adiabatic, isentropic expansion
- Launch angle 45 degrees
- Neglecting vibrations & friction
- Factors of safety
- Impulse stopping force



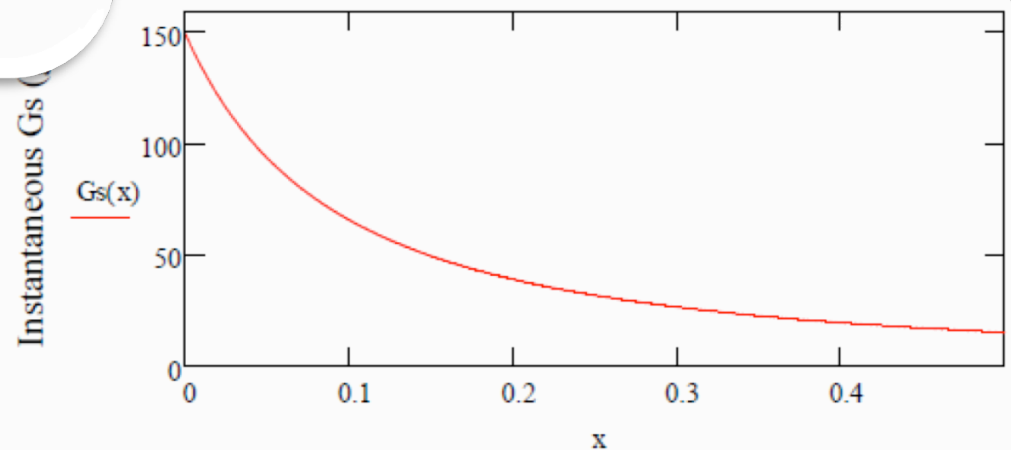
Mathematical Analysis

Pressure Distribution During Launch



Distance Traveled Through Tube (m)

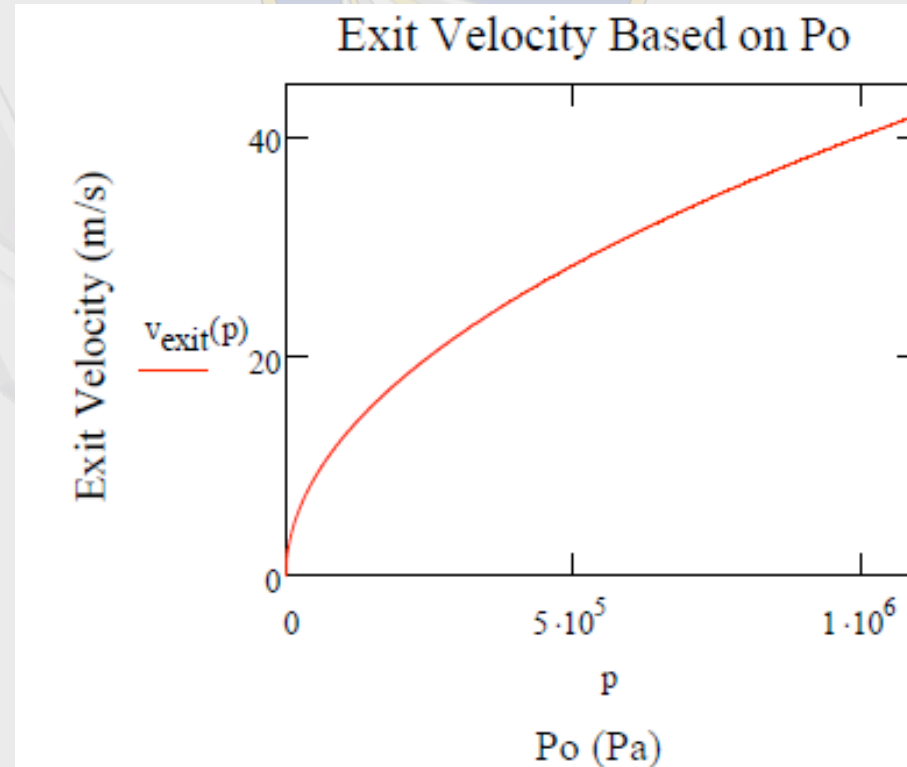
Instantaneous Gs Experienced During Launch



Distance Traveled Through Launch Tube (m)



Mathematical Analysis





Financial Aspect

Prototype Component	Qty	Price	Vendor	Final Design Component	Qty	Price	Vendor
PVC Cleaner	1	4.94	Lowes				
PVC Primer	1	2.52	Lowes	Carbon Fiber Reservoir	1	154.95	Guerrilla Air
Clear Cement	1	3.48	Lowes	In-Line Pressure Regulator	1	39.95	Lowes
Gas Tape	1	2.63	Lowes	Gas Tape	1	2.63	Lowes
5"x3' Acrylic PVC Pipe Sch 40	1	109.36	MMC	4"x2' PVC Pipe Sch 40	1	4.97	Lowes
4"x2" PVC Coupling	1	4.93	Lowes	Steel Braided Line	1	19.90	Lowes
1" PVC Ball Valve	1	5.17	Lowes	1" PVC Ball Valve	1	5.17	Lowes
2"x2' PVC Pipe Sch 40	1	2.53	Lowes	Aluminum Stand Fabrication	1	200.00	Eglin AFB
2"x1' Sch 20 Bushing	1	1.76	Lowes	Carbon Fiber Fabrication	1	375.00	Eglin AFB
5" PVC Cap	2	5.44	Lowes	4" PVC Cap	2	5.44	Lowes
1" Rainbird Valve	1	15.78	Lowes	Rainbird Valve	1	15.78	Lowes
5" Neoprene Backing Disk	1	30.00	MMC	Neoprene Backing Disk	1	-	MMC
5" Foam Backing Disk	1pk	15.00	MMC	Foam Backing Disk	1pk	-	MMC
Total with 7.5% Tax		224.65		Total with 7.5% Tax		891.4	
				Total with Eglin Supplement		267.5	



Experimental Plan

Purpose

To better understand the processes and theories associated with launching a non-uniform projectile from a tube & to collect actual data to manipulate launcher performance.

Components

- Using the prototype PVC launcher
- 50yrd open field

Data Collection

- Methods to reduce “blow-by”
- The effects of barrel length
- Effects of recoil
- Assuring minimum exit velocity of 60ft/s
- Assuring not to exceed maximum acceleration of 600Gs
- How the pressure increases or decreases over the distance of the launch tube



About to Take Flight...

72hr Plan:

- Revise & submit "Spring Proposal"
- Meet with customer for update

7 Day Plan:

- Finalize & check status of all machining, testing, and equipment

14 Day Plan:

- Concentrate on executing spring semester plan effectively



For further info see geocities.com/jrod23dhs/group3.html



Recognition

Sponsor – Eglin AFB



- John Deep
- Jeff Wagener

THANKS TO ALL

Technical Support



- Dr. Alvi - Calculations
- Dr. Shih - Guidance
- Hobby Town USA - Guidance

